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VIRGINIA POLYTECHNIC INST AND STATE UNIV BLACKSBURG --ETC F/G 20/4
THREE-DIMENSIONAL INCOMPRESSIBLE BOUNDARY LAYERS ON BLUNT BODIE--ETC(U)
MAY 77 D L DWOYER, C H LEWIS, P R GOGINENI
VPI/SU-AERO-063-PT-2

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VPI & SU AERO-063

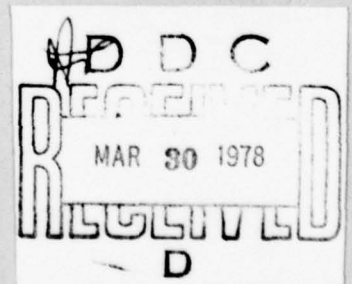
**Three-Dimensional Incompressible
Boundary Layers On Blunt Bodies
Including Effects Of Turbulence, Surface
Curvature And Heat And Mass Transfer
Part II: Computer Code User's Manual**

by

D. L. Dwoyer, Clark H. Lewis and P. R. Gogineni

This research was sponsored by the Applied Physics Laboratory of the Johns Hopkins University under Subcontract Number 600325.

Aerospace and Ocean Engineering Department
Virginia Polytechnic Institute and State University
Blacksburg, Virginia 24061



May 1977

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ABSTRACT

The computer code that was developed as a result of the analysis of Part I of this report, VPI-AERO-063, is herein described and documented. Included are programs input, expected output, description of mode of operation and other information useful in running the program.

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NOMENCLATURE

h_ξ	metric coefficient of the ξ coordinate
h_ω	metric coefficient of the ω coordinate
K_ξ	$(h_\xi h_\omega)^{-1} \partial h_\xi / \partial \omega$, curvature of the ξ coordinate line
K_ω	$(h_\xi h_\omega)^{-1} \partial h_\omega / \partial \xi$, curvature of the ω coordinate line
r	local body radius in stagnation point polar coordinates
s	surface distance along ξ coordinate
U_ξ	inviscid velocity component in ξ direction
U_ω	inviscid velocity component in ω direction
U_e	inviscid velocity component in ξ direction at the body surface
w	$w = W_e$ at stagnation point and on symmetry planes, $W = U_e$ elsewhere
W_e	inviscid velocity component in ω direction at the body surface
x	distance along stagnation point axis
α_1	W/U_e
α_2	W_e/U_e
α_3	W_e/W
β_1	$(h_\xi U_e)^{-1} \xi \partial U_e / \partial \xi$
β_2	$(h_\omega U_e)^{-1} \xi \partial U_e / \partial \omega$
β_3	$(h_\omega W_e)^{-1} \xi \partial W_e / \partial \omega$
β_4	$(h_\xi W_e)^{-1} \xi \partial W_e / \partial \xi$
ξ	orthogonal surface coordinate in longitudinal direction
σ	source density
ϕ	meridional angle in stagnation point coordinate system
ω	orthogonal surface coordinate in transverse direction

INTRODUCTION

A computer program has been developed for solving the three-dimensional boundary-layer flow on blunt bodies using a finite difference procedure. The program is capable of calculating laminar or turbulent flows and accounts for the effects of surface curvature. The equations solved, finite difference procedure used and results of test calculations with the code are presented in Ref. 1.

The boundary-layer program is not a stand alone program, but requires the generation of a data tape which describes the body geometry to be solved and the surface inviscid pressure distribution over the body. This report is divided into two sections, the first describes program TAPGEN which is capable of generating the required data tape for the boundary-layer code, and program ICBL3D which actually solves the boundary-layer equations for the body of interest.

All of the programs have been coded in FORTRAN IV and written for the IBM 370/158 computer with the JES2 operating system.

SECTION I

PROGRAM TAPGEN

In this section a brief description of the computer programs used to generate the inviscid data is presented to enable the user to execute the programs easily. The input data preparation and the outputs available have been described in detail. Finally the JCL required to execute the programs on IBM/370 Model 158 with JES2 operating system is listed.

The necessary computer programs are HESS, AOAT, BLOT, DERVAT and INVTAP. Program HESS generates the body geometry and inviscid velocity distribution for two onset flows: one in x-direction and the other in y-direction. These two flows are combined by program AOAT to give the inviscid velocity distribution at the given angle of attack. Programs HESS and AOAT may be replaced by program HESS3D for calculating the inviscid flow over bodies with non-axisymmetric shapes. Program BLOT reads this velocity distribution and computes the position of surface coordinates, metric coefficients and the inviscid velocities in surface coordinates. All this information is read by program DERVAT which computes the required derivatives. Program INVTAP fits appropriate Fourier series in ϕ ; computes the Fourier coefficients and writes them on a tape as required by the boundary layer programs. These programs taken together represent program TAPGEN.

1. Body Geometry and Inviscid Velocities

The boundary-layer equations have been written in

an orthogonal curvilinear surface coordinate system with its origin fixed at the stagnation point. The boundary-layer computations need the body geometry, metric coefficients and the inviscid velocity distribution. These are computed separately and a brief description of these computations follows.

The surface of the body is defined in the cartesian coordinate system shown in Fig. 1 . These coordinates form the input to the computer program which computes the inviscid velocity distribution at the given angle of attack. This computer program is capable of handling axisymmetric bodies at angle of attack and three-dimensional bodies at zero-angle of attack.

The inviscid flow velocities obtained above have been transformed into a orthogonal curvilinear surface coordinate system in which boundary-layer computations were made. The position of the surface coordinate system and the metric coefficients have been determined as required by the boundary-layer program. The mathematical details of these computations can be found in Ref. 2 .

Since the position (x, r, ϕ) of the surface coordinates and the position of the inviscid velocity would not be the same in general, the inviscid velocities at the surface coordinates have been obtained by double interpolation technique in (x, ϕ) . Then the results are written on a scratch tape unit for every $\xi = \text{constant}$. The quantities on tape are ϕ , h_ξ , h_ω , r , s , U_ξ , U_ω and x . Since the boundary-layer program needs the first

derivatives with respect to ξ , of h_ξ , h_ω , U_ξ , U_ω , x , r and the second derivative of r , they have been computed using the five-point Lagrangian formula and written on the same tape. These data are fitted with appropriate Fourier series in ϕ and the Fourier coefficients have been written on another tape. It is from this tape that the boundary-layer program reads and obtains the necessary information about the geometry and inviscid velocity distribution of the body.

Immediately following is a description of the input to the inviscid program package along with sample input/output and a description of the Job Control Language required.

Basically, the calculation of the inviscid data tape is broken up into three steps as illustrated in the flow chart in Fig. 2. First the inviscid velocity distribution is generated with either of the potential flow programs of Refs.

3 or 4. Next the surface coordinate system is generated by program BLOT. The inviscid velocity field is then interpolated onto the surface coordinate system, Fourier fitted and written on the boundary-layer code data tape in programs DERVAT and INVTAP. Program AOAT is used only with the axisymmetric HESS code and serves to combine the zero and ninety degree angle of attack solutions into the solution at the desired angle of attack.

2. Description of Input Data

Program HESS (Axisymmetric)

A. Card Column Parameter Locations

Card 1 - Header Card.

cc1-60 Header Any alphanumeric run description.

cc63-68 Case Case number.

Card 2 - Control Flag Card

cc1 NB Number of bodies ($1 \leq \text{NB} \leq 9$).

cc2 NNU Number of non-uniform flows ($0 \leq \text{NNU} \leq 5$).

cc3 IAXI Axisymmetric flow flag.

cc4 ICRØSS Crossflow flag.

cc5 IØFF Off-body point input flag.

cc6 IØONLY Basic-data-only flag.

cc7 IELPSE Ellipse generator flag (see also Card 5).

cc8-10 (blank)

cc11 IPRTRB Perturbation velocities only.

cc12 IPØTNL* Solve potential matrix.

cc14 IPTANV Prescribed tangential velocity (for the last IPTANV bodies).

cc15 IVØRT Strip-ring vorticity flag.

cc16 IØMITA Omit axisymmetric uniform flow solution.

cc17 IØMITC Omit crossflow uniform flow solution.

cc18 ISURFV Surface vorticity (instead of sources) for the final ISURFV bodies.

cc19 IPRSCV Prescribed values of the surface vortex strengths for the final ISURFV bodies will be input.

cc20 IALLV All bodies are surface vorticity bodies.

cc21 IEXCRS* Extra crossflow.

cc22 IGENBC Generated boundary conditions.

cc23 IRNGW Ring wing option.

cc28 IPNCH Punched output.

cc29,30 IUNIT Unit number for input coordinates (default = 05).

*Available if and only if NØNEWF = 1, ISIGF = 1 and IGEØMF = 1.

cc31	IVIJ	Matrix print flag.
cc32	ICØEF	Matrix-assembly coefficient print flag.
cc33	IPRINT	Very detailed matrix construction print flag.
cc34	IRAKF	Automatic rake generation flag (see also Cards 8 & 9)

Card 3 - Chord/Mach number card.

cc1-10	CHØRD	Reference chord length (default = 1.0).
cc11-20	XMACH	Mach number for Goethert correction (0.0 implies incompressible).
cc21-80	(blank)	

Card 4 - Body Control Card 1 of 2.

cc1	IGEØMF	0 = curved elements; 1 = flat elements.
cc2	ISIGF	0 = parabolic σ ; 1 = linear σ ; 2 = constant σ (on each element).
cc3	ICURVN	0 = internally calculated element curvatures; 1 = input curvature (see card 7).
cc4	NØNEWF	0 = use the newest formulas; 1 = use the old formulas (implies flat elements and constant σ).
cc5	IFØRMT	Input format flag (see Card Set 6).
cc6-10	NN	Number of defining endpoints for this body.
cc11-20	XMULT	x-multiplier value (default = 1.0).
cc21-30	YMULT	y-multiplier value (default = 1.0).
cc31-40	THETA	Coordinate rotation value (degrees, measured about -Z-axis.
cc41-50	ADDX	x-increment (to be applied to all the input coordinates for this body).
cc51-60	ADDY	y-increment (to be applied to all the input coordinates for this body).

Card 5 - Body Control Card 2 of 2

cc1-10	IBDN	"Body" number (sequential for bodies, zero for off-body points)
cc11-20	ISUBKS	Subcase flag
cc21-30	NLF	Non-lifting flag (for combination cases, only)
cc31-40	A	Semi-major axis for ellipse cases
cc41-50	B	Semi-minor axis for ellipse cases

} If
IELPSE
≠ 0

Card (Set) 6 - Body Definition Cards

IF IFØRMT=0: X-coordinates (6F10.5), then
Y-coordinates (6F10.5)

IF IFØRMT=1: X, Y coordinates (2F10.5) (i.e., one
"point-set" per card).

IF IFØRMT=2: X, Y coordinates (F10.5,10X,F10.5)
(i.e., one "point set" per card).

Card (Set) 7 - Input curvature values (needed only if
ICURVN ≠ 0)

(6F10.5) CURV(I), I=1, NN-1 The curvature values for the
NN-1 elements which constitute this
body

Repeat Cards 4-7 a total of (NB+IØFF) times

Card 8 - Rake Number Card (needed only if IRAKF ≠ 0)

cc1-10 NRAKES The number of "automatically" generated
mass-flow rakes

Card 9 - Rake Definition Card (needed only if IRAKF ≠ 0)

cc1-10	X1	} Coordinates of "start" of the rake
cc11-20	Y1	
cc21-30	X2	} Coordinates of the "end" of the rake
cc31-40	Y2	
cc41-45	N	Number of intervals to be used in the rake (note, $4 \leq N \leq 200$, and N must be an even integer)

Repeat Card 9 a total of NRAKES times.

Card 10 - Non-Uniform Flow Control Card (needed only if NNU \neq 0)

cc1-10	NUN	Flow identification number
cc11-20	MSF	0 = axisymmetric onset flow; 1 = cross-flow onset flow, 2 = both 0 and 1
cc21-30	TYPE	+1.0 = velocity will be input in x,y component form; 0 = velocities will be input in normal, tangential form; -1 = automatic generation of flow due to rotation about the Z-axis (for crossflow, only)
cc31-40	FG	Flow generator constant

Card (Set) 11 - Non-Uniform Flow Velocities (needed only if NNU \neq 0)

(6F10.5)	VX(I) or VN(I), I = 1, total number of control points
(6F10.5)	VY(I) or VT(I), I = 1, total number of control points

Repeat Cards 10 and 11 a total of NNU times.

Card (Set) 12 - Specified Tangential Flow Velocities
needed only if IPTANV \neq 0

(6F10.5)	TG(I), I = 1, total number of control points on the last IPTANV bodies
----------	--

B. Discussion of Parameters

In all cases, the integer values (indicated by the standard FORTRAN IV naming convention) must be right adjusted within the specified input fields. For floating-point type input, the input decimal point will override the quoted FORMAT specifications.

Card 1. Self-explanatory.

Card 2. Although most of these flags are unchanged from the original program, a brief description of each one will be given here for completeness.

NB The total number of separate "bodies" ($1 \leq \text{NB} \leq 8$).

NNU The total number of (user specified) input non-uniform flows ($0 \leq \text{NNU} \leq 5$).

IAXI Axisymmetric flow flag (0: no axisymmetric flow, 1: axisymmetric flow)

ICROSS Crossflow flag (0: no crossflow, 1: crossflow)

IOFF Off-body point input flag (0: no off-body points will be input, 1: off-body points will be input)

IONLY Basic-data-only flag (0: full execution, 1: basic data only)

IELPSE Ellipse generator flag (see also card 5) (0: standard body defining coordinates will be input, 1: coordinates for the ellipse whose properties are specified on card 5 are to be created automatically by the program).

IPRTRB Perturbation velocities only (0: standard total net velocities including the onset flow will be printed, 1: perturbation velocities only will be printed).

IPOTNL Solve the potential matrix (0: standard solution for the surface velocities, 1: solution for the surface velocity potential). Since the velocity potential problem for the higher order elements has not been coded in the present version, the solution for the

potential is available only for the case of flat elements and, in particular, through the usage of the "old" velocity formulas. These "old" formulas have been retained as a subsection of this new, higher order program and may be "reached" by setting the following quantities (on Card 4): $IGEO MF = 1$, $ISIGF = 2$, and $NONEWF = 1$; in such a case, the solution for the potential is still available.

IPTANV Prescribed tangential velocity flag (0: standard zero normal velocity solution is to be obtained, 1: a user defined set of prescribed tangential velocities will be input on the last IPTANV bodies; obviously, $IPTANV \leq NB$).

IVORT Strip ring vorticity onset flows for each of the bodies (0: no strip ring vorticity, 1: automatic generation of the strip ring vorticity onset flows).

IOMITA Omit the uniform axisymmetric flow solution (0: calculate the standard uniform axisymmetric flow solution if $IAXI \neq 0$, 1: omit this uniform axisymmetric flow solution even if $IAXI \neq 0$).

IOMITC Omit the uniform crossflow solution (0: calculate the uniform crossflow solution if $ICROSS \neq 0$, 1: omit the uniform crossflow solution even if $ICROSS \neq 0$).

ISURFV Surface vorticity for axisymmetric flow (in place of surface sources) flag (0: use the standard source

distribution on all elements, $\neq 0$: use vorticity instead of a source distribution as the singularity on the elements of the last ISURFV bodies). Note that if ISURFV $\neq 0$, then both IVØRT and IPTANV must be $\neq 0$, and then in particular, IPTANV must = ISURFV). If we assume that IPTANV = ISURFV = k, and that M is the total number of elements on the last k bodies, then the surface vorticity option causes the source induced velocity formulas to be used on the first N-M elements of each matrix row, and the vortex formulas to be used on the remaining M elements of each matrix row. The solution for the unknown strengths then proceeds as it would have if only IPTANV was nonzero.

IPRSCV Prescribed vorticity flag for axisymmetric flow (0: prescribed vorticity values will not be input, 1: prescribed vorticity values (W_i) for the last IPRSCV bodies will be input; note that if IPRSCV is $\neq 0$, then both IVØRT and IPTANV must $\neq 0$, and then, in particular, IPTANV must = IPRSCV). If IPRSCV is $\neq 0$, then the vortex strengths of the affected elements are taken to be $W_i/4\pi$.

IALLV Total vorticity flag for axisymmetric flow (0: all elements are assumed to be source type elements (unless ISURFV $\neq 0$), 1: all elements are assumed to be vortex type elements). Note that if IALLV is $\neq 0$, the use of IPTANV is optional, but not mandatory.

IEXCRS Extra crossflow flag (0: no "extra" crossflow, 1: generate an "extra" crossflow (i.e., having a potential which varies as the cosine of twice the circumferential angle)). Note that if IEXCRS \neq 0, then NØNEWF = 1, ISIGF = 2, and IGEOMF must = 1, since the higher order formulas for this kind of velocity potential have not been included in this program.

IGENBC Generated boundary condition flag (0: do not generate any additional boundary conditions for the crossflow case; 1: generate the onset crossflow due to rotation about an axis normal to the axis of symmetry (see also the notes for Card 10)).

IRNGW Ring wing option (0: no ring wing option, 1: use the ring wing option (see MDC Report J0741/01, April 1970 for further details)).

IPNCH Punched card output flag (0: no punched card output, 1: punched card output).

Card 3. This card is usually left blank, which results in a default chord length of unity, and no Mach number corrections (i.e., incompressible results). If a non-zero value of the Mach number is input, the program uses the Goethert correction to account for compressibility.

Card 4. Usage of the IGEOMF and ISIGF flags permit the user to "turn-off" any or all of the higher order element

curvature and/or varying source density terms. The default values are curved elements with parabolically varying source density. If $IG\emptyset MF = 0$, the program will automatically calculate the local element curvature values by the procedure described in Section II.A, unless $ICURVN$ is non-zero, in which case the user must supply these curvature values (see Card 7). The value for the $N\emptyset NEWF$ flag is ordinarily left blank (or zero), even if a flat element, constant source density solution is required. However, since certain of the original Douglas-Neumann Axisymmetric Potential Flow Program capabilities have not been made available in the higher order program (e.g., calculation of the potential (hence the added mass, etc.) the original formulas have been preserved within this version (see description of the $IP\emptyset TNL$ and $IEXCRS$ flags). The original capabilities can be obtained by setting $IG\emptyset MF = 1$, $ISIGF = 2$, and $N\emptyset NEWF = 1$ for each of the input bodies, in which case the input instructions of Reference 5 apply.

The other parameters on Card 4 are self-explanatory, with the understanding that the order of coordinate transformations are as listed on the input.

Card 5. The "bodies" are normally loaded prior to any off-body points (although the latter require the usage

of body control Cards 1 and 2, also). For this reason, the value of IBDN should be sequentially increasing beginning with unity. A non-zero value for ISUBKS means that the body definition cards for this body (Card Set 6), (say this is the i th body input under this header card) will not be included, but that the program is to use the i th set of points that were input under the previous header card. Obviously, this capability is useful only with "stacked" input cases.

- Card 6. The three input formats that are available are as shown. The value of IFORMT determines which format is used. The default format is the "old" format: x-coordinates, followed by the y-coordinates.
- Card 7. These values are to be input only if ICURVN \neq 0.
- Card 8. This card is needed only if IRAKF \neq 0. Note that $1 \leq \text{NRAKES} \leq 20$.
- Card 9. These are the rake definition cards, which define the "start" and "end" of each mass flow rake. The sign convention that is employed is as follows: a positive mass flux means that the flow is from left to right to an observer traversing the rake from point (X1, Y1) to point (X2, Y2). For example, for an inlet without a bullet, point 1 is typically at $y = 0.0$, and point 2 would have a y-coordinate located on the inlet wall. Note that the x-coordinates need not be the same, i.e., a tilted rake may be used,

if desired. Note also that one rake may "overlap" another if so desired, since they are each treated independently. It should be pointed out that a Simpson Rule integration is performed over the "N" intervals (and therefore N must be an even integer), with the program automatically generating N-1 "intermediate" points as "off-body points" for each input rake (program limit is 500 total off-body points). The velocity values at the first and last rake points are obtained by linear extrapolation of the two nearest values. In this way, the difficulty associated with calculating induced velocities at off-body points which lie very near to, or on, the surface is avoided. Typically, values of N between 10 and 20 appear to be satisfactory for most cases.

Card 10. Self-explanatory, except that FG, if entered is used in the following way to generate the rotation onset flow:

$$V_{x_i} = y_i$$

$$V_{y_i} = FG - x_i$$

$$V_N = V_{x_i} \sin \alpha_i - V_{y_i} \cos \alpha_i$$

$$V_T = V_{x_i} \cos \alpha_i + V_{y_i} \sin \alpha_i.$$

Card 11. Self-explanatory.

Program AOAT

Card 1. Format (3I3,F10.6)
cc1-3 IPRT = 1, writes the output of program HESS
 on Unit 6; = 0, does not write
cc4-6 N, Number of elements used in the program HESS
cc7-9 KPL, Number of meriodonial planes required
cc10-19 ALPD, Angle of attack in degrees

Program HESS (3-D)

Card 1.

<u>cc</u>	<u>CONTENTS</u>	<u>NOTES</u>
1-60	Title	Header information
61	IFLAG	= "0" for input body coordinates (this is the usual case) = "1" for generated ellipsoids = "2" for generated bodies with elliptical cross section
62	LIST	= "0" for full execution = "1" for basic data listing only
63	KX	= "F" or blank for no generated uniform x-flow = "T" for generated uniform x-flow (calculate x-flow matrices)
64	KY	= "F" or blank for no generated uniform y-flow = "T" for generated uniform y-flow (calculate y-flow matrices)

<u>cc</u>	<u>CONTENTS</u>	<u>NOTES</u>
65	KZ	= "F" or blank for no generated uniform z-flow = "T" for generated uniform z-flow (calculate z-flow matrices)
66	ISIG	= "0" for no input guesses for source densities = "1" for input guesses for source densities (rarely used)
67	IPRS	= "0" for no mid-iteration print = "1" for print of solution iterations
68	MPR	= "0" for no intermediate matrix print = "1" for V_{ij} print = "2" for A_{ij} print = "3" for V_{ij} and A_{ij} print (Never use for large case, or output is excessive)
69	METHØD	= "0" for automatic solution selection = "1" for direct matrix solution = "2" for modified Seidel iterative solution
70+71	NNØN	Number of input onset flows
72	NSYM	Number of symmetry planes
73	NØFF	= "0" for no off-body points = "1" for off-body points
74	KMACH	= "0" for no Mach number correction = "1" for input Mach number

<u>cc</u>	<u>CONTENTS</u>	<u>NOTES</u>
75-76	---	
77-80	KASE	Alpha-numeric case identification

Mach Number is input on a separate card only if KMACH = 1.

INPUT GENERATOR DATA

Card 2.

<u>cc</u>	<u>CONTENTS</u>	<u>NOTES</u>
1 → 5	NLM1	Number of "latitudinal" element divisions
6 → 10	MMIN	Number of "longitudinal" element divisions
11 → 20	B	Y semi-axis of elliptical cross section
21 → 30	C	Z semi-axis of elliptical cross section

(If ellipsoid is generated, X semi-axis is unity)

NOTE: This card is input only when IFLAG ≠ 0

X-Z INPUT

This input consists of the coordinate pairs ($X_1, Z_1, X_2, Z_2, \dots, X_{NLM1+1}, Z_{NLM1+1}$) that define the profile in the x-z plane of the body of elliptical cross-section to be generated by the program. These coordinates are read with an 8F10.0 format.

ON-BODY COORDINATES

General (non-elliptical) bodies are input by giving the

coordinates of a large number of points on the body. The three dimensional coordinates defining the body surface(s) are input 2 points-to-a-card. Each point has associated with it a quantity called its "status" that has the following meaning:

<u>VALUE</u>	<u>MEANING</u>
"0" or blank	This point is on the same n-line as the last point
1	This point starts a new n-line
2	This point starts a new section
3	This is the last point of the input

NOTE: These cards are only input if IFLAG=0

OFF-BODY POINT COORDINATES

The three-dimensional coordinates defining the off-body points are input with the identical format of the on-body coordinates. The only status flag used, however, is that of "3" for the last off-body point.

NOTE: These cards are input only if NØFF=1.

FLOW FLAGS

Each input onset flow is preceded by a card that describes the flow. All x-flows must precede all y-flows which precede all z-flows. (An "x-flow" is any flow that is solved using the matrix appropriate for a uniform onset flow parallel to the x-axis.)

ccMEANING

- | | |
|---|---|
| 1 | = "1" if this is an x-flow
= "2" if this is a y-flow
= "3" if this is a z-flow |
| 2 | = "0" if this is a non-uniform onset flow, i.e., values of the onset flow velocities will be specified for each element
= "1" if this is a uniform onset flow, the value of which is only specified once |
| 3 | = "0" if the onset flow velocity is expressed in component form
= "1" if the onset flow is expressed in normal velocity values |

NOTES: The onset flow velocities and sigma guesses are read with a "6F10.0" format.

This input is only required if NNØN = 0.

GENERAL PROCEDURES

All coordinates are input with the following format

2 (3F10.0, 11)

Except where noted elsewhere, zeroes or blanks may be interchanged on input.

Program BLOT

<u>Card Name</u> <u>Col. 1-5</u>	<u>Fortran</u> <u>Name</u>	<u>Format</u>	<u>Column</u>	<u>Description</u>
TITLE	TITLE	9A8	6-77	Problem Identification - this will appear as a heading in the output
LIMTS	JMAX	I5	6-10	Number of points across boundary layer
	KMAX	I5	11-15	Number of points around body (in phi-direction)
	LITER	I5	16-20	Number of iterations for initial profile
	IPRT	I5	21-25	Every how many steps printing will occur along the body
	KPRT	I5	26-30	Every how many steps printing will occur around the body
	PSIMAX	E10.0	31-40	Distance along body at which program will term- inate
	UINF	E10.0	41-50	Freestream velocity - V_{∞}
DW	DW(K)	E10.0	6-15	$\Delta\omega$'s around the body, $K = 1, KMAX - 1.$
			16-25	
			.	
			.	
			.	
			etc.	
DN	DN(J)	E10.0	6-15	$\Delta\eta$'s across the boundary layer. $J = 1, JMAX - 1.$
			16-25	
SIZE	A	E10.0	6-15	a - length of axis of triaxial ellipsoid ($A > 10^6$ gives paraboloids)

<u>Card Name</u> <u>Col. 1-5</u>	<u>Fortran</u> <u>Name</u>	<u>Format</u>	<u>Column</u>	<u>Description</u>
	RB	E10.0	16-25	R_b - radius of curvature of nose of body in the plane $y' = 0$
	RC	E10.0	26-35	R_c - radius of curvature of nose of body in the plane $z' = 0$
	EPS	E10.0	36-45	Convergence criterion used in calculating the body coordinates
	ALP	E10.0	46-55	α - angle of attack (deg)
	CAPTH	E10.0	56-65	Θ - indicates which finite-difference scheme is being used. = { 1 = Kruase scheme 2 = Crank-Nicolson scheme
	CRI	E10.0	66-75	= { 1 Implicit scheme 1/2 Crank-Nicolson scheme

Programs Dervat and INVTAP

These programs obtain the necessary data sets internally and they do not need any external input.

3. Description of Program Output

Program HESS(Axisymmetric)

Unit 6

The main output of the program, which consists of calculated surface velocities and/or pressures has not been changed. Its format is that of Ref.2. The surface coordinates with which the calculated velocities are associated are the

transformed coordinates, i.e., the original input coordinates altered by any specified translations, multiplications, or rotations. Similarly, the format of the calculated velocities at the off-body points is similar to that of Ref. 5. The only change is that output at rake points (q.v.) is included.

The initial output, which details the surface geometry, has been changed considerably. The header identifies what kind of solution has been computed: (1) curved or flat elements; (2) constant, linear, or parabolic source density element curvatures. The header also specifies whether "new" or "old" velocity formulas have been used, the latter of which apply only to the flat-element constant-source case. In the body of the output, the first two columns are the untransformed (input) coordinates, and the third and fourth columns are the transformed coordinates, which are the endpoints of the surface elements. The fifth and sixth columns are the coordinates of the control points of the elements. Element arc lengths are given in column 7 and a running total of arc length is displayed in column 8. Column 9 lists the differences between average slopes of successive elements, i.e., the differences in the slopes of the two straight lines, through the respective element endpoints. The actual slope discontinuity at an endpoint between two parabolic elements is normally much smaller than the difference between the average slopes of the two elements. Moreover, this discontinuity approaches zero if the body contour approaches a parabola. The final (tenth) column lists elements curvatures.

A new output is entitled Automatic Rake Calculation. For each rake the two input points that bound the rake and the input identification are output together with three calculated quantities. The first is the surface area of the cone frustum defined by the two input rake points. The third is the total flux of fluid that crosses this area per unit time. The second quantity, average velocity, is the ratio of the third and the first quantities. There is a rake output for all axisymmetric flows, both uniform and non-uniform, but there is no rake output for cross-flow, because its circumferential variation guarantees zero flux.

With regard to surface vorticity solutions, there are only two possibilities. Either there are no vorticity solutions or there is one solution for each body, which corresponds to a unit vorticity strength on that body and zero vorticity strength on all other bodies. The order of these solutions is the same as the order in which the bodies are input. Thus, in particular, an inlet with centerbody has two vorticity solutions - one with vorticity on the inlet and one with vorticity on the centerbody. The second of these is not meaningful and should be discarded. (It may, of course, occur first on the output.)

Units 1- 4,
7-15

These are scratch disks, which are used to write and read internally.

Unit 16

The coordinates x , y and the velocities T_1 , T_1 , T_2 are written in order one after the other for all axial stations.

Program AOAT

Unit 6

For each axial station (x), the pressure coefficient (CP), the velocities UX, UY, UZ are written. These velocities are in body axes. Finally, the stagnation point location XO is printed.

Unit 10

For each axial station, two records are written unformatted. The first record contains X and KPL. The second record contains the meridional plane angle, UX, UY, UZ and radius R for all planes.

Unit 22

This is not used.

Unit 30

Formatted image of Unit 10.

Program BLOT

Unit 6

Initially all input data is printed out. This includes several quantities which are computed strictly from the input. Next, initial profile calculations, including all iterations are printed. Finally, the solution at each desired location along and around the body is displayed. At each point where printing occurs, the following quantities appear:

X	}	x,y,z coordinates
Y		
Z		

R	r, radial distance to body surface
S	s, distance along surface in ξ -direction
ALPHA	$\bar{\alpha} = w_e/u_e$
ALBET2	$\bar{\alpha}_2 = w_e/u_e$
ALBET3	$\bar{\alpha}_3 = w_e/W$
XIKA	parameter $\xi K_\xi/\bar{\alpha}$
UE	$\left. \begin{matrix} u_e \\ w_e \end{matrix} \right\} \text{ inviscid edge velocities in } \xi \text{ and } \omega \text{ directions}$
WE	
BETA1	$\left. \begin{matrix} \beta_1 \\ \beta_2 \\ \beta_3 \\ \beta_4 \end{matrix} \right\} \text{ inviscid velocity gradient parameters}$
BETA2	
BETA3	
BETA4	
DFDN	$\left. \begin{matrix} \\ \end{matrix} \right\} \text{ Not calculated}$
DGDN	
HPSI	$\left. \begin{matrix} h_\xi \\ h_\omega \end{matrix} \right\} \text{ Metric coefficients}$
HOMEGA	
KAPPAS	$\left. \begin{matrix} K_\xi \\ K_\omega \end{matrix} \right\} \text{ Curvatures of the coordinate lines}$
KAPPAW	
TOTAL SHEAR STRESS	$\left. \begin{matrix} \\ \\ \end{matrix} \right\} \text{ Not calculated}$
TOTAL SKIN FRICTION	
ANGLE	

Unit 8

Formatted image of Unit 25

Unit 11

The diagnostic printout from the double interpolation subroutine SLOALL. Useful for debugging only.

Unit 25

For each (PSI) the longitudinal surface coordinate, the metric coefficients (HPSI, HW), R, S, UPSI, UW are written for each meridional plane, unformatted.

Program DERVAT

Unit 6

The angle of attack in radians and the number of meridional planes are written. Next the image of Unit 25 of program BLOT has been written.

Unit 26

The angle of attack in radians and KPL are written. Next, the station No. I, XS, HPSI, HW, R, S, UPSI, UW, X, and the derivatives of these $D(UPSI)$, $D(UW)$, $D(R)$, the second derivative $D^2(R)$; $D(HPSI)$, $D(HW)$, $D(X)$, are written in order for each meridional plane. These are written unformatted.

Unit 29

Formatted image of part of Unit 26. The data written is I, XS, S, X, $D(X)$, UPSI, $D(UPSI)$, UW, and $D(UW)$ for each plane.

Unit 30

Formatted image of part of Unit 26. First the angle of attack in radians and KPL are written. Next I, XS, HPSI, $D(HPSI)$, HW, $D(HW)$, R, $D(R)$ and $D^2(R)$ are written for each plane.

Program INVTAP

Unit 6	Errors and messages are written on this unit.
Unit 10	For each station the Fourier coefficients are written unformatted. This is the data set required by ICBL3D.
Unit 30	Formatted image of Unit 10. Also at each station the data written on Unit 26 of program DERVAT is rewritten.

JCL

The programs described above require large memory, which made it necessary to execute them one after the other in a multi-step job. Also the programs were compiled, linked and stored as members of a partition data set, named A30303. MARK1. SUBLIB. This has the advantages of avoiding handling large Fortran source decks and compiling everytime the job is executed. The input cards for each program have to be inserted after the //FT05F001 DD* card, except for the program BLOT for which they have to be after the //FT04F001 DD* card. Finally, the required inviscid data has been written on an on line disk (CHEKOV) as a partition data set named A505F3.INVTAP. The relevant JCL has been listed below.

1	Job Card
2	//JOB LIB DD DSN=A30303, MARK1, SUBLIB, DISP=(SHR, KEEP), //VOL=SER=USERPK, UNIT=SYSDA
3	//STEP1 EXEC PGM=HESS, TIME=1, REGION=520K
4	//FT06F001 DD DUMMY

```

5      //FT01F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
6      //FT02F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
7      //FT03F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
8      //FT04F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
9      //FT07F001 DD DUMMY
10     //FT08F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
11     //FT09F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
12     //FT10F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
13     //FT11F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
14     //FT12F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
15     //FT13F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
16     //FT15F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE
      =(TRK,(50,1))
17     //FT16F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE
      =(TRK,(50,1)),DSN=&&T1
18     //FT05F001 DD *
19     //STEP2 EXEC PGM=AOAT,COND=(0,NE),TIME=(0.30),
      REGION=150K
20     //FT06F001 DD DUMMY
21     //FT16F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T1
22     //FT22F001 DD DUMMY
23     //FT30F001 DD DUMMY
24     //FT10F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE=
      (TRK,(50,1)),DSN=&&T2

```

```

25      //FT05F001 DD *
26      //STEP3 EXEC PGM=BL0T,COND=(0,NE),TIME=5,REGION=350K
27      //FT06F001 DD SYSOUT=A
28      //FT08F001 DD SYSOUT=A
29      //FT11F001 DD DUMMY
30      //FT10F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T2
31      //FT25F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE=
        (TRK,(50,1)),DSN=&&T3
32      //FT04F001 DD *
33      //STEP4 EXEC PGM=DERVAT,COND=(0,NE),TIME=1,REGION=650K
34      //FT06F001 DD SYSOUT=A
35      //FT25F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T3
36      //FT26F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE=
        (TRK,(50,1)),DSN=&&T4
37      //FT29F001 DD SYSOUT=A
38      //FT30F001 DD SYSOUT=A
39      //STEP5 EXEC PGM=INVTAP,COND=(0,NE),TIME=1,REGION=650K
40      //FT06F001 DD SYSOUT=A
41      //FT25F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T4
42      //FT10F001 DD UNIT=SYSDA,DISP=(NEW,CATLG),SPACE=
        (TRK,(50,1)),
        // DSN=A505F3.INVTAP,VOL=SER=CHCKOV
43      //FT30F001 DD SYSOUT=A

```

JCL for use with Fortran source decks

In the JCL listed above the following changes have to be made:

1. Card 2 has to be removed.

2. In card 3 (PGM = HESS) has to be replaced by (FORTGCG)
19 (= AOAT) has to be replaced by (FORTGCG)
26 (= BLOT) has to be replaced by (FORTGCG)
33 (= DERVAT) has to be replaced by (FORTGCG)
39 (= INVTAP) has to be replaced by (FORTGCG)
3. After cards 3, 19, 26, 33, 39 insert the following card:

//FORT.SYSIN DD* and then appropriate Fortran source deck.

Listings of Job Control Language, sample data decks, sample output and program listings for the TAPGEN package are found in Appendices I through V.

SECTION II

PROGRAM ICBL3D

In this section program ICBL3D is described. The general structure of the program is briefly described and then the input data and output of the program are described in detail. Finally, the Job Control Language required to execute the program on an IBM 370/158 computer with the JES2 operating system is listed.

A flow chart of the program appears in Fig. 3 where the function of the various subroutines are indicated. As can be seen, the code requires a small amount of card input, but its' primary mode of data input is from a data tape designated as logical unit 10. This data tape can be prepared using program TAPGEN described earlier in this report.

The program proceeds in a forward marching fashion from the body stagnation point downstream and from the windward to leeward symmetry planes. The number of solution planes around the body in the ω -direction is fixed by the user, while the step size in the ξ -direction is adjusted internally by the program as it proceeds downstream. The initial step size is set by the user and this step size is continually adjusted by the program based upon the number of iterations required to achieve a solution at a previous station. Further, at the user's option, the edge value of the transformed normal coordinate η can be automatically increased or decreased to meet a predetermined criteria for profile decay at the boundary-layer outer edge.

There is a special feature of the program for full three-dimensional solutions. If the program cannot obtain a converged solution at some point between the windward and leeward symmetry planes it will drop that point (and all subsequent points to the leeward plane) from the solution around the body at all subsequent streamwise stations. This feature allows the solution to proceed even after separation or other problems are encountered near the leeward plane of the body.

1. Program Input

As mentioned, input to the program is from two different sources: (1) cards or card image data and (2) data sets residing on tape or disk. The data sets residing on tape or disk are generated by program TAPGEN and basically consist of Fourier coefficients. The first record on the tape contains the angle of attack in radians and an integer variable giving the number of coefficients in each Fourier series. All the records on the file after the first are identical in format. They consist of the value of ξ followed by the Fourier coefficients for the series fits to r , U_e , W_e , h_ξ , h_ω , s , $\partial U_e / \partial \xi$, $\partial W_e / \partial \xi$, $\partial r / \partial \xi$, $\partial^2 r / \partial \xi^2$, $\partial h_\xi / \partial \xi$, $\partial h_\omega / \partial \xi$, x , $\partial x / \partial \xi$ in that order. The series are in terms of the angle ϕ and their range is from the windward to leeward planes of symmetry for constant ξ . All data on the tape are written with unformatted WRITE's. The data must appear on logical unit 10.

The card or card image data to the program will now be described.

2. Description of Input Data

CARD 1. LABEL (20A4)

LABEL is the title of the case and is a single subscripted array. LABEL is passed to the output routines and appears as the header to the program output.

CARD 2. IE (49X,I3)

IE is the number of points taken normal to the body. The program is dimensioned for a maximum of 101 points in the normal profile arrays. 101 is the recommended value; however, savings in execution time occur by decreasing IE. IE must be an odd number.

CARD 3. INJCT (49X,I3)

INJCT is the subscript of the XSTA array (see below) giving the surface location at which mass injection begins. A zero value of INJCT is set to NSOLVE.

CARD 4. KADETA (49X,I3)

KADETA is an indicator for the adjustment of the transformed normal coordinate. If KADETA = 0, the maximum value of η is held constant. If KADETA = 1 the maximum η is adjusted when the velocity profiles fail to decay properly at the outer edge as prescribed by the input variable ADTEST. The value of η_{\max} as adjusted up or down for windward symmetry plane only runs. For full three dimensional runs η_{\max} is only adjusted up, and can be adjusted at every point. A value of 1 is recommended.

CARD 5. KEND2 (49X,I3)

KEND2 specifies the number of circumferential planes to be used in the solution, and thereby also sets the circumferential step size. The program can accept a maximum value of KEND2 = 61. A total of sixty-one planes would lead to excessive computing time, however, and a value of KEND2 = 11 is recommended.

CARD 6. KONSET (49X,I3)

KONSET is the subscript of the XSTA array giving the location of the onset of transition. At $X = XSTA(KONSET)$ the variable LAMTRB is reset to 2 and transition to turbulence begins. A zero value is reset to NSOLVE.

CARD 7. KPRT (49X,I3)

KPRT is a print control parameter which controls the printing of profiles in the ϕ direction. The program prints every KPRT'th profile around the body at a given value of ξ starting with the $\phi = 0^0$ profile.

CARD 8. KTRANS (49X,I3)

KTRANS is an indicator for the transition model. If KTRANS = 0 transition to turbulence will be instantaneous, if KTRANS = 1 a smooth transition to turbulence will take place over the distance XBAR (see below).

CARD 9. LAMTRB (49X,I3)

LAMTRB indicates whether the flow is laminar or turbulent. LAMTRB = 1 indicates the solution begins with laminar

flow. LAMTRB = 2 means fully turbulent flow. When KONSET \neq 0, LAMTRB = 1.

CARD 10. LPRT (49X,I3)

LPRT is the print control parameter in the streamwise direction. The program prints solutions at every LPRT'th station in ξ .

CARD 11. NIT1 (49X,I3)

NIT1 is an iteration counter used to adjust the streamwise step size. If the total number of iterations required to obtain the solution at a point is less than or equal to NIT1 then the ξ step size is doubled.

CARD 12. NIT2 (49X,I3)

NIT2 is an iteration counter used to adjust the streamwise step size. If the total number of iterations required to obtain a solution at a point is greater than NIT1 and less than NIT2 the ξ step size is unchanged.

CARD 13. NIT3 (49X,I3)

NIT3 is an iteration counter affecting ξ step size and the convergence of the solution. If the total number of iterations required for a solution at a particular point is greater than NIT3 the program halves the ξ step size and cuts back the value of ξ by the new step size. A solution at the smaller value of ξ is then tried. If this procedure fails three consecutive times execution is terminated.

NOTE: The ξ step size is adjusted only at the windward symmetry plane.

CARD 14. NOINJ (49X,I3)

NOINJ is the subscript of the XSTA array giving the surface location at which injection ends. NOINJ = 0 is reset to NOINJ = NSOLVE.

CARD 15. SFC (49X,A3)

SFC is a literal variable coded as either YES or NO to indicate whether surface curvature effects are to be included in the calculations.

CARD 16. NSOLVE (49X,I3)

NSOLVE is the number of variables in the XSTA array, and is therefore the subscript of the last XSTA value which indicates the end of the body. It is also the default value for INJCT, NOINJ, and KONSET.

CARD 17. ADTEST (49X,E14.6)

ADTEST is used in conjunction with KADETA. When KADETA is 1 ADTEST provides the convergence criteria for checking the streamwise velocity profile. When $|F(IE) - F(IE-4)|$ or $|G(IE) - G(IE-4)|$ is less than ADTEST/10 the maximum value of η is decreased by 10%. When it is greater than ADTEST the maximum value of η is increased by 10%.

CARD 18. AKSTAR (49X,E14.6)

AKSTAR is a numerical constant in the Van Driest inner eddy viscosity law (k^* in Ref. 1). The recommended value is 0.435.

CARD 19. ALAMDA (49X,E14.6)

ALAMDA is a numerical constant in the outer eddy viscosity law used in the program (λ in Ref. 1). The recommended value is 0.09.

CARD 20. ASTAR (49X,E14.6)

ASTAR is a numerical constant used in the damping term of Van Driest's inner eddy viscosity law in the program (A^* in Ref. 1). The recommended value is 26.0.

CARD 21. CWALL (49X,E14.6)

CWALL is the injection rate for mass transfer cases where the rate is a constant. $CWALL = v_w/u_\infty$.

CARD 22. CONV (49X,E14.6)

CONV is the solution convergence criterion. The dependent variable arrays of streamwise and cross flow velocities are all checked for convergence at all points. When the largest percentage difference between the current and previous iterations is less than or equal to CONV the solution is taken to be converged.

CARD 23. DXMAX (49X,E14.6)

DXMAX is the maximum step size, to be taken in the streamwise direction.

CARD 24. DX1 (49X,E14.6)

DX1 is the initial streamwise step size. Since this value will be adjusted internally it is not critical that

the user choose an accurate value. Usually a value of 0.01 is a good initial DX.

CARD 25. EDYLA W (49X,A3)

EDYLA W specifies the inner eddy viscosity law to be used in turbulent cases. Two options are available to the user: (1) EDYLA W = VAN DRIEST and (2) EDYLA W = REICHARDT. The program picks up only the first 3 letters of each name. The user should note the comments on these two laws for mass transfer problems stated in Ref. 1. In general the REICHARDT law is recommended.

CARD 26. ETAFAC (49X,E14.6)

ETAFAC controls the normal grid spacing. A value of 1.0 gives an equally spaced grid for the transformed normal coordinate. A value greater than 1.0 gives a finer grid at the wall than at the outer edge. A value of 1.04 is recommended with IE = 101 and ETAINF = 6.0. A value of 1.09 is recommended with IE = 101 and ETAINF = 100.0.

CARD 27. ETAINF (49X,E14.6)

ETAINF is the maximum value of η . A value between 6.0 and 10.0 is recommended for laminar flow. A value between 10.0 and 100.0 is recommended for turbulent flow. This value can be adjusted internally by specifying KADETA = 1.

CARD 28. PR (49X,E14.6)

PR is the value of the laminar Prandtl number in the fluid.

CARD 29. RTW (49X,E14.6)

RTW is the ratio of the wall temperature to stagnation temperature. It is used to calculate wall temperature when wall temperature is held constant.

CARD 29. TFS (49X,E14.6)

TFS is the free-stream static temperature in degrees Rankine.

CARD 30. CP (49X,E14.6)

CP is the specific heat of the fluid in $\text{ft}^2/\text{sec}^2 \text{ } ^\circ\text{R}$.

CARD 31. AMUINF (49X,E14.6)

AMUINF is the coefficient of viscosity in slugs/ft-sec.

CARD 32. PSTAG (49X,E14.6)

PSTAG is the free-stream total pressure in lbf/ft^2 .

CARD 33. PINF (49X,E14.6)

PINF is the free-stream static pressure in lbf/ft^2 .

CARD 34. XBAR (49X,E14.6)

XBAR is the relative length of the transition regime in turbulent cases. It is the ratio of the transition end point location to the transition onset distance.

CARD 35. UFS (49X,E14.6)

UFS is the free-stream velocity in ft/sec.

CARD 36 to 36+NSOLVE XSTA(I), I = 1, NSOLVE (F12.6)

XSTA is an important input array. It is an array of surface distances

solutions calculated. The program will always obtain solutions at these points regardless of internal adjustments to the stream-wise step size. Both the value 0.0 and the end point of the body must be included in the array as well as any distances describing the beginning or end of injection and transition.

CARD 36+NSOLVE+1 TO LAST CARD XTW(I), TWX(I), XCI(I), CIX(I)
(4E12.6)

TWX(I) is the wall temperature in degrees Rankine at XTW(I) which is a surface distance.

CIX(I) is the injection rate v_w/u_∞ at XCI(I) which is a surface distance.

These arrays are dimensioned for a maximum of 500 values each. If none of these cards appear in the input deck the program will automatically assume constant wall temperature and injection rate values based on RTW and CWALL. This input allows the wall temperature distribution and injection rate distribution to be read in versus their own surface distance tables. If both distributions are to be read in versus the same distance table, then either one of the two distance tables may be left blank. Another important feature is the fact that the distributions need not cover the same surface distance. For instance, the wall temperature distribution might be defined over the entire body while the injection rate distribution might only be defined over a short distance.

3. Description of Output Data

The output from program ICBL3D is in the form of printed output. The printed output is presented in two forms,

station data and profile data. This section will define the output by listing the variable name as it appears in the output along with the definition of the variable. The program also prints miscellaneous messages which are described at the end of this section.

Station Data

Line 1

X0	distance from the stagnation point along the body axis non-dimensionalized by the reference length (one foot)
R	radial distance from the stagnation point body axis to the body surface non-dimensionalized by the reference length
PHI	meridional angle about the stagnation point body axis measured from the windward symmetry plane in degrees

Line 2

XI	longitudinal surface coordinate ξ
DXI	$\Delta\xi$, streamwise step size
CWALL	local injection rate v_w/u_∞
NIT	number of iterations to obtain the solution

Line 3

HX	metric coefficient h_ξ evaluated at the body
HW	metric coefficient h_w evaluated at the body

Line 4

TE	nondimensional edge static temperature, $T_\infty C_p / u_\infty^2 = T_e$
----	--

UE nondimensional ξ component of velocity
at the boundary-layer edge $u_e/u_\infty = U_e$

VE nondimensional ω component of velocity
at the boundary-layer edge $w_e/u_\infty = W_e$

Line 5

DUEDX $\partial U_e / \partial \xi$

DVEDX $\partial W_e / \partial \xi$

DUEDW $\partial U_e / \partial \omega$

DVEDW $\partial W_e / \partial \omega$

Line 6

LOCAL EDGE REYNOLDS NUMBER = $\rho_\infty u_e \xi / \mu_\infty$

Line 7

CFXINF $C_{f_{\xi_\infty}} = 2 \tau_{w_\xi} / \rho_\infty u_\infty^2$, streamwise skin friction coefficient

CFXEDG $C_{f_{\xi_e}} = 2 \tau_{w_\xi} / \rho_\infty u_e^2$, streamwise skin friction coefficient based on edge conditions

CFWINF $C_{f_{\omega_\infty}} = 2 \tau_{w_\omega} / \rho_\infty u_\infty^2$, transverse skin friction coefficient based on free-conditions

CFWEDG $C_{f_{\omega_e}} = 2 \tau_{w_\omega} / \rho_\infty u_e^2$, transverse skin friction coefficient based on edge conditions.

Line 8

QW $q_w / \rho_\infty u_\infty^3$, wall heat transfer coefficient

CHIMAX maximum vorticity Reynolds number $n^2 / \nu \partial u / \partial n$

Line 9

LONGITUDINAL SKIN FRICTION τ_{w_ξ} in lbf/ft^2

DELTA*(X) δ_ξ^* , streamwise boundary-layer displacement thickness in feet.

THETA(X) θ_ξ , streamwise boundary-layer momentum thickness in feet.

Line 10

TRANSVERSE SKIN FRICTION τ_{w_ω} , in lbf/ft²

DELTA*(PHI) δ_ω^* , transverse boundary-layer displacement thickness in feet.

THETA(PHI) θ_ω , transverse boundary-layer momentum thickness in feet.

Line 11

WALL HEAT TRANSFER RATE q_ω in Btu/ft²/sec

DELTA(FT) δ , the boundary-layer thickness in feet.

Profile Data

One group of boundary layer profile data is printed by the program. Every other point is printed in the profile arrays.

The following columns are identified:

ETA η , the transformed normal coordinate.

Y n , the physical normal distance nondimensionalized by the reference length.

F $h_\xi u / h_{\xi,0} U_e$

FN $\partial F / \partial \eta$

G $h_\omega w / h_{\omega,0} W$ where $W = W_e$ at the stagnation point and symmetry planes and $W = U_e$ elsewhere.

GN $\partial G / \partial \eta$

T T / T_e

TN $\partial (T / T_e) / \partial \eta$

V transformed normal velocity profile.

EPLUS ϵ^+ the eddy viscosity coefficient

Miscellaneous Messages

The program has a few internal messages which are written to indicate problems with the solution, or coordinate adjustments. A message is printed by subroutine ADDETA whenever η_∞ is adjusted up or down. The direction of adjustment is given along with X, the old η_∞ and the new η_∞ .

A message is printed by subroutine CHANGX indicating the beginning of transition or mass transfer. Included in the messages are the values of X and the particular integer counter involved. A similar message is also printed by CHANGX when mass transfer ends.

Whenever the program fails to obtain a converged solution within NIT3 iterations, a message is printed by subroutine CONTRL to that effect which includes the values of the transverse and streamwise solution counters and NIT. If this should occur three consecutive times, a message will be printed indicating that execution is terminating.

If a particular boundary-layer problem drops all of its circumferential solution planes due to convergence problems, a message will be printed by CONTRL indicating that execution is terminating.

A normal termination of the program is indicated by the message "THE END" printed out after the last station results.

3. Sample Input/Output

Listings of Job Control Language, sample input decks, sample outputs and format sheets for the input data cards and the program listings are found in Appendices VI through X.

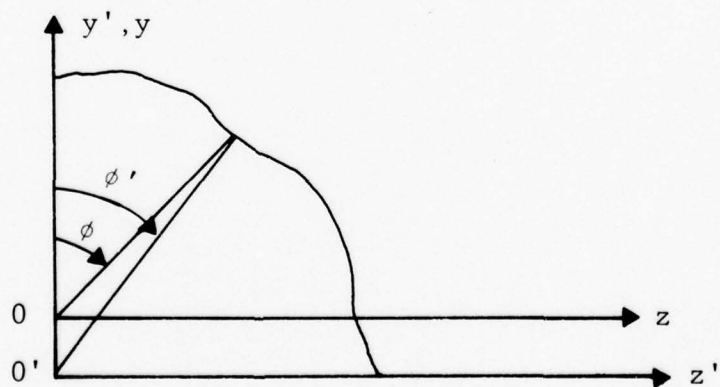
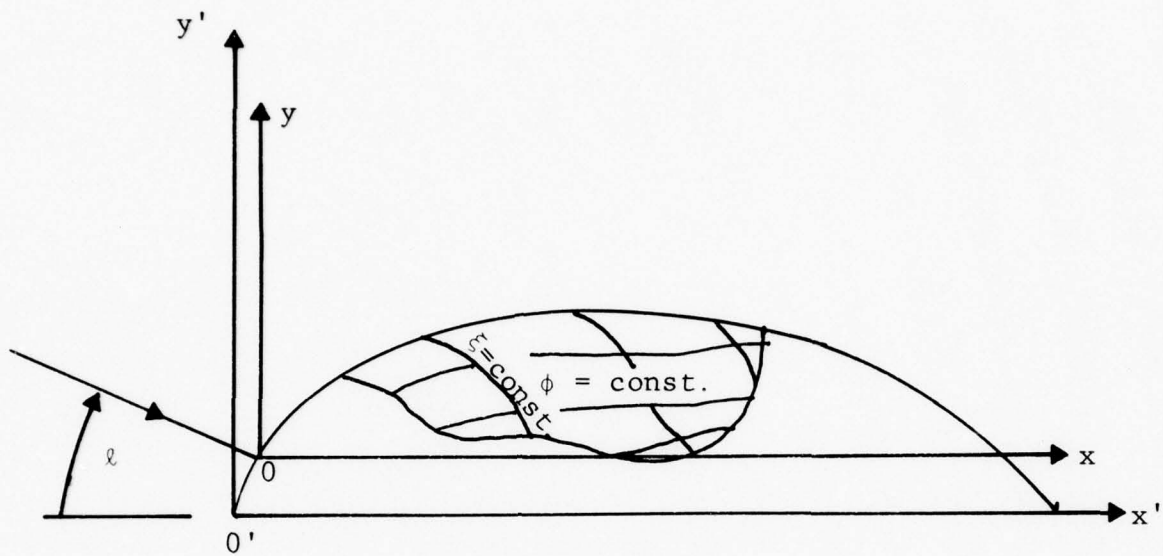


Figure 1. BODY COORDINATE SYSTEMS FOR INVISCID AND BOUNDARY-LAYER CALCULATIONS

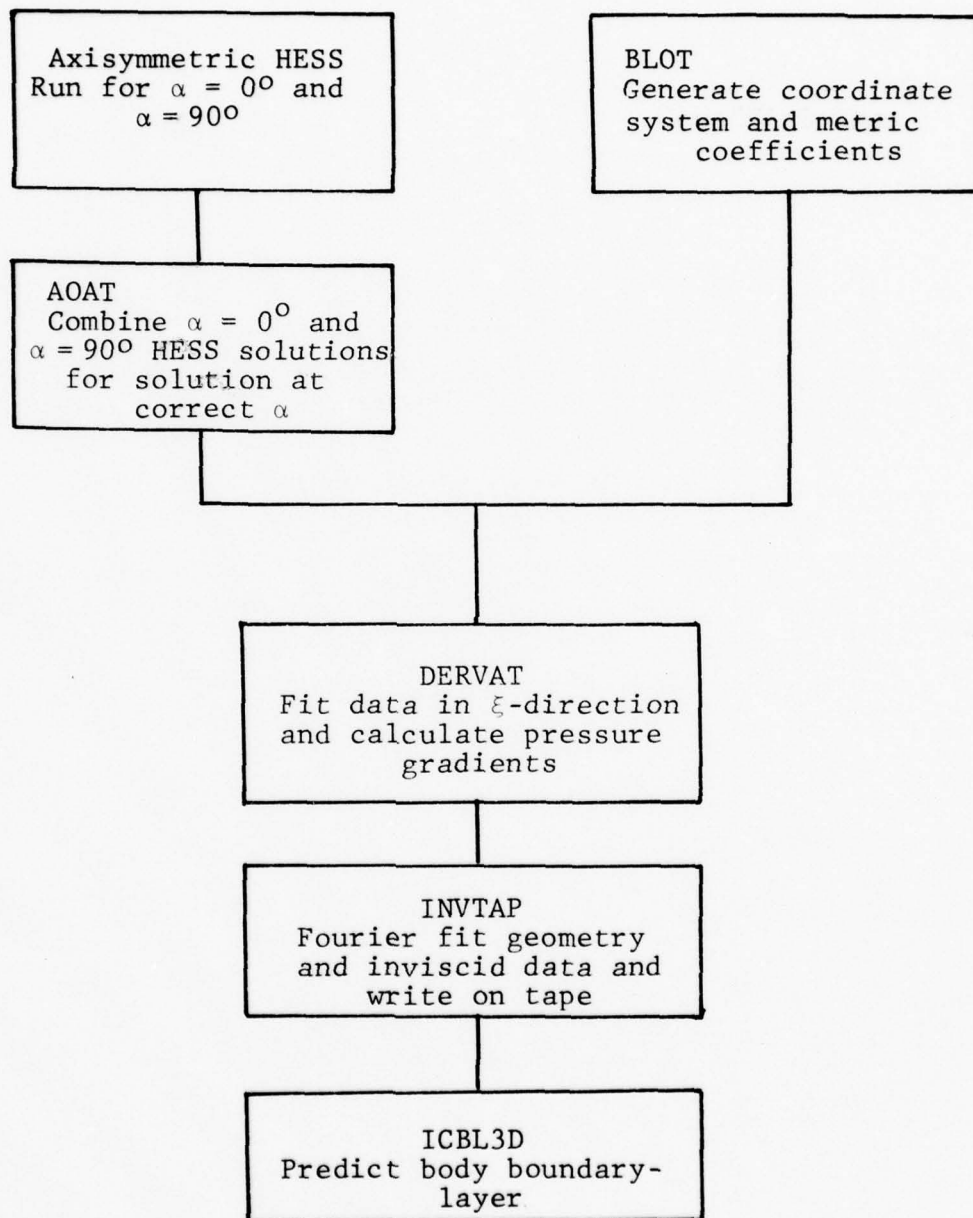


Figure 2a. BLOCK DIAGRAM FOR ICBL3D AXISYMMETRIC BODIES

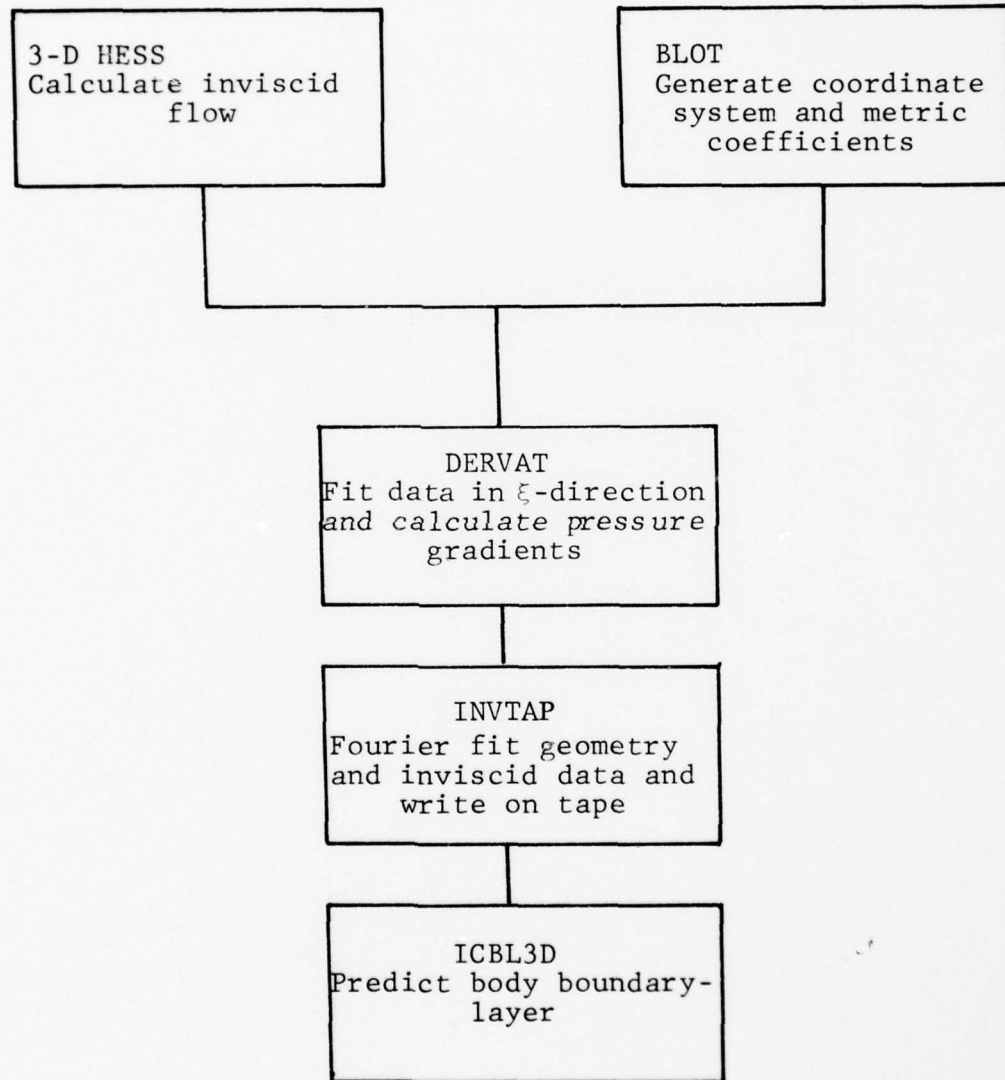


Figure 2b. BLOCK DIAGRAM FOR ICBL3D NONAXISYMMETRIC BODIES

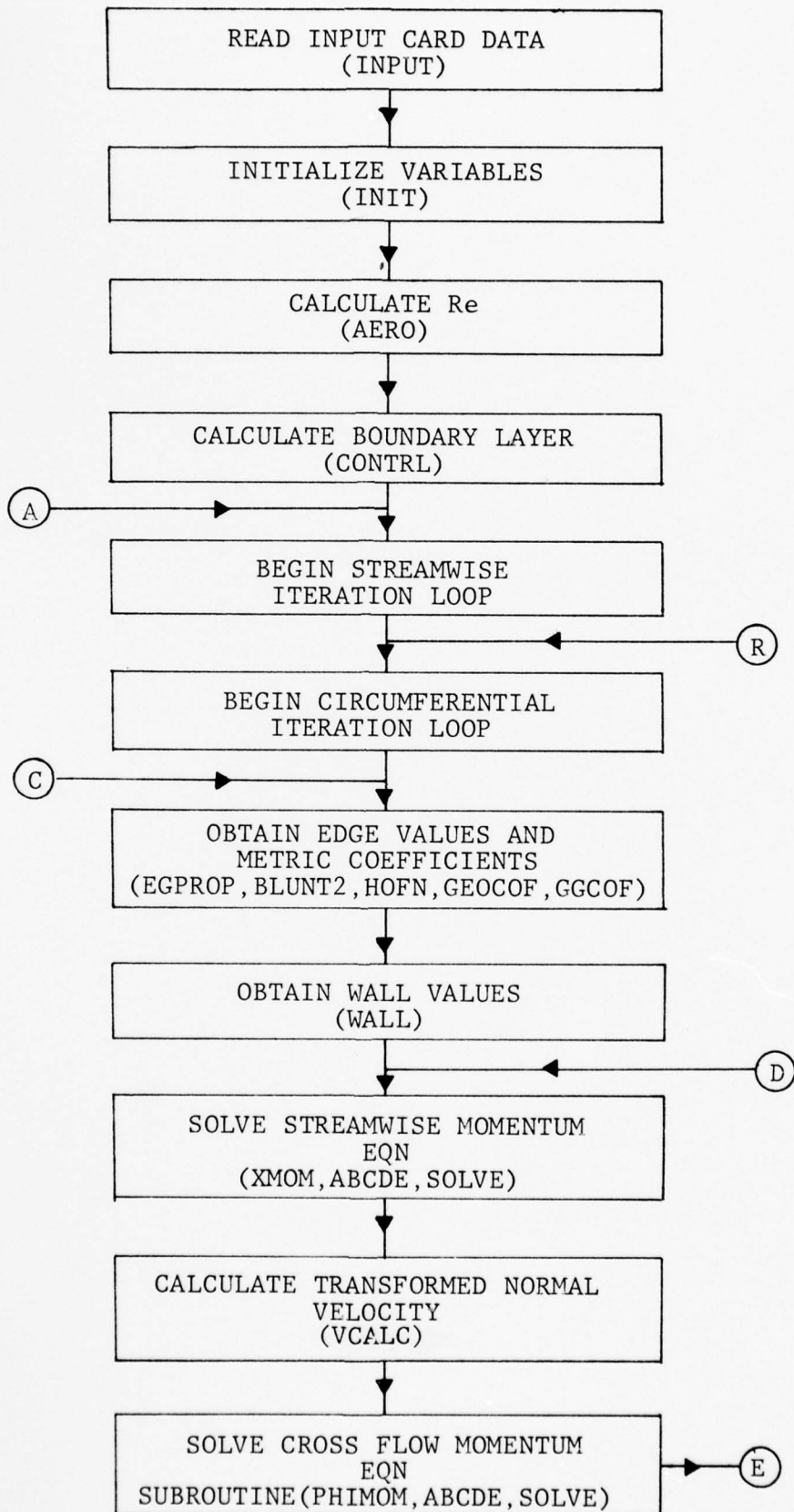


Figure 3. FLOW CHART OF PROGRAM ICBL3D

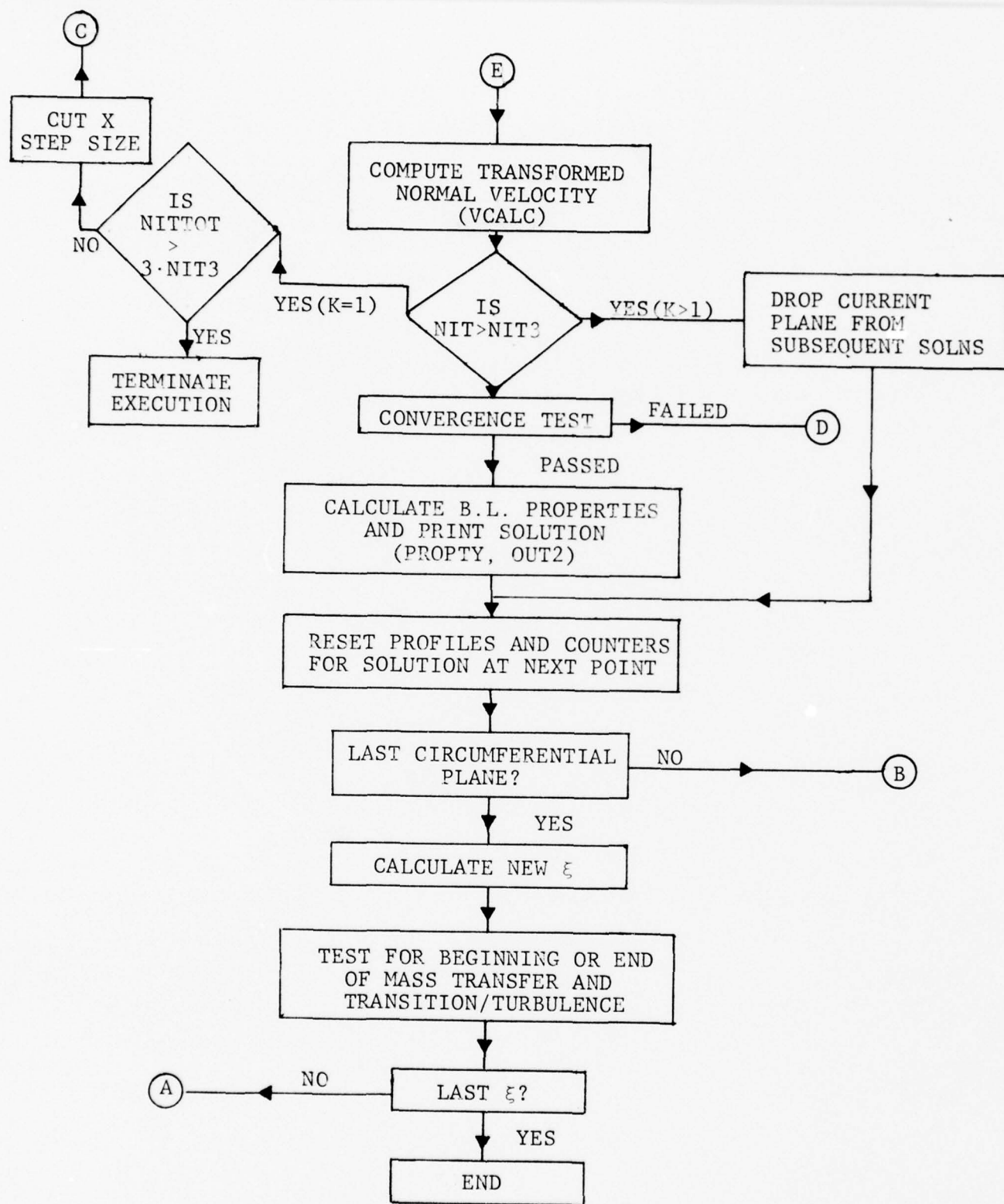


Figure 3 continued: FLOW CHART OF PROGRAM ICBL3D
() = Subroutine Names

REFERENCES

1. Dwoyer, D. L., Lewis, C. H. and Gogineni, P. R., "Three-Dimensional Incompressible Boundary Layers on Blunt Bodies Including Effects of Turbulence, Surface Curvature and Heat and Mass Transfer. Part I: Analysis and Results." VPI&SU Aero-063, May 1977.
2. Blottner, F. G. and Ellis, M., "Three-Dimensional Incompressible Boundary Layer on Blunt Bodies." Sandia Laboratories Report No. SLA-73-0366, Albuquerque, New Mexico, April 1973.
3. Hess, J. L. and Martin, R. P., Jr., "Improved Solution for Potential Flow about Arbitrary Axisymmetric Bodies by the Use of a Higher-Order Surface Source Method. Part I. Theory and Results." NASA Contractor Report No. NASA CR 134694, NASA Lewis Research Center, Cleveland, Ohio, July 1974.
4. Hess, J. L. and Smith, A.M.O., "Calculation of Non-Lifting Potential Flow about Arbitrary Three-Dimensional Bodies." March 1962, McDonnell-Douglas Report No. ES 40622.

APPENDIX I

Job Control Language for Program TAPGEN

DATE 05/17/77

```
// JOB CARD
/*PRIORITY URGENT
/*JOBPARM LINES=10,CARDS=0
//STEP1 EXEC FORTGCG,TIME=4,REGION=520K,PARM.FORT='NOSOURCE'
//FORT.SYSIN DD *
```

PROGRAM HESS FORTRAN SOURCE DECK

```
//GO.FT05F001 DD *
```

INPUT DATA DECK FOR PROGRAM HESS

```
//FT06F001 DD SYSOUT=A
//FT07F001 DD DUMMY
//FT01F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT02F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT03F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT04F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT08F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT09F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT10F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT11F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT12F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT13F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT15F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT16F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE=(TRK,(50,1)),DSN=&&T1
/*
//STEP2 EXEC FORTGCG,COND=(0,NE),TIME=1,REGION=300K
//FORT.SYSIN DD *
```

PROGRAM AOAT FORTRAN SOURCE DECK

```
//GO.FT05F001 DD *
```

INPUT DATA DECK FOR PROGRAM AOAT

```
//FT06F001 DD DUMMY
//FT16F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T1
//FT22F001 DD DUMMY
//FT30F001 DD DUMMY
//FT10F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE=(TRK,(50,1)),DSN=&&T2
/*
//STEP3 EXEC FORTGCG,COND=(0,NE),TIME=10,REGION=500K
//FORT.SYSIN DD *
```

PROGRAM BLOT FORTRAN SOURCE DECK

```
//GO.FT05F001 DD DUMMY
```

INPUT DATA DECK FOR PROGRAM BLOT

DATE 05/17/77

```
//FT06F001 DD DUMMY
//FT08F001 DD DUMMY
//FT11F001 DD DUMMY
//FT12F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1))
//FT13F001 DD UNIT=SYSDA,DISP=(NEW,DELETE),SPACE=(TRK,(50,1)),
//          DCB=(RECFM=VS,LRECL=2884,BLKSIZE=2888)
//FT10F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T2
//FT25F001 DD UNIT=SYSDA,DISP=(NEW,CATLG,CATLG),SPACE=(TRK,(50,1)),
//          DSN=AF01F3.T.T4
//FT04F001 DD *
/*
//STEP4 EXEC FORTGCG,TIME=4,REGION=500K
//FORT.SYSIN DD *
```

PROGRAM DERVAT FORTRAN SOURCE DECK

```
//FT06F001 DD SYSOUT=A
//FT25F001 DD UNIT=SYSDA,DISP=(OLD,KEEP),DSN=AF01F3.T.T4
//FT26F001 DD UNIT=SYSDA,DISP=(NEW,PASS),SPACE=(TRK,(50,1)),DSN=&&T4
//FT29F001 DD SYSOUT=A
//FT30F001 DD SYSOUT=A
//STEP5 EXEC FORTGCG,COND=(0,NE),TIME=4,REGION=300K
//FORT.SYSIN DD *
```

PROGRAM INVTAP FORTRAN SOURCE DECK

```
//GO.FT05F001 DD DUMMY
//FT06F001 DD SYSOUT=A
//FT10F001 DD UNIT=SYSDA,DISP=(NEW,CATLG),SPACE=(TRK,(50),RLSE),
//          DSN=A505F3.INVTAP.EL412,VOL=SER=USERPK,
//          DCB=(RECFM=VS,LRECL=117,BLKSIZE=121)
//FT25F001 DD UNIT=SYSDA,DISP=(OLD,DELETE),DSN=&&T4
//FT30F001 DD SYSOUT=A
/*
//
```

APPENDIX II

TAPGEN Sample Input

INPUT DATA DECK FOR PROGRAM HESS				120 ELEMENTS	2
ELLIPSOID (4:1) AT				ALPHA= 10 DEG	
1011001	00	00	0		
01001	121	1.0	0	1.0	0.25
	1			1.0	

INPUT DATA DECK FOR PROGRAM BLOT									
TITLE	CASE	7	ELLIPSOID(4:1)	-	10	DEG	ANGLE OF ATTACK		
LIMITS	31	26	01	10	1	0	6.00	100.0	
DM	.02	.02	.02	.02	.02	.02	.02	.02	.02
DM	.02	.02	.02	.02	.02	.02	.02	.02	.02
DM	.02	.02	.02	.02	.02	.02	.02	.02	.02
DN	.3	.3	.3	.3	.3	.3	.3	.3	.3
DN	.3	.3	.3	.3	.3	.3	.3	.3	.3
DN	.3	.3	.3	.3	.3	.3	.3	.3	.3
DN	.3	.3	.3	.3	.3	.3	.3	.3	.3
OPSI	.01	.25	.25	.25	.25	.0000001	+10.0	1.0	0.5
SIZE	4.00								

APPENDIX III

TAPGEN Sample Output

INVISICID EDGE CONDITIONS FOR BOUNDARY LAYER SOLUTION
TAKEN FROM THE INVISICID FLOW FIELD DATA

NUMBER OF PLANES IN THE INVISICID DATA = 24

ALPHA= C.O

UNIFORM FLOW STARTING SOLUTION FOR THE INVISICID FLOW FIELD

WALL DATA AT STATION 1, XI = .)

K	HPSI	HW	P	C	UPSI	UW	X	D(UPSI)	D(UW)	D(P)	D2(P)	E(HPSI)	D(HW)	D(X)
1	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
2	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
3	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
4	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
5	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
6	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
7	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
8	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
9	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
10	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
11	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
12	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
13	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
14	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
15	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
16	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
17	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
18	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
19	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
20	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
21	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
22	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
23	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
24	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
25	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0
26	1000000	0	0	0	0	0	0	0	0	0	0	0	0	0

APPENDIX IV

Program TAPGEN Data FORMAT Sheets

IBM

FORTTRAN Coding Form

PROGRAM	HESS	PAGE	CF
PROGRAMMER		CARD ELECTRIC NUMBER	

CARD 2 - CONTROL FLAG CARD		FORTTRAN STATEMENT	
STATEMENT NUMBER	STATEMENT	STATEMENT NUMBER	STATEMENT
1	(711, 3X, 211, 1X, 1011, 4X, 11, 12, 411)	1	
2	NB	2	
3	NNU	3	
4	IAXI	4	
5	ICROSS	5	
6	IØFF	6	
7	IØNLY	7	
8	IØLPSE	8	
9	IPØTRB	9	
10	IPØTNL*	10	
11	IPØTANV	11	
12	IPØRT	12	
13	IØMITA	13	
14	IØMITC	14	
15	ISURFV	15	
16	IPRSCV	16	
17	IALLV	17	
18	IEXCRS*	18	
19		19	
20		20	
21		21	
22		22	
23		23	
24		24	
25		25	
26		26	
27		27	
28		28	
29		29	
30		30	
31		31	
32		32	
33		33	
34		34	
35		35	
36		36	
37		37	
38		38	
39		39	
40		40	
41		41	
42		42	
43		43	
44		44	
45		45	
46		46	
47		47	
48		48	
49		49	
50		50	
51		51	
52		52	
53		53	
54		54	
55		55	
56		56	
57		57	
58		58	
59		59	
60		60	
61		61	
62		62	
63		63	
64		64	
65		65	
66		66	
67		67	
68		68	
69		69	
70		70	
71		71	
72		72	
73		73	
74		74	
75		75	
76		76	
77		77	
78		78	
79		79	
80		80	
81		81	
82		82	
83		83	
84		84	
85		85	
86		86	
87		87	
88		88	
89		89	
90		90	
91		91	
92		92	
93		93	
94		94	
95		95	
96		96	
97		97	
98		98	
99		99	
100		100	

* Available if and only if NØNEWF = 1, ISIGF = 2, IGEØMF = 1.

FORTRAN Coding Form

HES

[illegible]

CARD	2	IGEN BC	GENERATED	BOUNDARY	CONDITIONS
		IRNGW	RING WING	OPTION	
		IPNCH	PUNCHED	OUTPUT	
		IUNIT	UNIT	NUMBER FOR INPUT	COORDINATES (default = 05)
		IVIJ	MATRIX	PRINT FLAG	
		ICDEF	MATRIX-ASSEMBLY	COEFFICIENT	PRINT FLAG
		IPRINT	VERY DETAILED	MATRIX CONSTRUCTION	PRINT FLAG
		IRAKF	AUTOMATIC	RAKE GENERATION	FLAG

IBM

FORTRAN Coding Form

GX28-7327-6 U/M 050**
Printed in U.S.A.

PROGRAM	HESS	DATE	PAGE	OF	CHARACTER NUMBER
PROGRAM					

FORTRAN STATEMENT		CHARACTER NUMBER
CARD 3 - CHORD/MACH NUMBER CARD *		
CHORD	XMACH	
2F10.6		
CHORD	REFERENCE CHORD LENGTH (DEFAULT = 1.0)	
XMACH	MACH NUMBER FOR GOETHERT CORRECTION (0.0 IMPLIES INCOMPRESSIBLE)	
* THIS CARD USUALLY LEFT BLANK		

Number of times per day eating

IBM

GX28-7327-6 U/M 050**
Printed in U.S.A.

FORTRAN Coding Form

PROGRAMMER	HESS	DATE		PUNCHING INSTRUCTIONS	GRAPHIC PUNCH	PAGE OF	CARD ELECTRIC PUNCH
------------	------	------	--	-----------------------	---------------	---------	---------------------

CARD 5 BODY CONTROL CARD 2 OF 2		FORTRAN STATEMENT		IDENTIFICATION
IBDN	ISUBKS	NLF	A	B
		(3110, 2F10.3)		
IBDN	"BODY" NUMBER (SEQUENTIAL FOR BODIES, ZERO FOR OFF BODY POINTS)			
ISUBKS	SUBCASE FLAG			
NLF	NON-LIFTING FLAG (FOR COMBINATION CASES, ONLY)			
A	SEMI-MAJOR AXIS FOR ELLIPSE CASES } IF IELPSE#0			
B	SEMI-MINOR AXIS FOR ELLIPSE CASES }			

AOAT

ENDING

CPA 6-12

PAGE OF

FORTRAN STATEMENT

CONCLUSION

DATA SET

CARD	1
------	---

IPRT N KPL ALPD

(313.F10.6)

IPRT	= 1	WRITES THE OUTPUT OF PROGRAM HESS ON UNIT 6 ;	= 0, DOES NOT WRITE
N		NUMBER OF ELEMENTS USED IN THE PROGRAM HESS	
KPL		NUMBER OF MERIDIONAL PLANES REQUIRED	
ALPD		ANGLE OF ATTACK IN DEGREES	

IBM

THREE-DIMENSIONAL, INCOMPRESSIBLE BOUNDARY LAYER ON BLUNT BODIES,
PART II: COMPUTER CODE - USER'S GUIDE, Molly A. Ellis, F. G. Blottner
RE: SLA-73-0704

FORTTRAN Coding Form

GY28-7327-6 U.M. 0501
Printed in U.S.A.

PROBLEM IDENTIFICATION	PROBLEM NUMBER	1
INSTRUCTIONS	DATE	

TITLE		PROBLEM IDENTIFICATION									
1	TITLE										(5X, 9A8)
2	LIMITS	JMAX	KMAX	LITER	IPRT	KPRT	PSIMAX	UINF			(5X, 515, 2E10.0)
3	Δw's AROUND THE BODY										(5X, 7E10.0)
4	Δη's ACROSS THE BOUNDARY LAYER										(5X, 7E10.0)
5	Δξ - STEPSIZE TO BE USED ALONG THE BODY										(5X, E10.0)
6	SIZE	A	RB	RC	EPS	ALP	CAPTH	CRI			(5X, 7E10.0)

JMAX		KMAX		LITER		IPRT		KPRT		PSIMAX		UINF	
No. of pts. across boundary layer		No. of pts. around body (in phi-direction)		No. of iterations for initial profile		Every how many steps printing will occur along the body		Every how many steps printing will occur around the body		Distance along body at which program will terminate		Freestream velocity - V_∞	
A		RB		RC		EPS		ALP		CAPTH		CRI	
a - length of axis of triaxial ellipsoid		Rb - radius of curvature of nose; plane y'=0		Rc - radius of curvature of nose; plane z'=0		Convergence criterion; body coordinates		α - angle of attack (deg)		θ - indicates finite-difference scheme used		1 Implicit scheme, 1/2 Crank-Nicolson scheme	

APPENDIX V

TAPGEN Program Listings

1. Program BLOT Listing

```

C0000010
C0000020
C0000030
C0000040
C0000050
C0000060
C0000070
C0000080
C0000090
C0000100
C0000110
C0000120
C0000130
C0000140
C0000150
C0000160
C0000170
C0000180
C0000190
C0000200

PROGRAM BL3D (INPUT,OUTPUT,TAPE60=INPUT,TAPE61=OUTPUT)
      BL3D IS THE MAIN PROGRAM AND CALLS SEVERAL OTHER SUBROUTINES
      IMPLICIT REAL*8 (A-H,O-Z)
      COMMON /COL/ A , EPS
1     JMAX2 , KMAX , JMAXI
2     PTIMAX , SO , KMAXI,JK , PI
3     WINF , XO , UINF , YD
      COMMON /COB/ IN , IOUT
      CALL ERRSET(207,256,10,0,0,209)
      PI = 3.1415926536
      IN = 5
      IOUT = 6
      CONTINUE
      CALL INPUT
      CALL PRECAL
      CALL GEOM
      GO TO 10
      END

```

```

SUBROUTINE CALCUV (K,CS,SN,U,IMET,V)
      CALCUV COMPUTES EITHER R OR X, DEPENDING UPON WHICH IS THE
      ** ARGUMENTS **
      K - SUBSCRIPT IN PHI-DIRECTION.
      CS - COS OF ANGLE UPON WHICH R OR X DEPENDS.
      SN - SIN OF ANGLE UPON WHICH R OR X DEPENDS.
      U - THE UNKNOWN - R OR X.
      IMET - 2, IMPLIES SOLVING FOR X.
           - 1, IMPLIES SOLVING FOR R.
      V - THE KNOWN VARIABLE - X OR R.

      IMPLICIT REAL *8 (A-H,O-Z)
      COMMON/CO30/ IA, JN2, OTA, PRX, PRXOR
      1
      DIMENSION V(100)
      IF (IMET - NE. 1) GO TO 20
      CONTINUE
      XK = V(K)
      CALL CXPOVA (XK)
      U = RELTP (CS,SN,K)
      GO TO 60
      CONTINUE
      GO TO (30,40), IA
      CONTINUE
      U = ELLSP(CS,SN,K)

```


BL000470
BL000480
BL000490
BL000500
BL000510
BL000520
BL000530
BL000540

BL000550
BL000560
BL000570
BL000580
BL000590
BL000600
BL000610
BL000620
BL000630
BL000640

[illegible]

K - SUBSCRIPT IN THE PHI-DIRECTION.
K - INCREMENTAL DISTANCE ALONG SURFACE OF THE BODY.

DIMENSION MET(2)
IF (K.NE.0) GO TO 60

```

10 DS = DPST
   DS = DS / DSQRT(1.000+UV(1)*UV(1))
   V(1) = VOLD(1) + DV
   CONTINUE CASE, K (NOTI)=0.
   CALL CALCV (1,COSPH(1),SINPH(1),US,[MET,V)
   UV(1) = US
   DV(1) = (U(1) - VOLD(1)) / DV
   VORG = V(1)*DSQRT(1.000 + UV(1)*UV(1))
   V(1) = V(1) + DV
   IF (V(1).EQ.VORG) GO TO 20
   IF (DABS(V(1)-VORG).GT.DABS(V(1)*EPS)) GO TO 10
   CONTINUE CASE, K (NOTI)=0.
   CALL CALCV (1,COSPH(1),SINPH(1),US,[MET,V)
   UV(1) = US
   GO TO 130,*0), IMET
   CONTINUE
   IF (IBEAT.EQ.0) GO TO 31
   CALL OPRX(COSPH(1),U(1),PRX)
   GO TO 32
31 PRX = RB*(1.000-XPOVA) / (U(1)*OTA+YO*CO$PH(1))
32 PRX = (YO+U(1)*COSPH(1)*(1.000-RB/RC))*SINPH(1)/(U(1)*OTA +
   1 YO*CO$PH(1) + RB*PRX*(1.000-XPOVA))
   GO TO 50
   CONTINUE
   IF (IBEAT.EQ.0) GO TO 41
   CALL OPRX(COSPH(1),V(1),PRX)
   GO TO 42
41 PRX = RB*(1.000-XPOVA)/(V(1)*OTA + YO*CO$PH(1))
42 PRX = (YO+V(1)*COSPH(1)*(1.000-RB/RC))*SINPH(1)/(V(1)*OTA
   1 +YO*CO$PH(1) + RB*PRX*(1.000-XPOVA))
   CONTINUE
   HPST(1) = 1.000
   DV = 0.000
   MET(1) = 0
   MET(2) = 0
   RETURN
   GENERAL CASE, K (NOTI)=0.
C
60 CONTINUE
   V(K+1) = V(K) + DV
   MET(1,MET) = MET(1,MET) + 1
   COSPH3 = COS(PI(K+1) + 0.500*DPHI(K))
   SINPH3 = DSIN(PI(K+1) - 0.500*DPHI(K))
   SINPH4 = DSIN(PI(K+1) + 0.500*DPHI(K))
   SINPH4 = DSIN(PI(K+1) - 0.500*DPHI(K))
   IF ((MET(1).NE.1).AND.(MET(2).NE.1)) GO TO 110
   CONTINUE
   DCO$PH(1) = 0.500*DPHI(K)
   COSPH1 = DCO$PH(1) - 0.500*DPHI(K)
   SINPH1 = DSIN(PI(K) + 0.500*DPHI(K))
   SINPH2 = DSIN(PI(K) - 0.500*DPHI(K))
70

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```

CALL CALCV(K,CSPH1,SINPH1,US,IMET,V)
U1 = US
CALL CALCV (K,CSPH2,SINPH2,US,IMET,V)
U2 = US
GO TO (80,90), IMET
CONTINUE
IF (IBEX1.EQ.0) GO TO 81
CALL DPRX(COSPH(K),U(K),PRX)
UV(K)=PRX
GO TO 100
81 UV(K) = RB * (1.000-XPOVA) / (U(K)*OTA+YO*COSPH(K))
GO TO 100
CONTINUE
IF (IBEX1.EQ.0) GO TO 91
CALL DPRX(COSPH(K),V(K),PRX)
UV(K)=PRX
GO TO 100
91 UV(K) = (V(K)*OTA+YO*COSPH(K)) / (RB*(1.000-XPOVA))
GO TO 120
TERM1 = (U1 - U2) / DPHI(K)
CONTINUE
TERM1 = TERM2
CONTINUE
CALL CALCV (K+1,CSPH3,SINPH3,US,IMET,V)
U3 = US
CALL CALCV(K+1,CSPH4,SINPH4,US,IMET,V)
U4 = US
CALL CALCV (K+1,COSPH(K+1),SINPH(K+1),US,IMET,V)
UV(K+1) = US
GO TO (140,150), IMET
CONTINUE
IF (IBEX1.EQ.0) GO TO 141
CALL DPRX(COSPH(K+1),U(K+1),PRX)
UV(K+1)=PRX
GO TO 160
141 UV(K+1) = RB*(1.-XPOVA)/(U(K+1)*OTA+YO*COSPH(K+1))
GO TO 160
CONTINUE
IF (IBEX1.EQ.0) GO TO 151
CALL DPRX(COSPH(K+1),V(K+1),PRX)
UV(K+1)=PRX
GO TO 160
151 UV(K+1) = (V(K+1)*OTA + YO*COSPH(K+1)) / (RB*(1.000-XPOVA))
CONTINUE
DUDV = 0.500*(UV(K) + UV(K+1))
TERM2 = (U3 - U4) / DPHI(K)
DUDPH = 0.500 * (TERM1 + TERM2)
DV = -PI2*DUDPH*DUDV / (1.000+DUDV *DUDV ) *DW(K)
VORG = V(K+1)

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```

170      V(K+1) = V(K) + DV
      IF (V(K+1) - ED - VORG) GO TO 170
      CONTINUE
      HMO(K) = HMO(K)
      IF (HMO(K) - HMO(K)) GO TO 190
      CONTINUE
      HMO(K) = DSORT(U(K+1) - U(K))**2 + (PI*(U(K+1) + U(K))**2) * DW(K)**2
      PRX = U(K+1) - V(K)**2 / DW(K)
      PROR = (YD(U(K+1)) * COSPH(K+1)) * (1.000 - RB/RC)) * SINPH(K+1) / (U(K+1) * OTA)
      YD = (COSPH(K+1) * RB * PRX * (1.000 - XPOVA))
      GO TO 200
      CONTINUE
      HMO(K) = DSORT(U(K+1) - U(K))**2 + (PI*(V(K+1) + V(K))**2) * DW(K)**2
      PRX = (V(K+1) - V(K))**2 / DW(K)
      YD = (COSPH(K+1) * RB * PRX * (1.000 - XPOVA))
      CONTINUE
      DELV = V(K+1) - VOLD(K+1)
      UVH = (U(K+1) - UOLD(K+1)) / DELV
      DS = DSORT(1.000 + UVH * UVH) * DELV
      HPSI(K+1) = DS / DPSI
      RETURN
      END
200

```

```

SUBROUTINE CXPOVA (X)
      CXPOVA CALCULATES THE RELATION XPRIME/A.
      **ARGUMENTS **
      X - X-COORDINATE.
      IMPLICIT REAL*8 (A-H,O-Z)
      COMMON /COI/ A, JMAX2, JMAX, EPS, JMAX1, JK, PI, UINF, YD, PRX, PROR
      1 COMMON /CO30/ XPOVA, IA, JK2, OTA
      GO TO (10,20), IA
      CONTINUE
      XPOVA = (X + XO) / A
      GO TO 30
      CONTINUE
      XPOVA = 0.000
      CONTINUE
      RETURN
10
20
30

```

BL002410

```

C      DOUBLE PRECISION FUNCTION ELLSP(COSP,SINP,K)
C
C      THIS FUNCTION DEFINES X WHEN THE BODY GEOMETRY IS AN ELLIPSOID.
C      ** ARGUMENTS **
C      COSP - COS OF THE ANGLE.
C      SINP - SIN OF THE ANGLE.
C      K - SUBSCRIPT IN THE PHI DIRECTION.
C
C      IMPLICIT REAL*8 (A-H,O-Z)
C      COMMON /COL/ A , EPS
C      1 COMMON /COL/ JMAX2 , KMAX , JMAX , JMAX1
C      2 COMMON /COL/ P1MAX , SO , PI
C      3 COMMON /COL/ W1NF , X0 , U1NF , U1NF
C      COMMON /COL/ ALPHA(100), DNI(50) , DN2(50) , R(100)
C      ELLSP = 0.0
C      DO 10 I = 1, 100
C      10 ELLSP = ELLSP + ALPHA(I)*R(I)
C      RETURN
C      END

```


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```

1 COMMON/C023/ RB , WRTX(100), S(100), M(100), X(100)
1 COMMON/C030/ RB , AO , BO , CO , P
1 COMMON/C030/ IA, JK2 , RC , OTA , PRX , PRXOR
1 COMMON /EXT/ XPOVA
DIMENSION ZSLO(2)
OTA = COSP*COSP + RB/RC*SINP*SINP
IF(1BEXT.EQ.1) GO TO 10
RP2 = R(K)*R(K) + 2.0D0*YO * R(K)*COSP + YO*YO
D = RB *RP2/(R(K)*R(K)*OTA + 2.0D0*YO*R(K)*COSP + YO*YO)
ELLSP = A - XO - DSQRT(A*A-A*RP2/D)
GO TO 20
10 TAMP=SINP/COSP
PHI=ATAN(TAMP)
PP=PHI
ZSLO(1)=R(K)
CALL SLOALL(PP,ZSLO,1,13,2,0,0,3,0)
ELLSP=ZSLO(2)
20 RETURN
END

```

-79-

SUBROUTINE INPUT

INPUT READS INPUT DATA AND INITIALIZES A FEW PARAMETERS.

```

IMPLICIT REAL*8 (A-H,O-Z)
COMMON /COL/ A , EPS , JMAX , JMAX1 , JMAX1
COMMON /COL/ PSTMAX , KMAX , KMAX1 , JK , UINF , PI
COMMON /COL/ WINE , SO , SI , YD , DN2(50) , R(100)
1 COMMON /C02/ WRTX(100), S(100), X(100)
1 COMMON /C04/ WRTX(100), S(100), W(100), DPH(100), PSI
1 COMMON /C04/ DM(100), PH(100), PSI , PSIO
2 COMMON /C08/ SINPH(100), SINTHB
COMMON /C08/ LUT , LUT
COMMON /C08/ LUTER,JK1 , PZ , SNTHB2
COMMON /C08/ IPRT , KPT , BO , CO , P
COMMON /C023/ RB , RC , CPTHM , CPTH , CPTH2
COMMON /C025/ RB , CAPTH , CPTHM , KHALF
COMMON /C041/ CRI
COMMON /EXT/ IBEXT
DIMENSION TITLE(9)
READ (4,9000,END=10) (TITLE(I),I=1,9)
GO TO 20

```

BL003030
BL003040
BL003050
BL003060
BL003070
BL003080
BL003090
BL003100
BL003110
BL003120
BL003130
BL003140
BL003150
BL003160
BL003170
BL003180
BL003190
BL003200
BL003210
BL003220
BL003230
BL003240
BL003250
BL003260
BL003270

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```

C      IF (EOF,60) 10,20
10     CONTINUE
      WRITE(6,11) PSI,PSIMAX
      FORMAT(11//LOX,PSI=' ',F10.4,5X,'GREATER THAN PSIMAX=',F10.4,LOX,
11     '1,50 THE END.')
      STOP
20     CONTINUE
      WRITE(10UT,9900) (TITLE(I),I=1,9)
      READ(4,9010) JMAX, KMAX, LITER, IPRT, KPRT, IBEXT, PSIMAX, UINF
      KMAX1 = JMAX - 1
      JMAX1 = JMAX - 1
      READ(4,9020) (DW(K),K=1,KMAX1)
      READ(4,9020) (DN(J),J=1,JMAX1)
      READ(4,9020) DPSI
      READ(4,9020) A, RB, RC, EPS, ALP, CAPTH, CRI
      KHALF = KMAX/2 + 1
      IF (MOD(KMAX,2) .NE. 0) KHALF = KHALF + 1

C      RETURN
C
C      9000 FORMAT (5X,9A8)
C      9900 FORMAT (1H1,20X,9A8)
C      9010 FORMAT (5X,615,3E10.0)
C      9020 FORMAT (5X,7E10.0)
C      END
C      SUBROUTINE PARVAL (K,IP)
C
C      PARVAL - PARAMETER EVALUATION - KEY PARAMETERS NEEDED FOR
C      THE SOLUTION OF F,G, ADD V ARE CALCULATED IN THIS SUBROUTINE.
C      ** ARGUMENTS **
C      K - SUBSCRIPT IN THE PHI-DIRECTION.
C      IP - SUBSCRIPT IN THE PSI-DIRECTION.
C
C      IMPLICIT REAL*8 (A-H,O-Z)
C      REAL*8 KAPPAS, KAPPAW
C      COMMON /COL/ A
C      1 JMAX2 ; KMAX ; EPS ; JMAX ; JMAX1 ; JK ; PI
C      2 PSIMAX ; SO ; XO ; SI ; UINF
C      3 WINF ; YD ; DN2(50) ; R(100)
C      1 COMMON /CO2/ ALPHA(100), S(100) ; M(100) ; X(100)
C      1 COMMON /CO3/ RWRTX(100), F(50,100), G(50,100) ; GO(50,100)
C      1 COMMON /CO3/ UEL(100), UED(100), V(50,100), WE(100)
C      2 WE(100) ; COSPH(100), DPHI(100) ; DPSI
C      1 COMMON /CO4/ ALP ; PHI(100), PSI ; PSIO
C      1 COMMON /CO4/ DW(100) ; SINPH(100) ; SINTR
C      2 COMMON /CO5/ HPSI(100) ; HWO(100) ; ALP2
C      1 COMMON /CO6/ ALBET1 ; ALBET2 ; ALBET3 ; BETAI

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```

1  BETA2 , BETA3 , BETA4 , KAPPAS ,
2  KAPPAW , ALPHA(50) , ALPHA(50) , ALPHA(50) , ALPHA(50) ,
COMMON /C07/ , ALPHA(50) , ALPHA(50) , ALPHA(50) , ALPHA(50) ,
COMMON /C09/ , PI2 , SNTHB2
COMMON /C10/ , PIH , WE1
COMMON /C11/ , HH , SIGNAL
COMMON /C12/ , COSAL , PRXS , PRXORS
COMMON /C13/ , COSPH2 , DM2 , XPOVAS , X2
COMMON /C14/ , R2 , SOLD(100) , THB(100) , XOLD(100) ,
COMMON /C15/ , XWR , RE(100)
COMMON /C18/ , CAPW(100)
COMMON /C19/ , PMOU(100)
COMMON /C20/ , ALBAY , PUOW
COMMON /C21/ , ALB1(100) , ALB2 , ALB3 , BETA5
COMMON /C22/ , BETA6 , BO , CO , P
COMMON /C23/ , RB , AD , RC , PHIPR , RP , UPHP
COMMON /C24/ , USP , X , CPTM , CPTH2
COMMON /C25/ , CAPTH , AL3T
COMMON /C26/ , XKA , WE1AV
COMMON /C29/ , IA , JK2 , OTA , PRX , PRXOR
COMMON /C30/ , XPOVA , UXP , UYP , UZP
COMMON /C31/ , UXP , CV , CW
COMMON /C32/ , CU , ICAP
COMMON /C36/ , HPS , FKWPSI
COMMON /C38/ , ALSTG , WE2
COMMON /C40/ , UER2 , XTST
COMMON /C42/ , DRDSI
COMMON /C43/ , KDUW
COMMON /C50/ , KDUW
KDUW=K
IF (IP .EQ. 1) GO TO 100
UEO(K+1) = UE(K+1)
WEO(K+1) = WEO(K+1)
Y = R(K+1)*COSPHIK(1)
Z = R(K+1)*SINPHIK(1)
CALL UXVZ (Y,Z,XPOVAS)
UE(K+1) = (UXP + (UYP*COSPHIK(1)+UZP*SINPH(K+1))*PRXS)
/DSORT(1,+PRXS*PRXS)
1

```

THE CALCULATION OF UE INVOLVES SETTING UE TO A RADICAL WHICH CAN ASSUME EITHER A (+) OR (-) SIGN. A TEST IS MADE CONCERNING THE SUMS OF TWO PARTIAL DERIVATIVES IN ORDER TO DETERMINE THE APPROPRIATE SIGN FOR UE. IT IS ASSUMED THAT FOR X GREATER THAN OR = XTST THE (+) SIGN SHOULD BE USED.

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```

C
10 IF (X(K+1).GE.XTST) GO TO 10
   IF ((PRXS*DRDST).LT.0.) UE(K+1) = - UE(K+1)
   CONTINUE
   WE(K+1) = ((UYP*COSPH(K+1)+UZP*SINPH(K+1))-UXP*PRXS)*PRXORS
   I + UZP*COSPH(K+1) - UYP*SINPH(K+1)/DSQRT(1.+PRXS*PRXS)
   I * PRXORS*PRXORS)
   IF (K.EQ.KMAX1) GO TO 80
   Y = R2*COSPH2
   Z = R2*SINPH2
   CALL UYXZ(Y,Z,XPOVA)
   UE2 = (UXP + (UYP*COSPH2+UZP*SINPH2)*PRX1/DSQRT(1.+PRX*PRX)
   IF (X2.GE.XTST) GO TO 20
   DRDSIN = (R(K+2) - ROLD(K+2)) / DPSI
   IF ((PRX*DRDSIN).LT.0.) UE2 = -UE2
   CONTINUE
   WE2 = ((UYP*COSPH2+UZP*SINPH2)-UXP*PRX)*PRXOR + UZP*COSPH2-UYP
   I * SINPH2/DSQRT(1.+PRX*PRX)*PRXOR*PRXOR)
   IF (K.NE.0) GO TO 80
   CONTINUE
   CAPW(K+1) = WE(K+1)
   H2 = (HPSI(K+2) - HPSI(K+1)) / (DM(K+1) * DM(K+1))
   WE10 = WE1
   WE1 = WE2/(PI2*DW2)
   WE1AV = .5 * (WE1 + WE10)
   USUBE = .5 * (UEO(K+1) + UE(K+1))
   RBL = .5 * (ROLD(K+1) + R(K+1))
   KAPPAM = (R(K+1) - ROLD(K+1)) / (RBL*DPSI)
   FKPSO = FKPS
   FKPS = KAPPAM * PSI
   DMEL = (WE1 - WE10) / DPSI
   HPS = .5*(PSI + PSIO)
   FKWPST = KAPPAM * HPS
   IF (WE1AV.LE.000000001) GO TO 40
   BETA4 = (HPS*DMEL) / WE1AV
   KAPPAS = H2 * USUBE / (PI2*PI*RBL*WE1AV)
   GO TO 50
   CONTINUE
   BETA4 = 0.0
   KAPPAS = FKWPST / HPS
50 CONTINUE
   ALBET3 = HPS * WE1AV / (RBL*USUBE)
   IF ((IP.EQ.2) GO TO 60
   BETAL = (UE(K+1) - UEO(K+1))/(DPSI*.5*(UE(K+1)/PSI+UEO(K+1)/PSIO))
   GO TO 70
   CONTINUE
   BETAL = (UE(K+1) - UEO(K+1)) / (DPSI*.5*(UE(K+1)/PSI + UO1))
70 CONTINUE
   BETA6 = BETA4
   BETA2 = 0.0
   ALST6 = ALPHA(K+1)

```

BL004300
 BL004310
 BL004320
 BL004330
 BL004340
 BL004350
 BL004360
 BL004370
 BL004380
 BL004390
 BL004400
 BL004410
 BL004420
 BL004430
 BL004440
 BL004450
 BL004460
 BL004470
 BL004480
 BL004490
 BL004500
 BL004510
 BL004520
 BL004530
 BL004540
 BL004550
 BL004560
 BL004570
 BL004580
 BL004590
 BL004600
 BL004610
 BL004620
 BL004630
 BL004640
 BL004650
 BL004660
 BL004670
 BL004680
 BL004690
 BL004700
 BL004710
 BL004720
 BL004730
 BL004740
 BL004750
 BL004760
 BL004770
 BL004780
 BL004790
 BL004800

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      ALPHA(K+1) = 0.0
      ALB1(K+1) = ALPHA(K+1)
      ALB2 = ALPHA(K+1)
      ALB3 = 1.0
      ALBET2 = 0.0
      XKA = 0.0
      BETA3 = 0.0
      GO TO 90
    CONTINUE
    CALL AVERG(K,IP)
    IF (ALB1(K+1) .LT. 1.00-06) GO TO 90
    XKA = HPS * KAPPAS / ALB1(K+1)
    CONTINUE
    ALP2 = ALPHA(K+1) * ALPHA(K+1)
    GO TO 210
  90 CONTINUE
    IF (A.NE.0) GO TO 110
    CSP = COSPH(1)
    SNP = SINPH(1)
    GO TO 120
  100 CONTINUE
    PHIIV = S*(PHI(K) + PHI(K+1))
    CSP = DCOS(PHIIV)
    SNP = DSIN(PHIIV)
    COMDE = SO*CSP*CSP + S1*SNTHB2*SNP*SNP
    UE(K+1) = 0.0
    WE(K+1) = 0.0
    CAPMI(K+1) = WE(K+1)
    IF (K.EQ. 1) WE1 = WE(2) / (PI*DW(2))
    ALPHA(K+1) = (S1-SO)*SNTHB*SNP*CSP/COMDE
    ALP2 = ALPHA(K+1) * ALPHA(K+1)
    PMOU(K+1) = (S1-SO)*(CSP*CSP-SNTHB2*SNP*SNP)/COMDE
    PUOW = 1.0
    FKPS = 1.0
    ALB3 = 1.0
    IF (DABS(SO-S1).GT..0000000001) GO TO 140
  130 CONTINUE
    XKA = 2.0
    BETA4 = 2.0
    GO TO 150
  140 CONTINUE
    XKA = 0.0
    BETA4 = 1.0
    CONTINUE
    IF (K.EQ.0) GO TO 170
  150 CONTINUE
    HOLD = H
  160 CONTINUE
  170 CONTINUE
    H = SNTHB / (CSP*CSP + SNTHB2*SNP*SNP)

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```

180 IF (K.EQ.0) GO TO 190
    CONTINUE GO TO 190
190 ALBET3 = PMOU(K+1)
    GO TO 200
    CONTINUE
    UO1 = H*SO/SINTHB
    1 (COSPH(2)*COSPH(2)+SINTHB2*SINTHB(2)*SINTHB(2))
    ALBET3 = W02/(PI2*H*UO1*OW(1))
    CONTINUE
200 PIH = H * PI2
    ALPAV = ALPHA(K+1)
    COMDE = SO*COSPH(K+1)*COSPH(K+1)+SINTHB2*SINTHB(K+1)*SINTHB(K+1)
    ALPHA(K+1) = (S1-SO)*SINTHB*SINTHB(K+1)*COSPH(K+1)/COMDE
    CONTINUE
210 RETURN
    END
    BL005320
    BL005330
    BL005340
    BL005350
    BL005360
    BL005370
    BL005380
    BL005390
    BL005400
    BL005410
    BL005420
    BL005430
    BL005440
    BL005450
    BL005460
    BL005470
    BL005480

```

```

SUBROUTINE PRECAL
    SUBROUTINE PRECAL DOES ONE-TIME CALCULATIONS AND INITIALIZES
    VARIABLES. ALSO THE INITIAL INPUT IS PRINTED OUT IN THIS
    SUBROUTINE.
    IMPLICIT REAL*8 (A-H,O-Z)
    COMMON /COL/ A, EPS, KMAX, JMAX, JMAX1, JMAX2, KMAX1, JK, PI,
    1 PMAX, SO, X, Y, DN2(50), W(100), X(100),
    2 WRTX(100), S(100), F(50,100), G(50,100), GO(50,100),
    1 UE(100), UEG(100), V(50,100), WF(100),
    2 WE(100), COSPH(100), DPHI(100), PSI,
    1 DM(100), PHI(100), SINTHB, HMO(100),
    2 HPSI(100), HM(100), ALBET3, ALP2, BETA1,
    1 ALBET2, BETA3, BETA4, KAPPAS,
    2 KAPPAM, IOUT,
    1 COMMON /COL/ IN, TOL, LITER, JK1, PI2,
    2 SNTHB, COSAL, ROLO(100), SOLD(100), THB(100), XOLD(100),
    1 XWTR(100)
    BL005490
    BL005500
    BL005510
    BL005520
    BL005530
    BL005540
    BL005550
    BL005560
    BL005570
    BL005580
    BL005590
    BL005600
    BL005610
    BL005620
    BL005630
    BL005640
    BL005650
    BL005660
    BL005670
    BL005680
    BL005690
    BL005700
    BL005710
    BL005720
    BL005730
    BL005740
    BL005750
    BL005760
    BL005770

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```
COMMON/CO17/ IPRT      , KPRT
COMMON/CO23/ AO        , BO
1 COMMON/CO25/ RB CAPTH , RC
COMMON/CO30/ IA, JK2   , CPTHM
COMMON/CO30/ XPOVA     , QTA
1 COMMON/CO32/ CU       , CV
COMMON/CO34/ ICON      , JTST
COMMON/CO36/ ICAP      , NIT
COMMON/CO41/ CRT       ,
COMMON/CO43/ DRDST     , KHALF
COMMON/TEXT/ IREXV     , XTST
DIMENSION DNSUM(50)
VOL=1001
DNSUM(1)=0.0
ALP=0.174533 * ALP
PI2=2.*PI
JMAX1=JMAX-1
JMAX2=JMAX-2
KMAX1=KMAX-1
ICAP=CAPTH
CPTHM=1.-CAPTH
CPTHM=1.-CAPTH
CPTH2=2.-CAPTH
JTST=JMAX/2+1
XTST=1
XOLD(1)=0.0
XOLD(1)=-DPST
S(1)=0.0
X(1)=-DPST
PSI=PSI
XWRT(1)=0.0
DO 10 J=1,JMAX1
DN2(J)=DN(J)*DN(J)
CONTINUE
F(1,1)=0.0
G(1,1)=0.0
FGINCR=1./JMAX1
G(JMAX,1)=1.0
DO 20 J=2,JMAX1
F(J,1)=F(J-1,1) + FGINCR*(F(JMAX,1)
G(J,1)=G(J-1,1) + FGINCR*(G(JMAX,1)
CONTINUE
PHI(1)=0.0
DCOS(ALP)
SINAL=DSIN(ALP)
```

C

10

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```

30      DO 30 K=2,KMAX1
        DPHI(K) = PI2 * DM(K)
        PHI(K) = PHI(K-1) + DPHI(K-1)
        CONTINUE
        PHI(KMAX1) = PHI(KMAX1) + DPHI(KMAX1)
        DO 40 K=1,KMAX
          SINPH(K) = DSIN(PHI(K))
          COSPH(K) = DCOS(PHI(K))
          HW(K) = 0.0
          SOLD(K) = 0.0
          R(K) = 0.0
          FOIL(K) = 0.0
          GOIL(K) = 0.0
          FOIJMAX(K) = 1.0
          GOIJMAX(K) = 1.0
          CONTINUE
C**      COEFFICIENTS FOR SEVERAL GEOMETRIES ARE DEFINED.
40      IF (A-LT.1.E+06) GO TO 60
50      CONTINUE
        IA = 2
        GO TO 70
60      CONTINUE
        IA = 1
        GO TO 70
70      CONTINUE
        IF (IA.EQ.2) GO TO 110
80      CONTINUE
        IF (A-NE-RB) GO TO 100
C***** SPHERE
90      CONTINUE
        AD = 2.0 / 3.
        GO TO 130
C***** ELLIPSOID OF REVOLUTION
100     CONTINUE
        E = DSQRT(1.-RB/A)
        AD = 2.*1.-E**3 / (E**3) * (1.-E**3) * DLOG((1.-E)/(1.-E))
        GO TO 130
110     CONTINUE
        IF (RB.EQ.RC) GO TO 120
C***** ELLIPTIC PARABOLOID
        Q = DSQRT(RC/RB)
        AD = 0.0
        BO = 2.*Q / (1. + Q)
        GO TO 130
C***** PARABOLOID OF REVOLUTION
120     CONTINUE
        AD = 0.0

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```
7      '8X,KC = 'ELL.4/  
8      15X,KPRT = '1,12,10X,ALPHA = 'ELL.4,8X,RB = 'ELL.4,9X,ELL.4,9X  
9      32X,IBEXT = 'ELL.4/  
1     9920 FORMAT (1H0//1X,DM*)  
2     9930 FORMAT (2X,13,4X,ELL.4)  
3     9940 FORMAT (1H0//4X,J,9X,DN,9X,ETA)  
4     9950 FORMAT (2X,13,4X,ELL.4,4X,ELL.4)  
5     END
```

BL007310
BL007320
BL007330
BL007340
BL007350
BL007360
BL007370
BL007380
BL007390

DOUBLE PRECISION FUNCTION RELIP (COSP,SINP,K)
RELATIONSHIP FOR DEFINING THE COORDINATE R.

```
1     IMPLICIT REAL*8 (A-H,O-Z)  
2     COMMON /CO1/ A, EPS, JMAX, JK, UINF, PT, JMAX1  
3     JMAX2, KMAX, S1, Y0, DN2(50), R(100),  
4     W(100), CO, PRX, PRXOR  
5     COMMON /CO2/ ALPHA(100), DN(50),  
6     S(100), BO, RC, OTA  
7     COMMON /CO3/ RB, IA, JK2, XPOVA  
8     DIMENSION ISLO(2), RB/RC*SINP*SINP  
9     IF (ISEXT-EG) GO TO 10  
10    RELIP = (Y0*COSP+OSQRT((DABS(Y0*COSP))**2.-OTA*(Y0*Y0-(X(K)*XO)  
11    *RB*(2.-XPOVA))))/OTA  
12    GO TO 20  
13    TANP=SINP/COSP  
14    PHI=ATAN(TANP)  
15    XX=X(K)  
16    ZSLO(1)=PHI  
17    CALL SLOALL(XX,ZSLO,1,12,2,0,0,3,0)  
18    RELIP=ZSLO(2)  
19    RETURN  
20    END
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BL007400
BL007410
BL007420
BL007430
BL007440
BL007450
BL007460
BL007470
BL007480
BL007490
BL007500
BL007510
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BL007580
BL007590
BL007600
BL007610
BL007620
BL007630
BL007640
BL007650
BL007660
BL007670
BL007680
BL007690

SUBROUTINE UXYZ (Y,Z,XPOVA)

BL007700
BL007710

C

[illegible]

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```
PHID=DARCOS(YBRP)
PHID=PHID/RAD
IF(NERR.EQ.1) WRITE(8,11) XPOVA,PHID,PSI,PHI(K+1)
ZSLO(1)=PHID
CALL SLOALL(XPOVA,ZSLO,1,10,4,0,0,4,NERR)
UXP=ZSLO(2)*UINF
UYP=ZSLO(3)*UINF
UZF=ZSLO(4)*UINF
IF(NERR.EQ.1) WRITE(8,12) UXP,UYP,UZF
20 RETURN
12 FORMAT(10X,3E15.6)
11 FORMAT(10X,4E15.6)
END
```

BL008230
BL008240
BL008250
BL008260
BL008270
BL008280
BL008290
BL008300
BL008310
BL008320
BL008330
BL008340
BL008350

SUBROUTINE GEOM

CC GEOM SETS UP THE PARAMETERS AND CALLS FOR SOLVING THE VARI
CC EQUATIONS. IN DOING THIS A STEP IS TAKEN ALONG THE BODY
CC (THE PSI-DIRECTION) AND SOLUTIONS FOR ALL PHI AT THIS LOCATIO
CC ARE FOUND.

```
IMPLICIT REAL*8 (A-H,O-Z)
REAL*8 KAPPAS, KAPPAM
COMMON /COL/ A, EPS, JMAX, JMAX1, JMAX2, KMAX, KMAX1, JK, PI, PSIMAX, SO, S1,
UINF, UINF, XO, YO
COMMON /CO2/ ALPHA(100), DN(50), DN2(50), R(100), RWRTX(100), S(100), W(
100), X(100)
COMMON /CO3/ F(50,100), FO(50,100), G(50,100), GO(50,100), UE(100), UEO
1(100), V(50,100), WE(100), WEO(100)
COMMON /CD4/ ALP, COSPH(100), DPH(100), DPST, DM(100), PHI(100), PSI, PS
10, SINPH(100), SINTH8
COMMON /CO5/ HPSI(100), HW(100), HWO(100)
COMMON /CO6/ ALBET3, ALP2, BETAL, BETA2, BETA3, BETA4, KAPPAS, KAP
1PA
COMMON /CO7/ ALPHO(50), ALPH1(50), ALPH2(50), ALPH3(50), ALPH4(50), ALP
1H(50), ALPH6(50)
COMMON /CO10/ LITER, JKI, PI2, SINTH2
COMMON /CO11/ PIH
COMMON /CO12/ HM, ME1
COMMON /CO13/ COSP42, DW2, PRXS, PRXORS, R2, SINPH2, XPOVAS, X2
COMMON /CO14/ BOLO(100), SOLO(100), THB(100), XOLD(100), XWRTR(100)
COMMON /CO15/ ANGAL, A3, B3, C3, DFDNI, DGDNI, GAMMA(50), ISS
COMMON /CO17/ IPR, KPR1
COMMON /CO22/ ANGLE(50)
COMMON /CO24/ GPR, PHIPR, RP, UPHP, USP, XP
COMMON /CO26/ AIKA, AL3T
COMMON /CO27/ GOSAV(50), GSAV(50)
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AD-A051 971

VIRGINIA POLYTECHNIC INST AND STATE UNIV BLACKSBURG --ETC F/G 20/4
THREE-DIMENSIONAL INCOMPRESSIBLE BOUNDARY LAYERS ON BLUNT BODIE--ETC(U)
MAY 77 D L DWOYER, C H LEWIS, P R GOGINENI
VPI/SU-AERO-063-PT-2

UNCLASSIFIED

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2 OF 3
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COMMON /C030/ IA,JK2,OTA,PRX,PRXOR,XPOVA
COMMON /C034/ ICON,JTST,NIT
COMMON /C035/ FTST,FTSTL,GTSTL,VTSTL
COMMON /C036/ ICAP
COMMON /C037/ ENSAV(50),GNSAV(50)
COMMON /C041/ CRI,KHALF
COMMON /C043/ DRDST,XTST
ITER=0
PSIDIF=DPSI-PSIMAX
W(1)=0.000
DS=0.000
IP=0.000
NPRT=0
NPRT=IPRT-1
S(1)=DS
CONTINUE
NIT=1
K=0
ROLD(1)=R(1)
XOLD(1)=X(1)
SOLD(1)=S(1)
PSI=PSI+DPSI
S(1)=PSI
IF (PSI.GT.PSIMAX) GO TO 440
IP=IP+1
NPRT=NPRT+1
IF (IP.EQ.1) GO TO 30
IF (X(1).GE.XTST) GO TO 20
FIND COORDINATES IN THE PLANE OF SYMMETRY.
CALL COORD (R,X,XWRTR,ROLD,XOLD,2,K,DS)
GO TO 40
CONTINUE
CALL COORD (X,R,RMRTX,XOLD,ROLD,1,K,DS)
GO TO 40
CONTINUE
R(1)=0.000
X(1)=0.000
HPSI(1)=1.000
CONTINUE
IF (ITER.NE.0) GO TO 70
CALL CALCV(1,0)
DO 50 J=1,JMAX
GAMMA(J)=0.000
CONTINUE
PRINT 450
DO 60 J=1,JMAX
PRINT 460, J,F(J,1),G(J,1),V(J,1),ANGLE(J),GAMMA(J)
CONTINUE
CONTINUE

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      IF (IP.NE.1) GO TO 80
      CALL PARVAL (0,IP)
      FIND INITIAL PROFILE SOLUTION FOR F,G, AND V.

C
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C 80
      CONTINUE
      DO 90 J=1,JMAX
      FNSAV(J)=F(J,1)
      GNSAV(J)=G(J,1)
      CONTINUE
      ITER=1
      IF (IP.NE.1) GO TO 100
      ICON=0
      CALL PSIMOM (IP,0)
      CALL OMOMOM (IP,0)
      ICON=1
      CALL PSIMOM (IP,0)
      CALL OMOMOM (IP,0)
      IF (ITER.LE.1) PRINT 470, ITER
      IF (ITER.EQ.1) PRINT 480
      CALL PSIMOM (IP,0)
      CONTINUE
      IF (ITER.LT.1) GO TO 80
      DS=0.0001
      DRDSI=1/(1-ROLD(1))/DPST
      DO 430 K=1,KMAX
      OMEGA=PI(K)/PI2
      IF ((IP.NE.1).OR.(K.EQ.1)) GO TO 120
      DO 110 J=1,JMAX
      F(J,K)=F(J,K-1)
      G(J,K)=G(J,K-1)
      V(J,K)=V(J,K-1)
      CONTINUE
      ICON=0
      NIT=1
      KS=K-1+ICAP
      IF (KS.EQ.0) KS=KS+1
      KMI=K-1
      MPRT=MPRT+1
      IF (K.NE.KMAX) GO TO 230
      FIND F,G, AND V IN THE 180 DEGREE PLANE.

C
C
C 130
      CONTINUE
      DO 140 J=1,JMAX
      FNSAV(J)=F(J,KMAX)
      GNSAV(J)=G(J,KMAX)
      FIND AVERAGE F AND G VALUES.

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140 CONTINUE
    IF (IP.EQ.1) GO TO 160
    IF (NIT.EQ.1) GO TO 180
    DO 150 J=1,JMAX
    F(IJ,KMAX)=0.500*(F(IJ,KMAX)+FO(IJ,KS))
    G(IJ,KMAX)=0.500*(G(IJ,KMAX)+GO(IJ,KS))
    CONTINUE
    GO TO 180
150 GO TO 180
160 CONTINUE
    DO 170 J=1,JMAX
    F(IJ,KMAX)=0.500*(F(IJ,KMAX)+F(IJ,KMAX1))
    G(IJ,KMAX)=0.500*(G(IJ,KMAX)+G(IJ,KMAX1))
    CONTINUE
170 CONTINUE
180 CONTINUE
    IF (NIT.NE.1) GO TO 190
    CALL PARVAL (KMAX,IP)
    CONTINUE
190 IF (IP.EQ.1) GO TO 200
    IF (IP.NE.1) GO TO 200
    CALL SYMH (KMAX)
    ICON=1
    GO TO 210
200 CONTINUE
    CALL PSIMOM (IP,KMAX1)
    CALL OMOMOM (IP,KMAX1)
    FIST1=F(IJTST,KMAX)
    GIST1=G(IJTST,KMAX)
    VIST1=V(IJTST,KMAX)
    IF (NIT.EQ.1) GO TO 220
    CALL CONV
    ICON=1
    IF (ICON.NE.1) GO TO 220
    CONTINUE
210 CONTINUE
    CALL PSIMOM (IP,KMAX1)
    CALL OMOMOM (IP,KMAX1)
    ICON=0
    GO TO 410
220 CONTINUE
    NIT=NIT+1
    FIST=FIST1
    GIST=GIST1
    VIST=VIST1
    GO TO 130
C
C
C FIND SOLUTION FOR PHI LESS THAN 180 DEGREES.
230 CONTINUE
    ROLD(K+1)=R(K+1)
    XOLD(K+1)=X(K+1)
    IF (IP.EQ.1) GO TO 250
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C
CALCULATE THE COORDINATES AT A STEP AHEAD OF LOCATION WHERE
F, G, AND V SOLUTIONS ARE DESIRED.

240 XPOVAS=XPOVA
PRXS=PRX
PRXDS=PRXOR
IF (K.EQ.1) GO TO 240
CALL COORD (R,X,XWRTX,ROLD,XOLD,2,K,DS)
CONTINUE
GO TO 260
240 CONTINUE
CALL COORD (X,R,RWRTX,XOLD,ROLD,1,K,DS)
GO TO 260
250 CONTINUE
R(K+1)=0.000
X(K+1)=0.000
HPSI(K+1)=0.000
260 CONTINUE
S(K+1)=SOLD(K+1)+DS
IF ((IP.EQ.1).AND.(K.EQ.1)) GO TO 390
R2=R(K+1)
X2=X(K+1)
DM2=DM(K+1)
COSPH2=COSPH(K+1)
SINPH2=SINPH(K+1)
CONTINUE
DO 280 J=1,JMAX
FNSAV(J)=F(J,K)
GNSAV(J)=G(J,K)
IF ((K.EQ.1).AND.(NIT.EQ.1)) GNSAV(J)=GS(V(J))
CONTINUE
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C
FIND AVERAGE F AND G VALUES.
IF (IP.EQ.1) GO TO 320
IF (K.EQ.1) GO TO 290
IF (NIT.EQ.1) GO TO 340
CONTINUE
DO 310 J=1,JMAX
F(J,K)=0.000
G(J,K)=0.000
F(J,K)=F(J,K)+F(J,K)
G(J,K)=G(J,K)+G(J,K)
G(J,K)=0.500*(F(J,K)+G(J,K))
GO TO 310
300 CONTINUE
G(J,K)=GS(V(J))
310 CONTINUE
GO TO 340
320 CONTINUE

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330 DO 330 J=1,JMAX
340 F(J,K)=O.500*(F(J,K)+F(J,K-1))
      G(J,K)=O.500*(G(J,K)+G(J,K-1))
      CONTINUE
      IF (NIT.NE.1) GO TO 350
      CALL PARVAL (KMI,IP)
      CONTINUE
      IF (CPI.EQ.1.0D0) GO TO 360
      IF ((IP.NE.1).OR.(OMEGA.LE.0.25D0)) GO TO 360
      CALL SYMM (K)
      ICON=1
      GO TO 370
      CONTINUE
      CALL PSIMOM (IP,KMI)
      CALL OMEMOM (IP,KMI)
      FYSTI=F(JYST,K)
      GYSTI=G(JYST,K)
      VYSTI=V(JYST,K)
      IF (NIT.EQ.1) GO TO 380
      CALL CONV
      ICON=1
      IF ((ICON.NE.1) GO TO 380
      CONTINUE
      CALL PSIMOM (IP,KMI)
      CALL OMEMOM (IP,KMI)
      ICON=0
      GO TO 390
      CONTINUE
      FYSTI=F(JYST,I)
      GYSTI=G(JYST,I)
      VYSTI=V(JYST,I)
      NIT=NIT+1
      GO TO 270
      CONTINUE
      IF (KMI.NE.0) GO TO 410
      PSIDIF=PSI+DPSI-PSIMAX
      IF ((IPRT.NE.IPRT).AND.(PSIDIF.LE.0.001D0)) GO TO 400
      MPRT=0
      CALL PRTPRO (2,0)
      CONTINUE
      GO TO 420
      CONTINUE
      IF ((IPRT.NE.KPRT).AND.(K.NE.KMAX)) GO TO 420
      MPRT=0
      IF ((IPRT.NE.IPRT).AND.(PSIDIF.LE.0.001D0)) GO TO 420
      CALL PRTPRO (3,KMI)
      CONTINUE
      IF (K.EQ.KMAX) GO TO 430
      XOLD(K)=X(K+1)
      DRDSI=1/(K+1)-ROLD(K+1))/DPSI

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C C 430  
ROLD(K+I)=R(K+I)  
SOLD(K+I)=S(K+I)  
  
C C CONTINUE  
IF (NPRT.EQ.IPRT) NPRT=O  
PSIO=PSI  
  
GO BACK TO S AND TAKE ANOTHER STEP IN THE PSI-DIRECTION.  
  
C C GO TO IO  
CONTINUE  
RETURN  
  
C C  
C C 450 FORMAT (IH,O,I,LIMITATION O//8X,IHJ,5X,IHF,12X,IHG,12X,IHV,1OX,SHAL  
PHA,1OX,SHGAMMA)  
FORMAT (6X,I3,.5(ELL,4,2X),4(E9.2,2XI))  
C 470 FORMAT (IH,I,LIMITATION ,I3)  
C 480 FORMAT (3OX,I5INITIAL PROFILE)  
END
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```
COMMON /CO15/ ANGAL,A3,B3,C3,DFDNL,DGDN1,GAMMA(50),TSS
COMMON /CO21/ ANGLE(50)
COMMON /CO22/ GPR,PHIPR,RP,UPHP,USP,XP
COMMON /CO23/ XKA,ALST
COMMON /CO31/ UAP,UYP,UZP
COMMON /CO34/ ICON,JYST,NIT
COMMON /CO44/ TSF
Y=R(K+1)*COSPH(K+1)
Z=R(K+1)*SINPH(K+1)
UINF2=UINF*UINF
PRCOF=1.000-(UE(K+1)**2+WE(K+1)**2)/UINF2
GO TO (10,30,30), LOC

*****INITIAL PROFILE
CONTINUE
WRITE (6,100) PSI,PHI(1),X(1),ALPHA(1),DFDNL,Y,ALBET3,DGDN1,Z,UE(1),
1),WE(1),HPSI(1),R(1),HM(1),S(1),TSS,PCOF
DO 20 J=1,JMAX
WRITE (6,110) J,F(J,1),G(J,1),V(J,1),ANGLE(J),GAMMA(J)
CONTINUE
WRITE (8,90) ALP
WRITE (8,60)
WRITE (25) ALP
GO TO 50

*****GENERAL CASE
CONTINUE
WRITE (6,8000) XP, RP, PHIPR, GPR, USP, UPHP
WRITE (6,8001) HM(1)
8001 FORMY (HM)
FOLLOWING CARDS ARE TO BE USED FOR BLOTINERS PRESSURE PATCH
IF(P21 .GT. 0.59100 .AND. PSI .LT. 0.50500) GO TO 31
IF(P31 .GT. 0.59100 .AND. PSI .LT. 0.50500) GO TO 31
IF(P31 .GT. 0.59100 .AND. PSI .LT. 0.70500) GO TO 31
IF(P31 .GT. 0.79100 .AND. PSI .LT. 0.80500) GO TO 31
GO TO 50
31 IF(PHI(K+1) .LT. 2.200) GO TO 50
KK=K
IF (KK.EQ.0) KK=1
WRITE (6,120) PSI,PHI(K+1),X(K+1),ALPHA(K+1),BETA1,DFDNL,Y,ALBET2,
1BETA2,DGDN1,Z,ALBET3,BETA3,HPSI(K+1),X(KA,BETA4,HM(KK),R(K+1),UE(
2+1),KAPPAS,S(K+1),WE(K+1),NIT,KAPPAS,TSS,TSF,ANGAL,PCOF
UEND=UE(K+1)/UINF
WEND=WE(K+1)/UINF
UXPND=UXP/UINF
UYPND=UYP/UINF
UZPND=UZP/UINF
WRITE (25) PSI,HPSI(K+1),HM(KK),R(K+1),S(K+1),UEND,WEND,X(K+1)
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C      130) CALL ERRSET (207,256,10,0,0,209)
      10  PI=DARCOS(-1.00)
      20  DO 10 I=1,150
      30  READ (10,END=20) XP(I),KPL
      40  READ (10) ((ANYVA(I,K,J),J=1,KPL),K=1,5)
      50  CONTINUE
      60  NS=I-1
      70  DO 60 I=1,NS
      80  XP(I)=XP(I)-XO
      90  DO 30 J=1,KPL
      100 PHP=ANYVA(I,1,J)
      110 QPR=ANYVA(I,5,J)
      120 CPHP=DCOS(PHP)
      130 SPHP=DSIN(PHP)
      140 Y=QPR*CPHP-YO
      150 Z=RPR*CPHP
      160 ZBY=Z/Y
      170 P(J)=DATAN(ZBY)
      180 IF (P(J)-LT-0.00) P(J)=P(J)+PI
      190 R(J)=DSQRT(Y*Y+Z*Z)
      200 CONTINUE
      210 DO 50 J=1,KPL
      220 PP=ANYVA(I,1,J)
      230 JJ=0
      240 JJ=JJ+1
      250 IF (PP(JJ)-LT-PP) GO TO 40
      260 IF (JJ-1) JJ=3
      270 IF (JJ-1) JJ=3
      280 IF (JJ-1) JJ=3
      290 JJ=JJ-2
      300 JJ=JJ-1
      310 JP=JJ-1
      320 CALL INTER4 (PP,P(J2),P(J1),P(JJ),P(JP),R(J2),R(J1),R(JJ),R(JP),RR)
      330 WRITE (8,120) P(J2),P(J1),P(JJ),P(JP),R(J2),R(J1),R(JJ),R(JP),RR
      340 WRITE (8,130) I,J,JJ,PP,RR
      350 ANYVA(I,2,J)=RR
      360 CONTINUE
      370 WRITE (12) XP(I),KPL
      380 WRITE (12) ((ANYVA(I,K,J),J=1,KPL),K=1,2)
      390 WRITE (6,90) I,XP(I),KPL
      400 WRITE (6,100) (J,ANYVA(I,1,J),ANYVA(I,2,J),J=1,KPL)
      410 CONTINUE
      420 WRITE (6,110)
      430 DO 80 J=1,KPL
      440 DO 70 I=1,NS
      450 ADR(I)=ANYVA(I,2,J)
      460 CALL DERIV5 (XP,ADR,NS,0,0,DRDX)
      470 WRITE (13) ANYVA(I,1,J),NS
      480 WRITE (13) (ANYVA(I,2,J),I=1,NS), (XP(I),I=1,NS), (DRDX(I),I=1,NS)

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80      WRITE (6,90) J,ANYVA(1:J),NS
      WRITE (8,100) I,ANYVA(1:2,J),XP(1),DRDX(1),I=1,NS
      CONTINUE
      REWIND 12
      RETURN
90      FORMAT (//1X,13.6X,ELSE,6.5X,13)
100     FORMAT (//1X,4(13.2X,213.6))
110     FORMAT (//20X,8UNIT,13//)
120     FORMAT (5E13:6.5X,5E13.6)
130     FORMAT (316.5X,5E15.6)
      END

```

```

SUBROUTINE DERIV5 (X,Y,NP,J,P,IP,DYDX)
IMPLICIT REAL *8(A-H,O-Z)

C
C
      DIMENSION X(130), Y(130), DYDX(130)

      DO 10 J=1,NP
        K=J
        IF (IP.EQ.1) K=JP
        IF (K.LT.3) K=3
        IF (K.GT.(NP-2)) K=NP-2
        XX=X(K)
        YY=Y(K)
        XX1=X(K-1)
        YY1=Y(K-1)
        XX2=X(K+1)
        YY2=Y(K+1)
        XX3=X(K+2)
        YY3=Y(K+2)
        XX4=X(K-2)
        YY4=Y(K-2)
        XX5=X(K+3)
        YY5=Y(K+3)
        CALL FD5 (XX,XX1,XX2,XX3,XX4,XX5,YY1,
                  DYDX(J)=DER
                  IF (IP.EQ.1) GO TO 20
                )
        IF (CONTINUE
          END
        )
      10
      20

```

SUBROUTINE FD5 (X,X1,X2,X3,X4,X5,F1,F2,F3,F4,F5,FX)

8100	59
8100	60
8100	61
8100	62
8100	63
8100	64
8100	65
8100	66
8100	67
8100	68
8100	69
8100	70
8100	71

8L00	1
8L00	2
8L00	3
8L00	4
8L00	5
8L00	6
8L00	7
8L00	8
8L00	9
8L00	10
8L00	11
8L00	12
8L00	13
8L00	14
8L00	15
8L00	16
8L00	17
8L00	18
8L00	19
8L00	20
8L00	21
8L00	22
8L00	23
8L00	24
8L00	25
8L00	26
8L00	27

8100 1

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```
IMPLICIT REAL*8(A-H,O-Z)
A1=X-X4*(2.00*X-X2-X3)+(X-X2)*(X-X3)*(2.00*X-X4-X5)
A2=X-X4*(2.00*X-X2-X3)+(X-X2)*(X-X3)*(2.00*X-X4-X5)
A3=X-X4*(2.00*X-X2-X3)+(X-X2)*(X-X3)*(2.00*X-X4-X5)
A4=X-X4*(2.00*X-X2-X3)+(X-X2)*(X-X3)*(2.00*X-X4-X5)
A5=X-X4*(2.00*X-X2-X3)+(X-X2)*(X-X3)*(2.00*X-X4-X5)
D1=X-X4*(X-X3)*(X-X4)*(X-X5)
D2=X-X4*(X-X3)*(X-X4)*(X-X5)
D3=X-X4*(X-X3)*(X-X4)*(X-X5)
D4=X-X4*(X-X3)*(X-X4)*(X-X5)
D5=X-X4*(X-X3)*(X-X4)*(X-X5)
C1=A1/D1
C2=A2/D2
C3=A3/D3
C4=A4/D4
C5=A5/D5
F=C1*F1+C2*F2+C3*F3+C4*F4+C5*F5
RETURN
END
```

BL00 100
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BL00 100
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BL00 100
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BL00 100
BL00 100
BL00 100

-101-

```
SUBROUTINE INTER4 (X,X1,X2,X3,X4,F1,F2,F3,F4,F)
IMPLICIT REAL*8(A-H,O-Z)
A1=X-X2*(X-X3)*(X-X4)
A2=X-X1*(X-X3)*(X-X4)
A3=X-X1*(X-X2)*(X-X4)
A4=X-X1*(X-X2)*(X-X3)
D1=X-X1*(X-X2)*(X-X3)*(X-X4)
D2=X-X2*(X-X1)*(X-X3)*(X-X4)
D3=X-X3*(X-X1)*(X-X2)*(X-X4)
D4=X-X4*(X-X1)*(X-X2)*(X-X3)
C1=A1/D1
C2=A2/D2
C3=A3/D3
C4=A4/D4
F=C1*F1+C2*F2+C3*F3+C4*F4
RETURN
END
```

BL00 100
BL00 100
BL00 100
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BL00 100

SUBROUTINE SLOALL (XX,Z,II,IT,NV,MX,MZ,NOR,NPRT)
IMPLICIT REAL*8(A-H,O-Z)

BL00 100
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BL00 100
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[illegible]

10		20
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1	NOR2=NOR2+1	31 NOR2=2*NOR	100	WRITE (11,240) (X(K),Y(K,I,L),K-1,NORI)	8100
2	IF (JRK-EQ-301) XX1=10.0**20	IF (JRK-EQ-301) GO TO 50	101	CALL INTER3 (XX,X(1),X(2),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8101
3	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	102	CALL INTER4 (XX,X(1),X(2),X(3),X(4),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8102
4	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	103	CALL INTER5 (XX,X(1),X(2),X(3),X(4),X(5),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8103
5	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	104	CALL INTER6 (XX,X(1),X(2),X(3),X(4),X(5),X(6),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8104
6	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	105	CALL INTER7 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8105
7	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	106	CALL INTER8 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8106
8	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	107	CALL INTER9 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8107
9	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	108	CALL INTER10 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L))	8108
10	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	109	CALL INTER11 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L))	8109
11	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	110	CALL INTER12 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L))	8110
12	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	111	CALL INTER13 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L))	8111
13	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	112	CALL INTER14 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L))	8112
14	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	113	CALL INTER15 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L),Y(15,I,L))	8113
15	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	114	CALL INTER16 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),X(16),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L),Y(15,I,L),Y(16,I,L))	8114
16	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	115	CALL INTER17 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),X(16),X(17),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L),Y(15,I,L),Y(16,I,L),Y(17,I,L))	8115
17	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	116	CALL INTER18 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),X(16),X(17),X(18),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L),Y(15,I,L),Y(16,I,L),Y(17,I,L),Y(18,I,L))	8116
18	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	117	CALL INTER19 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),X(16),X(17),X(18),X(19),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L),Y(15,I,L),Y(16,I,L),Y(17,I,L),Y(18,I,L),Y(19,I,L))	8117
19	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	118	CALL INTER20 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),X(16),X(17),X(18),X(19),X(20),Y(1,I,L),Y(2,I,L),Y(3,I,L),Y(4,I,L),Y(5,I,L),Y(6,I,L),Y(7,I,L),Y(8,I,L),Y(9,I,L),Y(10,I,L),Y(11,I,L),Y(12,I,L),Y(13,I,L),Y(14,I,L),Y(15,I,L),Y(16,I,L),Y(17,I,L),Y(18,I,L),Y(19,I,L),Y(20,I,L))	8118
20	IF (JRK-EQ-301) GO TO 50	IF (JRK-EQ-301) GO TO 50	119	CALL INTER21 (XX,X(1),X(2),X(3),X(4),X(5),X(6),X(7),X(8),X(9),X(10),X(11),X(12),X(13),X(14),X(15),X(16),X(17),X(18),X(

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180 IF (MM.GT.(JJ-MM1)) MM=JJ-MM1
    MM1=MM-1
    MP1=MM-2
    MP2=MM+2
    IF (INPR.EQ.0) GO TO 180
    IF (INOR.EQ.2) WRITE (11,260) Z(11), (Y2(11,MR), MR=MM1,MP1)
    IF (INOR.EQ.3) WRITE (11,260) Z(11), (Y2(11,MR), MR=MM2,MP1)
    IF (INOR.EQ.4) WRITE (11,260) Z(11), (Y2(11,MR), MR=MM2,MP2)
    DO 230 J1=1,NV
    IF (J1.EQ.1) GO TO 230
    IF (INOR.EQ.3) GO TO 190
    IF (INOR.EQ.4) GO TO 200
    IF (INPR.EQ.1) WRITE (11,270) J1,(Y2(J1,MR), MR=MM1,MP1)
    CALL INTER3 (Z(11),Y2(11,MM1),Y2(11,MM),Y2(11,MP1),Y2(11,MM1),Y2(
11,MM),Y2(11,MP1),Z(J1))
    GO TO 210
    IF (INPR.EQ.1) WRITE (11,270) J1,(Y2(J1,MR), MR=MM2,MP1)
    CALL INTER4 (Z(11),Y2(11,MM2),Y2(11,MM1),Y2(11,MM),Y2(11,MP1),Y2(
11,MM2),Y2(11,MM1),Y2(11,MM),Y2(11,MP1),Z(J1))
    GO TO 210
    IF (INPR.EQ.1) WRITE (11,270) J1,(Y2(J1,MR), MR=MM2,MP2)
    CALL INTER5 (Z(11),Y2(11,MM2),Y2(11,MM1),Y2(11,MM),Y2(11,MP1),Y2(11,
11,MP2),Y2(11,MM2),Y2(11,MM1),Y2(11,MM),Y2(11,MP1),Y2(11,MP2),Z(J1)
21 IF (INPR.EQ.1) WRITE (11,280) Z(11),Z(J1),J1
    X1=XX
    GO TO 230
220 CONTINUE
    Z(J1)=Y2(J1,MM)-(Y2(J1,MM)-Y2(J1,MM1))/(Y2(11,MM)-Y2(11,MM1))*(Y2
11,MM)-Z(11))
    X1=XX
    CONTINUE
    RETURN
230
240 FORMAT (1X,5I2E13.6))
250 FORMAT (10X,E15.6,15X,E15.6)
260 FORMAT (10X,E15.6,5X,E15.6)
270 FORMAT (10X,I3,5X,E15.6)
280 FORMAT (10X,E15.6,5X,E15.6,2X,I3)
    END

SUBROUTINE INTER5 (X,X1,X2,X3,X4,X5,F1,F2,F3,F4,F5,F)
IMPLICIT REAL*8(A-H,O-Z)
SUBROUTINE INTER5 IS CALLED BY SUBROUTINES BLUNT1, BLUNT2, EDGE,
INIT, MACH, SLOW, WALL, ZRO, AND INTRP5.

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8100 6
8100 7
8100 8
8100 9
8100 10
8100 11
8100 12
8100 13
8100 14
8100 15
8100 16
8100 17
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8100 19
8100 20
8100 21
8100 22
8100 23
8100 24
8100 25
8100 26
8100 27
8100 28
8100 29

SUBROUTINE INTERS INTERPOLATES FOR THE VALUE F CORRESPONDING TO
POINT X USING 5 POINT LAGRANGIAN INTERPOLATION FORMULA.

ASSUMES X1 .LE. X .LE. X5.

A1=(X-X2)*(X-X3)*(X-X4)*(X-X5)
A2=(X-X1)*(X-X4)*(X-X5)
A3=(X-X1)*(X-X3)*(X-X5)
A4=(X-X1)*(X-X2)*(X-X5)
A5=(X-X1)*(X-X3)*(X-X4)
D1=(X1-X2)*(X1-X3)*(X1-X4)*(X1-X5)
D2=(X2-X1)*(X2-X3)*(X2-X4)*(X2-X5)
D3=(X3-X1)*(X3-X2)*(X3-X4)*(X3-X5)
D4=(X4-X1)*(X4-X2)*(X4-X3)*(X4-X5)
D5=(X5-X1)*(X5-X2)*(X5-X3)*(X5-X4)
C1=A1/D1
C2=A2/D2
C3=A3/D3
C4=A4/D4
C5=A5/D5
F=C1*F1+C2*F2+C3*F3+C4*F4+C5*F5
RETURN
END

8100 1
8100 2
8100 3
8100 4
8100 5
8100 6
8100 7
8100 8
8100 9
8100 10
8100 11
8100 12
8100 13
8100 14
8100 15
8100 16
8100 17
8100 18
8100 19
8100 20
8100 21
8100 22

SUBROUTINE INTER3 (X,X1,X2,X3,F1,F2,F3,F)
IMPLICIT REAL*8(A-H,O-Z)
SUBROUTINE INTER3 IS CALLED BY SUBROUTINE ADEETA.
SUBROUTINE INTER3 INTERPOLATES FOR THE VALUE F CORRESPONDING TO
POINT X USING 3 POINT LAGRANGIAN INTERPOLATION.

ASSUMES X1 .LE. X .LE. X3.

A1=(X-X2)*(X-X3)
A2=(X-X1)*(X-X3)
A3=(X-X1)*(X-X2)
D1=(X1-X2)*(X1-X3)
D2=(X2-X1)*(X2-X3)
D3=(X3-X1)*(X3-X2)
C1=A1/D1
C2=A2/D2
C3=A3/D3
F=C1*F1+C2*F2+C3*F3
RETURN
END

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SUBROUTINE INTER4 (X,X1,X2,X3,X4,F1,F2,F3,F4,F)
IMPLICIT REAL*8(A-H,O-Z)
A1=(X-X2)*(X-X3)*(X-X4)
A2=(X-X1)*(X-X3)*(X-X4)
A3=(X-X1)*(X-X2)*(X-X4)
A4=(X-X1)*(X-X2)*(X-X3)
D1=(X1-X2)*(X1-X3)*(X1-X4)
D2=(X2-X1)*(X2-X3)*(X2-X4)
D3=(X3-X1)*(X3-X2)*(X3-X4)
D4=(X4-X1)*(X4-X2)*(X4-X3)
C1=A1/D1
C2=A2/D2
C3=A3/D3
C4=A4/D4
F=C1*F1+C2*F2+C3*F3+C4*F4
RETURN
END

```

8100 1
8100 2
8100 3
8100 4
8100 5
8100 6
8100 7
8100 8
8100 9
8100 10
8100 11
8100 12
8100 13
8100 14
8100 15
8100 16
8100 17

THESE SUBROUTINES ARE DUMMIED TO GET THE INVISCID PRESSURE
DISTRIBUTION FROM BLOTTNERS PROGRAM

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SUBROUTINE AVERG (K,IP)
END
SUBROUTINE CALCV (IP,K)
END
SUBROUTINE CONV
END
SUBROUTINE OMOMOM (IP,K)
END
SUBROUTINE PSIMOM (IP,K)
END
SUBROUTINE SYMM (K)
END
SUBROUTINE TRIDAG (M,MO,K,IP,WN)
END

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8100 1
8100 2
8100 3
8100 4
8100 5
8100 6
8100 7
8100 8
8100 9
8100 10
8100 11
8100 12
8100 13
8100 14
8100 15
8100 16
8100 17

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SUBROUTINE DPRX (COSP,RR,PRX)
IMPLICIT REAL*8(A-H,O-Z)
DIMENSION ZSLO(3)
PHI=DARCOS(COSP)
XX=PHI
ZSLO(1)=RR
CALL SLOALL (XX,ZSLO,1,13,3,0,0,3,0)
PRX=ZSLO(3)
RETURN
END
```

8100 1
8100 2
8100 3
8100 4
8100 5
8100 6
8100 7
8100 8
8100 9
8100 10

2. Program DERVAT Listing

DATE 05/13/77

```

C      PROGRAM MAIN
C      IMPLICIT REAL*8(A-H,O-Z)
C      DIMENSION ANYVAL(100,14,40), XS(100), ADUM(100), DADUM(100)
C      CALL ERSET (207,256,13,0,0,209)
C
C      THIS PROGRAM COMPUTES DERIVATIVES OF THE VELOCITY COMPONENTS, BODY
C      RADIUS AND SCALE FACTORS. THE RAW DATA IS READ FROM UNIT 25. THE
C      X DERIVATIVES ARE COMPUTED USING A 5 POINT LAGRANGIAN POLYNOMIAL
C      IN SUBROUTINES DERIV5 AND FDS. THE ORIGINAL DATA ALONG WITH THE
C      REQUIRED DERIVATIVES ARE WRITTEN ON UNIT 26
C
      I=1
      J=0
      READ (25) ALPHA
      J=J+1
      READ (25) XS(J), (ANYVA(I,K,J),K=1,7)
      IF (J.EQ.1) GO TO 10
      ANYVA(I,1,J)=ANYVA(I,1,1)
      XS=XS(J)-XS(J-1)
      IF (XS.DGT.0.10-02) GO TO 20
      GO TO 10
      BACKSPACE 25
      J=J-1
      KPL=J
      XS(1)=XS(J)
      I=I+1
      WRITE (6,200) ALPHA,KPL
      CONTINUE
      DO 40 J=1,KPL
      READ (25,END=50) XS(1), (ANYVA(I,K,J),K=1,7)
      I=I+1
      GO TO 30
      NS=I-1
      WRITE (6,170)
      DO 60 I=1,NS
      WRITE (6,180) I,XS(I), (ANYVA(I,K,1),K=1,7)
      WRITE (6,190) (ANYVA(I,K,J),K=1,7),J=2,KPL)
      DO 90 J=1,KPL
      K1=1
      IF (K.EQ.1) K1=12
      IF (K.EQ.2) K1=13
      IF (K.EQ.3) K1=10
      IF (K.EQ.4) GO TO 90
      IF (K.EQ.5) K1=8
      IF (K.EQ.6) K1=9
      IF (K.EQ.7) K1=14
      DO 70 I=1,NS
      ADUM(I)=ANYVA(I,K,J)
      CALL DERIVE (XS,ADUM,NS,0,0,DADUM)
      DO 80 I=1,NS
      ANYVA(I,K1,J)=DADUM(I)

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45

DATE 05/13/77

```

90      CONTINUE
100     DO 130 J=1,KPL
110     ADJUST=ANYVA(I,10,J)
120     CALL DERIV5 (XS,ADUM,NS,0,0,DADUM)
130     DO 120 I=1,NS
140     CONTINUE
150     ANYVA(I,1,J)=DADUM(I)
160     CONTINUE
170     WRITE (20,210) ALPHA,KPL
180     DO 140 I=1,NS
190     CONTINUE
200     WRITE (20,220)
210     DO 150 J=1,KPL
220     WRITE (30,230) I,XS(I),ANYVA(I,1,1),ANYVA(I,2,1),ANY
230     VA(I,3,1),ANYVA(I,3,1),ANYVA(I,10,1),ANYVA(I,11,1),
240     WRITE (29,230) I,XS(I),ANYVA(I,4,1),ANYVA(I,7,1),ANY
250     VA(I,5,1),ANYVA(I,8,1),ANYVA(I,9,1)
260     DO 150 J=2,KPL
270     WRITE (30,240) ANYVA(I,1,J),ANYVA(I,2,J),ANYVA(I,13,
280     J),ANYVA(I,3,J),ANYVA(I,10,J),ANYVA(I,11,J)
290     WRITE (29,240) ANYVA(I,4,J),ANYVA(I,7,J),ANYVA(I,5,J),
300     ANYVA(I,8,J),ANYVA(I,6,J),ANYVA(I,9,J)
310     CONTINUE
320     STOP
330     FORMAT (15X,4HISTA,4X,2HXS,13X,4HHPST,11X,2HHW,13X,1HR,14X,1HS,14X
340     1,4HUPSI,11X,2HUM,13X,4HXBLL/)
350     FORMAT (5X,13,2X,8E15.6)
360     FORMAT (25X,7E15.6)
370     FORMAT (15X,4HISTA,4X,2HXS,13X,4HHPST,11X,7HD(HPSI),8X,2HHW,13X,5H
380     1D(HM),10X,1HR,14X,4HD(R),11X,5HD(R))
390     FORMAT (15X,4HISTA,4X,2HXS,13X,1HS,14X,1HX,14X,4HD(X),11X,4HUPSI,1
400     11X,7HD(UPSI),8X,2HUM,13X,5HD(UM))
410     FORMAT (5X,13,2X,8E15.6)
420     FORMAT (25X,7E15.6)
430     END
440     SUBROUTINE DERIV5 (X,Y,NP,JP,IP,DYDX)
450     IMPLICIT REAL *8(A-H,O-Z)
460     DIMENSION X(50), Y(50), DYDX(50)
470     DERO 1
480     DERO 2
490     DERO 3
500     DERO 4
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870

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DATE 05/13/77

UUUU

THIS SUBROUTINE SETS UP THE FITTING OF FIFTH ORDER LAGRANGIAN POLYNOMIALS TO DATA IN ARRAYS X AND Y. THE ACTUAL CALCULATIONS ARE CARRIED OUT IN SUBROUTINE FD5.

```

DO 10 J=1,NP
  K=J
  IF (IP.EQ.1) K=JP
  IF (X(J-1,NP-2)) K=NP-2
  XX=X(J,K-2)
  XX2=X(J,K-1)
  XX3=X(J,K)
  XX4=X(J,K+1)
  XX5=X(J,K+2)
  XX6=X(J,K+3)
  XX7=X(J,K+4)
  XX8=X(J,K+5)
  XX9=X(J,K+6)
  XX10=X(J,K+7)
  XX11=X(J,K+8)
  XX12=X(J,K+9)
  XX13=X(J,K+10)
  XX14=X(J,K+11)
  XX15=X(J,K+12)
  XX16=X(J,K+13)
  XX17=X(J,K+14)
  XX18=X(J,K+15)
  XX19=X(J,K+16)
  XX20=X(J,K+17)
  XX21=X(J,K+18)
  XX22=X(J,K+19)
  XX23=X(J,K+20)
  XX24=X(J,K+21)
  XX25=X(J,K+22)
  XX26=X(J,K+23)
  XX27=X(J,K+24)
  XX28=X(J,K+25)
  XX29=X(J,K+26)
  XX30=X(J,K+27)
  XX31=X(J,K+28)
  XX32=X(J,K+29)
  XX33=X(J,K+30)
  XX34=X(J,K+31)
  XX35=X(J,K+32)
  XX36=X(J,K+33)
  XX37=X(J,K+34)
  XX38=X(J,K+35)
  XX39=X(J,K+36)
  XX40=X(J,K+37)
  XX41=X(J,K+38)
  XX42=X(J,K+39)
  XX43=X(J,K+40)
  XX44=X(J,K+41)
  XX45=X(J,K+42)
  XX46=X(J,K+43)
  XX47=X(J,K+44)
  XX48=X(J,K+45)
  XX49=X(J,K+46)
  XX50=X(J,K+47)
  XX51=X(J,K+48)
  XX52=X(J,K+49)
  XX53=X(J,K+50)
  XX54=X(J,K+51)
  XX55=X(J,K+52)
  XX56=X(J,K+53)
  XX57=X(J,K+54)
  XX58=X(J,K+55)
  XX59=X(J,K+56)
  XX60=X(J,K+57)
  XX61=X(J,K+58)
  XX62=X(J,K+59)
  XX63=X(J,K+60)
  XX64=X(J,K+61)
  XX65=X(J,K+62)
  XX66=X(J,K+63)
  XX67=X(J,K+64)
  XX68=X(J,K+65)
  XX69=X(J,K+66)
  XX70=X(J,K+67)
  XX71=X(J,K+68)
  XX72=X(J,K+69)
  XX73=X(J,K+70)
  XX74=X(J,K+71)
  XX75=X(J,K+72)
  XX76=X(J,K+73)
  XX77=X(J,K+74)
  XX78=X(J,K+75)
  XX79=X(J,K+76)
  XX80=X(J,K+77)
  XX81=X(J,K+78)
  XX82=X(J,K+79)
  XX83=X(J,K+80)
  XX84=X(J,K+81)
  XX85=X(J,K+82)
  XX86=X(J,K+83)
  XX87=X(J,K+84)
  XX88=X(J,K+85)
  XX89=X(J,K+86)
  XX90=X(J,K+87)
  XX91=X(J,K+88)
  XX92=X(J,K+89)
  XX93=X(J,K+90)
  XX94=X(J,K+91)
  XX95=X(J,K+92)
  XX96=X(J,K+93)
  XX97=X(J,K+94)
  XX98=X(J,K+95)
  XX99=X(J,K+96)
  XX100=X(J,K+97)
  XX101=X(J,K+98)
  XX102=X(J,K+99)
  XX103=X(J,K+100)
  XX104=X(J,K+101)
  XX105=X(J,K+102)
  XX106=X(J,K+103)
  XX107=X(J,K+104)
  XX108=X(J,K+105)
  XX109=X(J,K+106)
  XX110=X(J,K+107)
  XX111=X(J,K+108)
  XX112=X(J,K+109)
  XX113=X(J,K+110)
  XX114=X(J,K+111)
  XX115=X(J,K+112)
  XX116=X(J,K+113)
  XX117=X(J,K+114)
  XX118=X(J,K+115)
  XX119=X(J,K+116)
  XX120=X(J,K+117)
  XX121=X(J,K+118)
  XX122=X(J,K+119)
  XX123=X(J,K+120)
  XX124=X(J,K+121)
  XX125=X(J,K+122)
  XX126=X(J,K+123)
  XX127=X(J,K+124)
  XX128=X(J,K+125)
  XX129=X(J,K+126)
  XX130=X(J,K+127)
  XX131=X(J,K+128)
  XX132=X(J,K+129)
  XX133=X(J,K+130)
  XX134=X(J,K+131)
  XX135=X(J,K+132)
  XX136=X(J,K+133)
  XX137=X(J,K+134)
  XX138=X(J,K+135)
  XX139=X(J,K+136)
  XX140=X(J,K+137)
  XX141=X(J,K+138)
  XX142=X(J,K+139)
  XX143=X(J,K+140)
  XX144=X(J,K+141)
  XX145=X(J,K+142)
  XX146=X(J,K+143)
  XX147=X(J,K+144)
  XX148=X(J,K+145)
  XX149=X(J,K+146)
  XX150=X(J,K+147)
  XX151=X(J,K+148)
  XX152=X(J,K+149)
  XX153=X(J,K+150)
  XX154=X(J,K+151)
  XX155=X(J,K+152)
  XX156=X(J,K+153)
  XX157=X(J,K+154)
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  XX159=X(J,K+156)
  XX160=X(J,K+157)
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  XX162=X(J,K+159)
  XX163=X(J,K+160)
  XX164=X(J,K+161)
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  XX166=X(J,K+163)
  XX167=X(J,K+164)
  XX168=X(J,K+165)
  XX169=X(J,K+166)
  XX170=X(J,K+167)
  XX171=X(J,K+168)
  XX172=X(J,K+169)
  XX173=X(J,K+170)
  XX174=X(J,K+171)
  XX175=X(J,K+172)
  XX176=X(J,K+173)
  XX177=X(J,K+174)
  XX178=X(J,K+175)
  XX179=X(J,K+176)
  XX180=X(J,K+177)
  XX181=X(J,K+178)
  XX182=X(J,K+179)
  XX183=X(J,K+180)
  XX184=X(J,K+181)
  XX185=X(J,K+182)
  XX186=X(J,K+183)
  XX187=X(J,K+184)
  XX188=X(J,K+185)
  XX189=X(J,K+186)
  XX190=X(J,K+187)
  XX191=X(J,K+188)
  XX192=X(J,K+189)
  XX193=X(J,K+190)
  XX194=X(J,K+191)
  XX195=X(J,K+192)
  XX196=X(J,K+193)
  XX197=X(J,K+194)
  XX198=X(J,K+195)
  XX199=X(J,K+196)
  XX200=X(J,K+197)
  XX201=X(J,K+198)
  XX202=X(J,K+199)
  XX203=X(J,K+200)
  XX204=X(J,K+201)
  XX205=X(J,K+202)
  XX206=X(J,K+203)
  XX207=X(J,K+204)
  XX208=X(J,K+205)
  XX209=X(J,K+206)
  XX210=X(J,K+207)
  XX211=X(J,K+208)
  XX212=X(J,K+209)
  XX213=X(J,K+210)
  XX214=X(J,K+211)
  XX215=X(J,K+212)
  XX216=X(J,K+213)
  XX217=X(J,K+214)
  XX218=X(J,K+215)
  XX219=X(J,K+216)
  XX220=X(J,K+217)
  XX221=X(J,K+218)
  XX222=X(J,K+219)
  XX223=X(J,K+220)
  XX224=X(J,K+221)
  XX225=X(J,K+222)
  XX226=X(J,K+223)
  XX227=X(J,K+224)
  XX228=X(J,K+225)
  XX229=X(J,K+226)
  XX230=X(J,K+227)
  XX231=X(J,K+228)
  XX232=X(J,K+229)
  XX233=X(J,K+230)
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  XX237=X(J,K+234)
  XX238=X(J,K+235)
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  XX240=X(J,K+237)
  XX241=X(J,K+238)
  XX242=X(J,K+239)
  XX243=X(J,K+240)
  XX244=X(J,K+241)
  XX245=X(J,K+242)
  XX246=X(J,K+243)
  XX247=X(J,K+244)
  XX248=X(J,K+245)
  XX249=X(J,K+246)
  XX250=X(J,K+247)
  XX251=X(J,K+248)
  XX252=X(J,K+249)
  XX253=X(J,K+250)
  XX254=X(J,K+251)
  XX255=X(J,K+252)
  XX256=X(J,K+253)
  XX257=X(J,K+254)
  XX258=X(J,K+255)
  XX259=X(J,K+256)
  XX260=X(J,K+257)
  XX261=X(J,K+258)
  XX262=X(J,K+259)
  XX263=X(J,K+260)
  XX264=X(J,K+261)
  XX265=X(J,K+262)
  XX266=X(J,K+263)
  XX267=X(J,K+264)
  XX268=X(J,K+265)
  XX269=X(J,K+266)
  XX270=X(J,K+267)
  XX271=X(J,K+268)
  XX272=X(J,K+269)
  XX273=X(J,K+270)
  XX274=X(J,K+271)
  XX275=X(J,K+272)
  XX276=X(J,K+273)
  XX277=X(J,K+274)
  XX278=X(J,K+275)
  XX279=X(J,K+276)
  XX280=X(J,K+277)
  XX281=X(J,K+278)
  XX282=X(J,K+279)
  XX283=X(J,K+280)
  XX284=X(J,K+281)
  XX285=X(J,K+282)
  XX286=X(J,K+283)
  XX287=X(J,K+284)
  XX288=X(J,K+285)
  XX289=X(J,K+286)
  XX290=X(J,K+287)
  XX291=X(J,K+288)
  XX292=X(J,K+289)
  XX293=X(J,K+290)
  XX294=X(J,K+291)
  XX295=X(J,K+292)
  XX296=X(J,K+293)
  XX297=X(J,K+294)
  XX298=X(J,K+295)
  XX299=X(J,K+296)
  XX300=X(J,K+297)
  XX301=X(J,K+298)
  XX302=X(J,K+299)
  XX303=X(J,K+300)
  XX304=X(J,K+301)
  XX305=X(J,K+302)
  XX
```

-111-

```

SUBROUTINE FD5 (X,X1,X2,X3,X4,X5,F1,F2,F3,F4,F5,FX)
IMPLICIT REAL*8(A-H,O-Z)
DERO 1
DERO 2

```

THIS SUBROUTINE FITS DATA AND CALCULATES ITS FIRST DERIVATIVE
USING A FIFTH ORDER POLYNOMIAL

[illegible]

DATE 05/13/77

C4=A4/D4
C5=A5/D5
FX=C1+E1+C2*F2+C3*F3+C4*F4+C5*F5
RETURN
END

DERO 16
DERO 17
DERO 18
DERO 19
DERO 20

3. Program INVTAP Listing

DATE 05/13/77

```

C      PROGRAM INVTAP CREATES THE INVISCID DATA TAPE AS REQUIRED
C      BY THE BOUNDARY LAYER PROGRAMME; CALLING DISKIN
C      IMPLICIT REAL*8(A-H,O-Z)
C      WRITE (6,10)
C      CALL ERRSET (207,256,10,0,0,209)
C      CALL DISKIN
C      STOP
C10    FORMAT (/10X,60HGENERATING INVISCID DATA TAPE FOR BOUNDARY LAYER
1      PROGRAM//)
11    END
12    SUBROUTINE DISKIN
13    IMPLICIT REAL*8 (A-H,O-Z)
14    REAL*8 NOSE
15    DIMENSION RADSI(40), VS(40), XSURFS(40), HXS(40), HMS(40),
16    1USX(40), VXS(40), RADSX(40), RADSX(40), HXSX(40), HXSX(40), X(40),
17    2XX(40), ARADS(40), AVSX(40), ARADSX(40), ARDSXX(40), AHXS(40),
18    340), AUSX(40), AXS(40), FLOELD(14,40)
19    DATA BLUNT, SHARP, 5HBLUNT, 5HSHARP/
20    ICOUNT=0
21    WRITE (30,250)
22    WRITE (30,140)
23    WRITE (30,140)
24    READ (25) ALPHA,KL
25    READ THE FIRST SET OF INVISCID AND BODY DATA
26    READ (25) ISTA,XS,((FLOELD(J,K),J=1,14),K=1,KL)
27    XS = LONGITUDINAL SURFACE COORDINATE
28    FLOELD(1,K) = STREAMWISE SCALE FACTOR
29    FLOELD(2,K) = CROSS SCALE FACTOR
30    FLOELD(3,K) = RADIUS
31    FLOELD(4,K) = SURFACE DISTANCE FROM NOSE
32    FLOELD(5,K) = STREAMWISE VELOCITY COMPONENT
33    FLOELD(6,K) = CROSS FLOW VELOCITY COMPONENT
34    FLOELD(7,K) = AXIAL DISTANCE
35    FLOELD(8,K) = XS DERIVATIVE OF STREAMWISE VELOCITY COMPONENT
36    FLOELD(9,K) = XS DERIVATIVE OF CROSS FLOW VELOCITY COMPONENT
37    FLOELD(10,K) = XS DERIVATIVE OF BODY RADIUS
38    FLOELD(11,K) = 2ND X DERIVATIVE OF BODY RADIUS
39    FLOELD(12,K) = XS DERIVATIVE OF STREAMWISE SCALE FACTOR
40    FLOELD(13,K) = XS DERIVATIVE OF CROSS SCALE FACTOR
41    FLOELD(14,K) = XS DERIVATIVE OF AXIAL DISTANCE
42    KL IS THE NUMBER OF PLANES
43    K=1 IS THE WINDWARD PLANE
44    K=KL IS THE LEeward PLANE
45    M000
46    M000
47    M000
48    M000
49    M000
50    M000
51    M000
52    M000
53    M000
54    M000
55    M000
56    M000
57    M000
58    M000
59    M000
60    M000
61    M000
62    M000
63    M000
64    M000
65    M000
66    M000
67    M000
68    M000
69    M000
70    M000
71    M000
72    M000
73    M000
74    M000
75    M000
76    M000
77    M000
78    M000
79    M000
80    M000
81    M000
82    M000
83    M000
84    M000
85    M000
86    M000
87    M000
88    M000
89    M000
90    M000
91    M000
92    M000
93    M000
94    M000
95    M000
96    M000
97    M000
98    M000
99    M000
100   M000
101   M000
102   M000
103   M000
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107   M000
108   M000
109   M000
110   M000
111   M000
112   M000
113   M000
114   M000
115   M000
116   M000
117   M000
118   M000
119   M000
120   M000
121   M000
122   M000
123   M000
124   M000
125   M000
126   M000
127   M000
128   M000
129   M000
130   M000
131   M000
132   M000
133   M000
134   M000
135   M000
136   M000
137   M000
138   M000
139   M000
140   M000
141   M000
142   M000
143   M000
144   M000
145   M000
146   M000
147   M000
148   M000
149   M000
150   M000
151   M000
152   M000
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C C C
PI=PI*DCOS(PI-1.000)
WRITE (30,140) KL
WRITE (30,150) KL
WRITE (30,140)
ALPH=ALPHA*(180.000/PI)
WRITE (30,160) ALPH
WRITE (30,140)
C C C
UNIT 10 IS THE EDGE PROPERTY DATA SET
WRITE (10) ALPH,KL
WRITE (30,140)
WRITE (30,240)
WRITE (30,140)
C C
WRITE (30,170) ISTA,XS
WRITE (30,140)
WRITE (30,180)
DO 10 K=1,KL
WRITE (30,200) K,(FLOFLD(I,K),I=1,7)
CONTINUE
DO 20 K=1,KL
WRITE (30,190)
DO 20 K=1,KL
K,(FLOFLD(I,K),I=8,14)
WRITE (30,200)
GO TO 60
CONTINUE
C C C
FLOWFIELD DATA FROM THE INVISCID SOLUTION IS READ FROM UNIT 25
READ (25,END=130) ISTA,XS,(FLOFLD(J,K),J=1,14),K=1,KL)
FLOFLD(6,1)=0.000
WRITE (30,170) ISTA,XS
WRITE (30,140)
WRITE (30,180)
DO 40 K=1,KL
WRITE (30,190)
DO 40 K=1,KL
K,(FLOFLD(I,K),I=1,7)
CONTINUE
WRITE (30,190)
DO 50 K=1,KL
WRITE (30,200) K,(FLOFLD(I,K),I=8,14)
CONTINUE
DO 70 K=1,KL
FLOFLD(1,K)
HXS(K)=FLOFLD(2,K)
RADS(K)=FLOFLD(3,K)
XSURF(K)=FLOFLD(4,K)
US(K)=FLOFLD(5,K)

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DATE 05/13/77

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VS(K)=FLOFLD(6,K)
XS(K)=FLOFLD(7,K)
US(K)=FLOFLD(8,K)
VSX(K)=FLOFLD(9,K)
RADSSX(K)=FLOFLD(10,K)
HXSX(K)=FLOFLD(11,K)
HMSX(K)=FLOFLD(12,K)
XX(K)=FLOFLD(13,K)
CONTINUE
VS(1)=0.00
VS(KL)=0.00
VSX(1)=0.00
VSX(KL)=0.00

```

70

80

EDGE PROPERTIES ARE CONVERTED TO FOURIER COEFFICIENTS

```

IF (KL.EQ.1) GO TO 80
CALL FORTER (RADS,ARADS,KL,1)
CALL FORTER (US,AUS,KL,1)
CALL FORTER (VS,AVS,KL,2)
CALL FORTER (HXS,AHXS,KL,1)
CALL FORTER (HMS,AHMS,KL,1)
CALL FORTER (XSURFS,AXSURFS,KL,1)
CALL FORTER (VSX,AVSX,KL,2)
CALL FORTER (RADSSX,ARADSSX,KL,1)
CALL FORTER (HXSX,AHXSX,KL,1)
CALL FORTER (HMSX,AHMSX,KL,1)
CALL FORTER (XX,AXX,KL,1)
GO TO 100
CONTINUE
RADSS(1)=RADSS(1)
AUS(1)=US(1)
AVS(1)=VS(1)
AHXS(1)=HX(1)
AHMS(1)=HM(1)
AXSURFS(1)=SURFS(1)
AVSX(1)=VSX(1)
ARADSSX(1)=RADSSX(1)
AHXSX(1)=HXSX(1)
AHMSX(1)=HMSX(1)
AXX(1)=XX(1)
DO 90 I=2,KL
  ARADS(I)=0.00
  AUS(I)=0.00

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DATE 05/13/77

```

90 AVS(1)=0.000
100 AHXS(1)=0.00
    AHXS(1)=0.00
    AVSRFS(1)=0.00
    AUSX(1)=0.00
    AVSX(1)=0.00
    ARDSX(1)=0.00
    ARDSXX(1)=0.00
    AHXS(1)=0.00
    AHXS(1)=0.00
    AX(1)=0.00
    AX(1)=0.00
    CONTINUE
    CONTINUE
    WRITE (30,210)
    WRITE (30,140)
    WRITE (10) XS, ARADS, AUS, AVS, AHXS, AHWS, AXSRFS, AUSX, AVSX, ARADSX, ARDS
    IXX, AHXSX, AHWSX, AX, AX
    WRITE (30,220)
    DO 110 K=1, KL
    WRITE (30,200) K, AXSRFS(K), ARADS(K), AHXS(K), AHWS(K), AUS(K), AVS(K)
    CONTINUE
    DO 120 K=1, KL
    WRITE (30,230)
    IXX(K), AX(K), AX(K)
    CONTINUE
    WRITE (30,140)
    GO TO 30
    RETURN
C
C
C
C
    FORMAT (1H, 30NUMBER OF PLANES IN THE INVISCID DATA =, I3)
    FORMAT (10X, 6HALPHA=, F6.2)
    FORMAT (10X, 20WALL DATA AT STATION, 1X, I3, 7H, XI =, E9.4)
    FORMAT (10X, 20WALL DATA AT STATION, 1X, I3, 7H, XI =, E9.4)
    FORMAT (10X, 4X, 4HPSI, 11X, 2HHW, 13X, 1HR, 14X, 1HS, 14X, 4HUPSI, 11X,
    12HUM, 13X, 1HX//)
    FORMAT (10X, 4X, 7HD(UPSI), 8X, 5HD(UW), 10X, 4HD(1R), 11X, 5HD(21R), 10X,
    17HD(PSI), 8X, 5HD(HW), 10X, 4HD(XI//)
    FORMAT (4X, 12, 8EL5.6)
    FORMAT (10X, 20FOURIER COEFFICIENTS)
    FORMAT (10X, 1HX, 4X, 6HAXSRFS, 9X, 5HARADS, 10X, 4HAHXS, 11X, 3HAHWS, 11X,
    1HAUS, 12X, 3HAUS//)
    FORMAT (10X, 4X, 4HAUSX, 11X, 4HAUSX, 11X, 4HAUSX, 11X, 4HAUSX, 11X, 4HAUSX, 11X,
    1HAHXSX, 10X, 5HAXSRFS, 10X, 2HAX, 13X, 3HAXX//)
    FORMAT (10X, 58HUNIFORM FLOW STARTING SOLUTION FOR THE INVISCID FLOW
    1W FIELD)

```


DATE 05/13/77

250 FORMAT (40X,52HINVISCID EDGE CONDITIONS FOR BOUNDARY LAYER SOLUTION,192
1N/47X,39HTAKEN FROM THE INVISCID FLOW FIELD DATA)193
END194

CCCCCCCCCCCC
SUBROUTINE FORIER (AR,BR,KL,KK)1
SUBROUTINE FORIER FITS THE DATA IN ARRAY AR WITH EITHER A FOURIER
SINE OR COSINE SERIES.
BR IS THE OUTPUT ARRAY OF FOURIER COEFFICIENTS
KL IS THE NUMBER OF INPUT DATA POINTS
KK = 1 OUTPUTS COEFFICIENTS FOR A COSINE SERIES
KK = 2 OUTPUTS COEFFICIENTS FOR A SINE SERIES
AR IS THE INPUT ARRAY OF FUNCTION VALUES
IMPLICIT REAL*8(A-H,O-Z)
DIMENSION AR(40), BR(40)
PI=3.141592653589793
G=DFLOAT(KL-1)
FAC=2.000/G
IF (KK.EQ.2) GO TO 30
M=KL-1
DO 20 N=1,KL
F=DFLOAT(N-1)
A=0.000
DO 10 K=2,M
E=DFLOAT(K-1)
A=A+AR(K)*DCOS(F*PI*E/G)
CONTINUE
BR(N)=FAC*((AR(1)+AR(KL)*DCOS(F*PI))/2.000)+A)
CONTINUE
BR(1)=BR(1)/2.000
BR(KL)=BR(KL)/2.000
RETURN
CONTINUE
DO 50 N=1,M
F=DFLOAT(N)
B=0.000
DO 40 K=2,M
E=DFLOAT(K-1)
B=B+AR(K)*DSIN(F*PI*E/G)
CONTINUE
BR(N)=FAC*B
CONTINUE
END

APPENDIX VI

Job Control Language for Program ICBL3D

DATE 05/17/77

```
// JOB CARD
/*PRIORITY URGENT
/*JOBPARM LINES=10
//STEPL EXEC FORTGCG,TIME=1,REGION=520K,PARM.FORT='NOSOURCE'
//FORT.SYSIN DD *
```

ICBL3D SOURCE DECK

```
//GO.SYSIN DD *
```

ICBL3D DATA DECK

```
//GO.FT06F001 DD SYSOUT=A
//GO.FT07F001 DD DUMMY
//GO.FT08F001 DD DSN=%%FT08,DISP=(NEW,DELETE),
// UNIT=SYSDA,SPACE=(6472,(61,10),,CONTIG),
// DCB=(RECFM=F,BLKSIZE=6472,DSORG=DA)
//GO.FT10F001 DD DSN=A505F3.INVTAP.EL412,DISP=(OLD,KEEP)
/*
//
```

APPENDIX VII

ICBL3D Sample Input

FILE: PSIR43E DATA A VIRGINIA TECH CONVERSATIONAL MONITOR SYSTEM

4:1 ELLIPSOID AT 2 DEG - - WITHOUT CURVATURE EFFECTS

INVT	61		
KADETA	1		
KENMUZ	10		
KUNSET	1		
KPRIT	1		
KTRANS	1		
LAMTRB	1		
LPRT	1		
NIT1	1		
NIT2	1		
NIT3	1		
NUINJ	1		
SEC	1		
NSOLVE	1		
ADTEST	1		
AKSTAR	1		
ALAMDA	1		
ASTAR	1		
CWALL	1		
CUNV	1		
DXMAX	1		
DXL	1		
EDYLAW	1		
ETAFAC	1		
ETAINE	1		
PR	1		
RIM	1		
T	1		
TCB	1		
AMUNE	1		
PSIAC	1		
PRIME	1		
XBAR	1		
UFS	1		
	1		
	100		

REI	1.90
	9.9983378E-01
	500.
	6306.
	7.650E-02
	2116.80
	1734.3
	1.
	100.

	0.
	.0630
	1.3
	1.5

APPENDIX VIII

ICBL3D Sample Output

THREE-DIMENSIONAL BOUNDARY LAYER PROGRAM
FOR
LAMINAR OR TURBULENT FLOW
WITH
SURFACE CURVATURE EFFECTS
DEVELOPED BY DEPARTMENT
AEROSPACE ENGINEERING AND STATE UNIVERSITY
VIRGINIA POLYTECHNIC INSTITUTE AND STATE UNIVERSITY
BLACKSBURG, VA. 24061

4:1 ELLIPSOID AT 2 DEG - - WITHOUT CURVATURE EFFECTS

FREE STREAM, STAGNATION, AND VEHICLE DATA:

PSTAG = 0.2116800 04 PSTA
PSTAG = 0.5008330 03 DEG.R
PINF = 0.1734300 04 PSTA
PINF = 0.1000000 03 FT/SEC
PINF = 0.5000000 03 DEG.R
RHOINF = 0.7650000-01 SLUGS/FT**3
AMUINF = 0.7650000-01 SLUGS/FT-SEC
TW/TO = 0.9598340 00
ALPHA = 0.2000000 01 DEG.
CP = 0.6006000 04 FT**2/(SEC**2-DEG.R)
REINF = 0.1000000 03
PR = 0.1000000 01
POINTS AT WHICH A SOLUTION IS TO BE OBTAINED:

I XSTAT(I)
1 0.0
2 0.060000
3 0.300000
4 1.500000

X0 = 0.0 R = 0.0 PHI = 0.0 DEG. NIT = 4
XI = 0.0 DXI = 0.0
HX = 0.1000000 U1 HW = 0.0 CWall = 0.0

NONDIMENSIONAL EDGE PROPERTIES

TE = 0.3003000 03 UE = 0.0
DUEX = 0.4280700 01 DVEDX = 0.0
LOCAL EDGE REYNOLDS NUMBER = 0.0 DVEDW = 0.0

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS

CFXINF = 0.0 CFXEDG = 0.0 CFWINF = 0.0 CFWEDG = 0.0
QW = -0.7875010-02 CHIMAX = 0.0

DIMENSIONAL BOUNDARY LAYER PARAMETERS

LONGITUDINAL SKIN FRICTION = 0.0 PSF DELTA*(X) = 0.2753870-01 THETA(X) = 0.1190060-01
TRANSVERSE SKIN FRICTION = 0.0 PSF DELTA*(PHI) = 0.3053990-01 THETA(PHI) = 0.1317640-01

WALL HEAT TRANSFER RATE = -0.774735D 00 BTU DELTA (FT) = 0.102466D 00

ETA 0.0 0.300000 0.1030-01 0.0 0.255816 0.927864 0.0 0.184767 0.618948 0.999834 0.000033 0.0 0.3920-01 0.0 0.900000 0.2050-01 0.467315 0.776115 0.0 0.364616 0.610947 0.999861 0.00009 0.0 0.1480 0.0 0.900000 0.3080-01 0.634652 0.630470 0.0 0.530228 0.510328 0.999887 0.00007 0.0 0.3150 0.0 1.500000 0.4100-01 0.761532 0.489511 0.0 0.672636 0.516594 0.999912 0.00001 0.0 0.2250 0.0 1.800000 0.5130-01 0.852801 0.351330 0.0 0.785901 0.327540 0.999935 0.00003 0.0 0.7000 0.0 2.100000 0.6150-01 0.914439 0.251774 0.0 0.825534 0.230744 0.999971 0.00003 0.0 1.030 0.0 2.400000 0.7180-01 0.953344 0.164568 0.0 0.890655 0.149865 0.999982 0.00003 0.0 1.160 0.0 2.700000 0.8230-01 0.976232 0.056908 0.0 0.980735 0.049056 0.999990 0.00002 0.0 1.200 0.0 3.000000 0.91030 0.995332 0.029883 0.0 0.991245 0.024669 0.999997 0.00003 0.0 2.200 0.0 3.300000 0.1130 0.997911 0.006486 0.0 0.996364 0.024669 0.999997 0.00003 0.0 2.400 0.0 3.600000 0.1130 0.999213 0.002672 0.0 0.998604 0.011367 0.999997 0.00003 0.0 2.600 0.0 3.900000 0.1130 0.999911 0.000354 0.0 0.999933 0.001349 0.999997 0.00003 0.0 2.800 0.0 4.200000 0.1140 0.999913 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.000 0.0 4.500000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.200 0.0 4.800000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.400 0.0 5.100000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.600 0.0 5.400000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.800 0.0 5.700000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.000 0.0 6.000000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.200 0.0 6.300000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.400 0.0 6.600000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.600 0.0 6.900000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.800 0.0 7.200000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.000 0.0 7.500000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.200 0.0 7.800000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.400 0.0 8.100000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.600 0.0 8.400000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.800 0.0 8.700000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 6.000 0.0 9.000000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0

NONDIMENSIONAL EDGE PROPERTIES

TE = 0.300300D 03 UE = 0.0 DUEX = 0.440919D 01 DVEDX = -0.268870D -01

LOCAL EDGE REYNOLDS NUMBER = 0.0

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS

CFXINF = 0.0 CFXEDG = 0.0 CFMINF = 0.0 CFMEDG = 0.0

DIMENSIONAL BOUNDARY LAYER PARAMETERS

LONGITUDINAL SKIN FRICTION = 0.0 PSF

TRANSVERSE SKIN FRICTION = 0.0 PSF

WALL HEAT TRANSFER RATE = -0.798278D 00 BTU

DELTA*(X) = 0.269183D -01

DELTA*(PHI) = 0.192948D -01

DELTA (FT) = 0.102772D 00

THETA(X) = 0.116160D -01

THETA(PHI) = 0.109739D -01

ETA 0.0 0.300000 0.1030-01 0.0 0.255816 0.927864 0.0 0.184767 0.618948 0.999834 0.000033 0.0 0.3920-01 0.0 0.900000 0.2050-01 0.467315 0.776115 0.0 0.364616 0.610947 0.999861 0.00009 0.0 0.1480 0.0 0.900000 0.3080-01 0.634652 0.630470 0.0 0.530228 0.510328 0.999887 0.00007 0.0 0.3150 0.0 1.500000 0.4100-01 0.761532 0.489511 0.0 0.672636 0.516594 0.999912 0.00001 0.0 0.2250 0.0 1.800000 0.5130-01 0.852801 0.351330 0.0 0.785901 0.327540 0.999935 0.00003 0.0 0.7000 0.0 2.100000 0.6150-01 0.914439 0.251774 0.0 0.825534 0.230744 0.999971 0.00003 0.0 1.030 0.0 2.400000 0.7180-01 0.953344 0.164568 0.0 0.890655 0.149865 0.999982 0.00003 0.0 1.160 0.0 2.700000 0.8230-01 0.976232 0.056908 0.0 0.980735 0.049056 0.999990 0.00002 0.0 1.200 0.0 3.000000 0.91030 0.995332 0.029883 0.0 0.991245 0.024669 0.999997 0.00003 0.0 2.200 0.0 3.300000 0.1130 0.997911 0.006486 0.0 0.996364 0.024669 0.999997 0.00003 0.0 2.400 0.0 3.600000 0.1130 0.999213 0.002672 0.0 0.998604 0.011367 0.999997 0.00003 0.0 2.600 0.0 3.900000 0.1130 0.999911 0.000354 0.0 0.999933 0.001349 0.999997 0.00003 0.0 2.800 0.0 4.200000 0.1140 0.999913 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.000 0.0 4.500000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.200 0.0 4.800000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.400 0.0 5.100000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.600 0.0 5.400000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 3.800 0.0 5.700000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.000 0.0 6.000000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.200 0.0 6.300000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.400 0.0 6.600000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.600 0.0 6.900000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 4.800 0.0 7.200000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.000 0.0 7.500000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.200 0.0 7.800000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.400 0.0 8.100000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.600 0.0 8.400000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 5.800 0.0 8.700000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0 6.000 0.0 9.000000 0.1140 0.999932 0.000314 0.0 0.999933 0.001349 0.999997 0.00003 0.0

[illegible]

1.500000	0.4780-01	0.822997	0.250066	0.754698	0.165570	0.999950	0.000059	-0.1770	0.0
1.800000	0.5970-01	0.917088	0.162584	0.970920	0.385774	0.999972	0.000028	-0.1070	0.0
2.100000	0.6970-01	0.955335	0.038135	0.988691	0.033655	0.999983	0.000017	-0.1360	0.0
2.400000	0.7970-01	0.977568	0.054980	0.996382	0.015921	0.999991	0.000009	-0.1650	0.0
2.700000	0.8960-01	0.989519	0.123335	0.999236	0.035245	0.999995	0.000004	-0.1940	0.0
3.000000	0.9950-01	0.995442	0.213470	1.000013	0.011191	0.999998	0.000006	-0.2230	0.0
3.300000	0.1000-00	0.998141	0.305883	1.000019	0.000022	0.999999	0.000003	-0.2520	0.0
3.600000	0.2000-00	0.999271	0.402354	1.000082	0.000031	0.999999	0.000001	-0.2810	0.0
3.900000	0.3000-00	0.999705	0.500871	1.000031	0.000041	0.999999	0.000000	-0.3100	0.0
4.200000	0.4000-00	0.999859	0.600101	1.000033	0.000059	0.999999	0.000000	-0.3390	0.0
4.500000	0.5000-00	0.999911	0.700039	0.999993	0.000079	0.999999	0.000000	-0.3680	0.0
4.800000	0.6000-00	0.999929	0.800017	0.999990	0.000099	0.999999	0.000000	-0.3970	0.0
5.100000	0.7000-00	0.999937	0.900016	0.999992	0.000119	0.999999	0.000000	-0.4260	0.0
5.400000	0.8000-00	0.999943	0.999952	0.999993	0.000139	0.999999	0.000000	-0.4550	0.0
5.700000	0.9000-00	0.999952	0.999962	0.999994	0.000159	0.999999	0.000000	-0.4840	0.0
6.000000	0.1000-00	0.999966	0.999971	0.999996	0.000179	0.999999	0.000000	-0.5130	0.0
6.300000	0.2000-00	0.999976	0.999981	0.999997	0.000199	0.999999	0.000000	-0.5420	0.0
6.600000	0.3000-00	0.999986	0.999990	0.999998	0.000219	0.999999	0.000000	-0.5710	0.0
6.900000	0.4000-00	0.999995	0.999996	0.999999	0.000239	0.999999	0.000000	-0.6000	0.0
7.200000	0.5000-00	0.999999	0.999999	1.000000	0.000259	0.999999	0.000000	-0.6290	0.0
7.500000	0.6000-00	0.999999	0.999999	1.000000	0.000279	0.999999	0.000000	-0.6580	0.0
7.800000	0.7000-00	0.999999	0.999999	1.000000	0.000299	0.999999	0.000000	-0.6870	0.0
8.100000	0.8000-00	0.999999	0.999999	1.000000	0.000319	0.999999	0.000000	-0.7160	0.0
8.400000	0.9000-00	0.999999	0.999999	1.000000	0.000339	0.999999	0.000000	-0.7450	0.0
8.700000	0.1000-00	0.999999	0.999999	1.000000	0.000359	0.999999	0.000000	-0.7740	0.0
9.000000	0.2000-00	0.999999	0.999999	1.000000	0.000379	0.999999	0.000000	-0.8030	0.0

PHI = 60.00 DEG.

NIT = 2

DVEDW = 0.0

VE = 0.0
DUEDW = 0.0

NONDIMENSIONAL EDGE PROPERTIES

TE = 0.3003000 03
DUEDX = 0.4756840 01
LOCAL EDGE REYNOLDS NUMBER = 0.0

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS

CFXINF = 0.0
CFW = -0.8504480-02
CEXEDG = 0.0
CHIMAX = 0.0

DIMENSIONAL BOUNDARY LAYER PARAMETERS

LONGITUDINAL SKIN FRICTION = 0.0
TRANSVERSE SKIN FRICTION = 0.0
WALL HEAT TRANSFER RATE = -0.836662D 00 BTU

ETA	Y	F	FN	G	GN	T	TN	V	EPLUS
0.0	0.0	0.0	0.0	0.0	1.279138	0.999834	0.000092	0.0	0.0
0.300000	0.9730-02	0.257923	0.934920	0.337768	0.975409	0.999861	0.000091	-0.4180-01	0.0
0.600000	0.1950-01	0.470848	0.784612	0.587337	0.697945	0.999888	0.000089	-0.1590 00	0.0
0.900000	0.2920-01	0.639718	0.635624	0.759996	0.466004	0.999914	0.000082	-0.3370 00	0.0
1.200000	0.3890-01	0.767208	0.492387	0.810281	0.237996	0.999938	0.000072	-0.5610 00	0.0
1.500000	0.4860-01	0.858064	0.361523	0.836813	0.163144	0.999957	0.000059	-0.9200 00	0.0
1.800000	0.5840-01	0.918736	0.249608	0.872335	0.033669	0.999973	0.000044	-0.1100 01	0.0
2.100000	0.6810-01	0.956468	0.160968	0.918736	0.038173	0.999984	0.000033	-0.1400 01	0.0
2.400000	0.7820-01	0.978215	0.094422	0.956468	0.015039	0.999991	0.000019	-0.1700 01	0.0
2.700000	0.8750-01	0.989784	0.053414	0.995953	0.004791	0.999996	0.000011	-0.2020 01	0.0

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```

7.500000 0.2340 00 0.999593 0.00272 1.000048 -0.000062 1.000000 0.000000 -0.0550 01 0.0
7.800000 0.2430 00 0.999675 0.00272 1.000031 -0.000050 1.000000 0.000000 -0.0840 01 0.0
8.100000 0.2530 00 0.999756 0.00271 1.000016 -0.000037 1.000000 0.000000 -0.1140 01 0.0
8.400000 0.2620 00 0.999838 0.00271 1.000009 -0.000026 1.000000 0.000000 -0.1440 01 0.0
8.700000 0.2710 00 0.999919 0.00271 1.000003 -0.000015 1.000000 0.000000 -0.1740 01 0.0
9.000000 0.2810 00 1.000000 0.00270 1.000000 -0.000000 1.000000 0.000000 -0.2040 01 0.0

```

XJ = 0.0 R = 0.0
 XI = 0.0 DXI = 0.0
 HX = 0.100000 01 HW = 0.0

NONDIMENSIONAL EDGE PROPERTIES
 TE = 0.300300 03 UE = 0.0
 DUEX = 0.445732 01 DUEX = -0.309671 01
 LOCAL EDGE REYNOLDS NUMBER = 0.0

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS
 CFXINF = 0.0 CFXEDGE = 0.0
 QW = -0.769736 02 CHIMAX = 0.0

DIMENSIONAL BOUNDARY LAYER PARAMETERS
 LONGITUDINAL SKIN FRICTION = 0.0 PSF
 TRANSVERSE SKIN FRICTION = 0.0 PSF
 WALL HEAT TRANSFER RATE = -0.757259 00 BTU

ETA	Y	F	FN	G	GN	T	TN	V	EPLUS
0.0	0.0	0.0	0.918828	0.0	1.197140	0.999834	0.000085	0.0	0.0
0.0	0.1000-01	0.252595	0.767450	0.318785	0.930242	0.999860	0.000085	-0.3440-01	0.0
0.0	0.2010-01	0.460735	0.621592	0.559778	0.633885	0.999885	0.000083	-0.1310 00	0.0
0.0	0.3010-01	0.626173	0.484070	0.731856	0.473609	0.999909	0.000079	-0.2780 00	0.0
1.0	0.4020-01	0.752200	0.360150	0.847252	0.307018	0.999932	0.000070	-0.4630 00	0.0
1.0	0.5030-01	0.843638	0.254454	0.919428	0.184399	0.999951	0.000059	-0.6830 00	0.0
1.0	0.6030-01	0.906463	0.169797	0.961186	0.122497	0.999967	0.000047	-0.9230 00	0.0
2.0	0.7030-01	0.947131	0.106518	0.983295	0.071660	0.999979	0.000034	-1.140 01	0.0
2.0	0.8040-01	0.971832	0.062580	0.993858	0.023233	0.999988	0.000023	-1.440 01	0.0
3.0	0.9040-01	0.985859	0.034333	0.998294	0.009007	0.999993	0.000014	-1.710 01	0.0
3.0	0.1000 00	0.993290	0.017555	0.999851	0.002767	0.999997	0.000008	-1.940 01	0.0
3.0	0.2010 00	0.996957	0.004363	1.000241	0.001466	0.999999	0.000004	-2.250 01	0.0
3.0	0.3010 00	0.998644	0.003720	1.000250	-0.000168	0.999999	0.000002	-2.530 01	0.0
3.0	0.4010 00	0.999371	0.001559	1.000140	-0.000232	1.000000	0.000001	-2.800 01	0.0
4.0	0.5010 00	0.999669	0.000621	1.000119	-0.000163	1.000000	0.000000	-3.080 01	0.0
4.0	0.6010 00	0.999790	0.000261	1.000081	-0.000098	1.000000	0.000000	-3.350 01	0.0
4.0	0.7010 00	0.999842	0.000123	1.000058	-0.000059	1.000000	0.000000	-3.630 01	0.0
5.0	0.8010 00	0.999869	0.000073	1.000036	-0.000038	1.000000	0.000000	-3.900 01	0.0
5.0	0.9010 00	0.999888	0.000045	1.000027	-0.000022	1.000000	0.000000	-4.180 01	0.0
5.0	1.0010 00	0.999902	0.000040	1.000022	-0.000018	1.000000	0.000000	-4.450 01	0.0
6.0	0.0000 00	0.999915	0.000040	1.000022	-0.000015	1.000000	0.000000	-4.730 01	0.0
6.0	0.1010 00	0.999927	0.000035	1.000017	-0.000015	1.000000	0.000000	-5.000 01	0.0
6.0	0.2010 00	0.999937	0.000033	1.000013	-0.000012	1.000000	0.000000	-5.280 01	0.0
7.0	0.3010 00	0.999946	0.000031	1.000010	-0.000010	1.000000	0.000000	-5.550 01	0.0
7.0	0.4010 00	0.999955	0.000029	1.000007	-0.000008	1.000000	0.000000	-5.830 01	0.0
7.0	0.5010 00	0.999963	0.000027	1.000005	-0.000005	1.000000	0.000000	-6.100 01	0.0
7.0	0.6010 00	0.999971	0.000026	1.000003	-0.000005	1.000000	0.000000	-6.370 01	0.0
8.0	0.7010 00	0.999979	0.000025	1.000002	-0.000004	1.000000	0.000000	-6.650 01	0.0
8.0	0.8010 00	0.999986	0.000024	1.000001	-0.000003	1.000000	0.000000	-6.920 01	0.0
8.0	0.9010 00	0.999993	0.000023	1.000000	-0.000002	1.000000	0.000000	-7.200 01	0.0

ETA	Y	F	FN	G	GN	T	f	YN	V	EPLUS
0.0	0.0	0.0	0.931703	0.0	1.251378	0.999834	0.000091	0.0	0.0	0.0
0.0	0.140-01	0.256978	0.781579	0.335752	0.968981	0.998861	0.000091	0.0	0.4070-01	0.0
0.0	0.270-01	0.469001	0.633171	0.583416	0.696945	0.998861	0.000088	0.0	0.1500-01	0.0
0.0	0.340-01	0.637338	0.492981	0.756035	0.467256	0.998113	0.000082	0.0	0.3200-01	0.0
0.0	0.400-01	0.764559	0.361378	0.867442	0.290865	0.998113	0.000072	0.0	0.5200-01	0.0
0.0	0.450-01	0.855585	0.250572	0.934475	0.165322	0.998113	0.000059	0.0	0.7900-01	0.0
0.0	0.500-01	0.916667	0.162609	0.973893	0.082326	0.998113	0.000045	0.0	1.1300-01	0.0
0.0	0.550-01	0.954935	0.098333	0.983966	0.039361	0.998113	0.000031	0.0	1.3600-01	0.0
0.0	0.600-01	0.977203	0.055007	0.993569	0.013076	0.998113	0.000020	0.0	1.6900-01	0.0
0.0	0.650-01	0.989129	0.023857	0.998213	0.003158	0.998113	0.000011	0.0	1.9700-01	0.0
0.0	0.700-01	0.995786	0.013573	1.000013	0.000664	0.998113	0.000006	0.0	2.2800-01	0.0
0.0	0.750-01	0.999318	0.002866	1.000347	0.000198	0.998113	0.000001	0.0	2.5800-01	0.0
0.0	0.800-01	0.999481	0.000927	1.000175	0.000119	0.998113	0.000000	0.0	3.0000-01	0.0
0.0	0.850-01	0.999717	0.000350	1.000129	0.000074	0.998113	0.000000	0.0	3.5000-01	0.0
0.0	0.900-01	0.999749	0.000084	1.000101	0.000032	0.998113	0.000000	0.0	4.1300-01	0.0
0.0	0.950-01	0.999771	0.000036	1.000069	0.000017	0.998113	0.000000	0.0	4.7500-01	0.0
0.0	1.000-01	0.999789	0.000019	1.000057	0.000009	0.998113	0.000000	0.0	5.0600-01	0.0
0.0	1.050-01	0.999802	0.000009	1.000048	0.000004	0.998113	0.000000	0.0	5.3700-01	0.0
0.0	1.100-01	0.999822	0.000005	1.000039	0.000003	0.998113	0.000000	0.0	5.6900-01	0.0
0.0	1.150-01	0.999842	0.000003	1.000031	0.000002	0.998113	0.000000	0.0	5.9900-01	0.0
0.0	1.200-01	0.999863	0.000002	1.000025	0.000001	0.998113	0.000000	0.0	6.6600-01	0.0
0.0	1.250-01	0.999877	0.000001	1.000019	0.000000	0.998113	0.000000	0.0	7.2300-01	0.0
0.0	1.300-01	0.999895	0.000000	1.000014	0.000000	0.998113	0.000000	0.0	7.7200-01	0.0
0.0	1.350-01	0.999912	0.000000	1.000010	0.000000	0.998113	0.000000	0.0	8.1800-01	0.0
0.0	1.400-01	0.999930	0.000000	1.000006	0.000000	0.998113	0.000000	0.0	8.7500-01	0.0
0.0	1.450-01	0.999947	0.000000	1.000002	0.000000	0.998113	0.000000	0.0	9.3100-01	0.0
0.0	1.500-01	0.999965	0.000000	1.000000	0.000000	0.998113	0.000000	0.0	9.8400-01	0.0
0.0	1.550-01	0.999982	0.000000							

XU = 0.0 R = 0.0 PHI = 180.00 DEG.
 XI = 0.0 DXI = 0.0
 HK = 0.1000000 01 HW = 0.0
 NIT = 3
 NONDIMENSIONAL EDGE PROPERTIES
 TE = 0.3003000 03 UE = 0.0
 DUEX = 0.3380120 01 DUEX = 0.2614780-15
 DUEX = 0.0
 LOCAL EDGE REYNOLDS NUMBER = 0.0
 NONDIMENSIONAL BOUNDARY LAYER PARAMETERS
 CFXINF = 0.0 CFXEDG = 0.0
 CFXM = -0.6646620-02 CFXIMAX = 0.0
 DIMENSIONAL BOUNDARY LAYER PARAMETERS
 LONGITUDINAL SKIN FRICTION = 0.0 PSF
 TRANSVERSE SKIN FRICTION = 0.0 PSF
 WALL HEAT TRANSFER RATE = -0.6936880 00 BTU

DELTA*(X) = 0.3254310-01 THETA(X) = 0.1497100-01
 DELTA*(PHI) = 0.3637600-01 THETA(PHI) = 0.1583910-01
 DELTA (FT) = 0.1261040 00

ETA	X	Y	F	FN	G	GN	T	TN	V	EPLUS
0.0	0.0	0.0	0.2	0.926670	0.0	0.05461	0.999834	0.000089	0.0	0.0
0.0	0.0	0.0	0.255522	0.775962	0.161923	0.03551	0.999860	0.000089	-0.0380-01	0.0
0.0	0.0	0.0	0.466335	0.623497	0.350132	0.076361	0.999887	0.000086	-0.1470 00	0.0
0.0	0.0	0.0	0.633728	0.488902	0.525181	0.15961	0.999912	0.000081	-0.3110 00	0.0
0.0	0.0	0.0	0.760645	0.361206	0.667679	0.428987	0.999935	0.000071	-0.5110 00	0.0
0.0	0.0	0.0	0.851855	0.252092	0.781656	0.329837	0.999955	0.000059	-0.7570 00	0.0
0.0	0.0	0.0	0.913581	0.165190	0.865848	0.233970	0.999970	0.000045	-0.1300 01	0.0
0.0	0.0	0.0	0.932695	0.101112	0.923177	0.152881	0.999982	0.000032	-0.1300 01	0.0
0.0	0.0	0.0	0.975797	0.075669	0.991113	0.091908	0.999990	0.000021	-0.1500 01	0.0
0.0	0.0	0.0	0.988466	0.030383	0.99826	0.050760	0.999995	0.000012	-0.1870 01	0.0
0.0	0.0	0.0	0.994896	0.014822	0.990794	0.025760	0.999999	0.000007	-0.2450 01	0.0
0.0	0.0	0.0	0.997910	0.006667	0.996122	0.011985	0.999999	0.000003	-0.2700 01	0.0
0.0	0.0	0.0	0.999210	0.002759	0.998496	0.005109	1.000000	0.000002	-0.3300 01	0.0
0.0	0.0	0.0	0.999725	0.001048	0.999464	0.001993	1.000000	0.000000	-0.3300 01	0.0
0.0	0.0	0.0	0.999913	0.000364	0.999825	0.000710	1.000000	0.000000	-0.3300 01	0.0
0.0	0.0	0.0	0.999975	0.000115	0.999948	0.000251	1.000000	0.000000	-0.3600 01	0.0
0.0	0.0	0.0	0.999994	0.000033	0.999986	0.000068	1.000000	0.000000	-0.3900 01	0.0
0.0	0.0	0.0	0.999999	0.000008	0.999997	0.000018	1.000000	0.000000	-0.4200 01	0.0
0.0	0.0	0.0	1.000000	0.000002	0.999999	0.000004	1.000000	0.000000	-0.4800 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.5100 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.5100 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.5700 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.6300 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.6800 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.6800 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.7190 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.7490 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.7780 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.8000 01	0.0
0.0	0.0	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	-0.8000 01	0.0

 PHI = 0.0 DEG.
 CFWALL = 0.0
 NIT = 3

XU = 0.3132600-02 R = 0.1972310-01
 XI = 0.2000000-01 DXI = 0.0
 HK = 0.1000000 01 HW = 0.1239320 00
 NIT = 3

NONDIMENSIONAL EDGE PROPERTIES
 TE = 0.3003000 03 VE = 0.0
 DUEOX = 0.4190000 01 DUEOW = 0.4132900-02
 LOCAL EDGE REYNOLDS NUMBER = 0.1647760 00

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS
 CFXINF = 0.4390240-01 CFXEDG = 0.1855580 01
 QW = -0.8500930-02 CHIMAX = 0.1625710-02
 CFWENG = 0.0

DIMENSIONAL BOUNDARY LAYER PARAMETERS
 LONGITUDINAL SKIN FRICTION = 0.1679270 02 PSF
 TRANSVERSE SKIN FRICTION = 0.0 PSF
 WALL HEAT TRANSFER RATE = -0.8363130 00 BTU

DELTA*(X) = 0.2804930-01
 DELTA*(Y) = 0.2933520-01
 DELTA*(Z) = 0.1042720 00
 THETA(X) = 0.1211560-01
 THETA(Y) = 0.1331460-01

ETA	X	Y	F	FN	G	GN	T	TN	V	EPLUS
0.0	0.0	0.0	0.0	0.923241	0.0	0.733358	0.999834	0.000039	0.0	0.0
0.1	0.1050-01	0.0	0.255990	0.778292	0.214103	0.537309	0.999863	0.000039	0.3947-01	0.0
0.2	0.2090-01	0.0	0.467237	0.630803	0.403582	0.437809	0.999890	0.000038	0.1490 00	0.0
0.3	0.3140-01	0.0	0.634961	0.489744	0.560225	0.494544	0.999915	0.000038	0.3160 00	0.0
0.4	0.4180-01	0.0	0.762041	0.361734	0.699889	0.395359	0.999938	0.000037	0.5270 00	0.0
0.5	0.5230-01	0.0	0.853227	0.251726	0.803822	0.298308	0.999957	0.000037	0.7710 00	0.0
0.6	0.6270-01	0.0	0.914771	0.164409	0.879564	0.209408	0.999972	0.000034	1.0700 01	0.0
0.7	0.7320-01	0.0	0.953617	0.101166	0.930514	0.135634	0.999980	0.000031	1.3320 01	0.0
0.8	0.8360-01	0.0	0.976437	0.026681	0.963129	0.042507	0.999990	0.000020	1.6100 01	0.0
0.9	0.9410-01	0.0	0.988865	0.002689	0.981756	0.005792	0.999995	0.000012	1.9100 01	0.0
1.0	1.0450 00	0.0	0.995120	0.000349	0.991656	0.002393	0.999998	0.000006	2.2510 01	0.0
1.1	1.1500 00	0.0	0.998022	0.000186	0.996582	0.001866	0.999998	0.000004	2.5410 01	0.0
1.2	1.2500 00	0.0	0.999260	0.000074	0.998916	0.001338	1.000000	0.000001	2.8100 01	0.0
1.3	1.3500 00	0.0	0.999745	0.000036	0.999442	0.000943	1.000000	0.000000	3.0410 01	0.0
1.4	1.4500 00	0.0	0.999919	0.000016	0.999953	0.000528	1.000000	0.000000	3.2710 01	0.0
1.5	1.5700 00	0.0	0.999977	0.000008	0.999997	0.000316	1.000000	0.000000	3.4310 01	0.0
1.6	1.6700 00	0.0	0.999994	0.000004	0.999999	0.000216	1.000000	0.000000	3.5310 01	0.0
1.7	1.7800 00	0.0	0.999998	0.000002	0.999999	0.000164	1.000000	0.000000	3.5720 01	0.0
1.8	1.8800 00	0.0	1.000000	0.000001	1.000000	0.000110	1.000000	0.000000	3.5720 01	0.0
1.9	1.9900 00	0.0	1.000000	0.000000	1.000000	0.000080	1.000000	0.000000	3.5200 01	0.0
2.0	2.0900 00	0.0	1.000000	0.000000	1.000000	0.000050	1.000000	0.000000	3.3520 01	0.0
2.1	2.1900 00	0.0	1.000000	0.000000	1.000000	0.000030	1.000000	0.000000	3.1200 01	0.0
2.2	2.2900 00	0.0	1.000000	0.000000	1.000000	0.000020	1.000000	0.000000	2.8420 01	0.0
2.3	2.3900 00	0.0	1.000000	0.000000	1.000000	0.000010	1.000000	0.000000	2.5720 01	0.0
2.4	2.4900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	2.3020 01	0.0
2.5	2.5900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	2.0320 01	0.0
2.6	2.6900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	1.7630 01	0.0
2.7	2.7900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	1.4930 01	0.0
2.8	2.8900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	1.2230 01	0.0
2.9	2.9900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	0.9530 01	0.0
3.0	3.0900 00	0.0	1.000000	0.000000	1.000000	0.000000	1.000000	0.000000	0.6830 01	0.0

 NIT = 3
 PHI = 20.00 DEG.
 CWall = 0.0

NONDIMENSIONAL EDGE PROPERTIES
 XO = 0.3065410-02
 XI = 0.3003000 03
 HX = 0.1000230 01
 TE = 0.3003000 03 VE = 0.0
 DUEOX = 0.4190000 01 DUEOW = 0.4132900-02
 LOCAL EDGE REYNOLDS NUMBER = 0.1647760 00

LOCAL EDGE REYNOLDS NUMBER = 0.1662590 00

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS

CFXINF= 0.4666430-01 CFXEDG= 0.1863830 01
QW = -0.8574310-02 CHIMAX= 0.1626640-02

DIMENSIONAL BOUNDARY LAYER PARAMETERS

LONGITUDINAL SKIN FRICTION= 0.1739410 02 PSF
TRANSVERSE SKIN FRICTION = 0.6424110-01 PSF
WALL HEAT TRANSFER RATE = -0.3433510 00 BTU

CFWINF= 0.1679510-03 CFWEDG= 0.1679510-03

DELTA*(X) = 0.2782210-01 THETA(X) = 0.1202030-01
DELTA*(PHI) = 0.1698280-01 THETA(PHI) = 0.3554630-02
DELTA (FT) = 0.1035290 00

ETA	Y	F	FN	G	GN	T	TN	V	FPLUS
0.0	0.0	0.0	0.31517	0.0	0.03504	0.999834	0.00099	0.0	0.0
0.300000	0.1040-01	0.256903	0.783902	0.00913	0.002591	0.999863	0.00099	0.0	0.0
0.600000	0.2080-01	0.468687	0.632304	0.001563	0.001791	0.999890	0.00099	0.0	0.0
0.900000	0.3120-01	0.636609	0.483925	0.001594	0.001134	0.999915	0.00099	0.0	0.0
1.200000	0.4160-01	0.763621	0.369879	0.002257	0.000665	0.999938	0.00099	0.0	0.0
1.500000	0.5200-01	0.854567	0.253763	0.002404	0.000354	0.999957	0.00099	0.0	0.0
1.800000	0.6240-01	0.915794	0.163329	0.002479	0.000169	0.999972	0.00099	0.0	0.0
2.100000	0.7280-01	0.954326	0.091922	0.002512	0.000069	0.999983	0.00099	0.0	0.0
2.400000	0.8320-01	0.976884	0.029170	0.002527	0.000024	0.999991	0.00099	0.0	0.0
2.700000	0.9370-01	0.989122	0.014335	0.002527	0.000011	0.999998	0.00099	0.0	0.0
3.000000	0.1040 00	0.995255	0.006214	0.002527	0.000002	0.999999	0.00099	0.0	0.0
3.300000	0.1140 00	0.998086	0.002527	0.002526	0.000001	1.000000	0.00099	0.0	0.0
3.600000	0.1250 00	0.999288	0.001526	0.002526	0.000001	1.000000	0.00099	0.0	0.0
3.900000	0.1350 00	0.999755	0.000321	0.002526	0.000000	1.000000	0.00099	0.0	0.0
4.200000	0.1460 00	0.999922	0.000109	0.002526	0.000000	1.000000	0.00099	0.0	0.0
4.500000	0.1560 00	0.999977	0.000029	0.002526	0.000000	1.000000	0.00099	0.0	0.0
4.800000	0.1660 00	0.999993	0.000008	0.002526	0.000000	1.000000	0.00099	0.0	0.0
5.100000	0.1770 00	0.999999	0.000002	0.002526	0.000000	1.000000	0.00099	0.0	0.0
5.400000	0.1870 00	0.999999	0.000001	0.002526	0.000000	1.000000	0.00099	0.0	0.0
5.700000	0.1980 00	0.999999	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
6.000000	0.2080 00	0.999999	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
6.300000	0.2190 00	0.999999	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
6.600000	0.2290 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
6.900000	0.2390 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
7.200000	0.2500 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
7.500000	0.2600 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
7.800000	0.2710 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
8.100000	0.2810 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
8.400000	0.2910 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
8.700000	0.3020 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0
9.000000	0.3120 00	1.000000	0.000000	0.002526	0.000000	1.000000	0.00099	0.0	0.0

X0 = 0.2871000-02 R = 0.1978850-01 PHI = 40.00 DEG.
XI = 0.2000000-01 DXI = 0.2000000-01 NIT = 2
HX = 0.1000900 01 HW = 0.1243450 00 CWall = 0.0

NONDIMENSIONAL EDGE PROPERTIES

TE = 0.3003000 03 UE = 0.8411200-01
DUEOX= 0.4268730 01 DVEDX=-0.1588990-01

LOCAL EDGE REYNOLDS NUMBER = 0.1682240 00

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS

DVFDW=-0.1076200-02

CFXINF= 0.4254980-01 CFXEDG= 0.1857430-01 CFWEDG= 0.2099880-03
 QW = -0.8672120-02 CHIMAX= 0.1633740-02

DIMENSIONAL BOUNDARY LAYER PARAMETERS

LONGITUDINAL SKIN FRICTION= 0.1742280 02 PSF
 TRANSVERSE SKIN FRICTION = 0.3032040-01 PSF
 WALL HEAT TRANSFER RATE = -0.8531540 00 BTU

DELTA*(X) = 0.2763040-01
 DELTA*(PHI) = 0.1823190-01
 DELTA (FT) = 0.1025050 00

T-ETA(X) = 0.1194230-01
 T-ETA(PHI) = 0.9203520-02

ETA	Y	F	FN	G	GN	T	TN	V	EPLUS
0.0	0.0	0.0	0.933618	0.0	0.004334	0.999534	0.000102	0.0	0.0
0.1	0.0300	0.257234	0.731710	0.001129	0.002272	0.999534	0.000097	0.0	0.0
0.2	0.0600	0.499234	0.632231	0.001566	0.002272	0.999534	0.000083	0.0	0.0
0.3	0.0900	0.737145	0.539751	0.002006	0.001494	0.999534	0.000070	0.0	0.0
0.4	0.1200	0.964319	0.454319	0.002506	0.000916	0.999534	0.000057	0.0	0.0
0.5	0.1500	1.185497	0.380697	0.003069	0.000520	0.999534	0.000043	0.0	0.0
0.6	0.1800	1.401612	0.313182	0.003625	0.000318	0.999534	0.000030	0.0	0.0
0.7	0.2100	1.616012	0.250721	0.004182	0.000200	0.999534	0.000017	0.0	0.0
0.8	0.2400	1.829449	0.193484	0.004749	0.000154	0.999534	0.000011	0.0	0.0
0.9	0.2700	2.041488	0.141384	0.005325	0.000106	0.999534	0.000006	0.0	0.0
1.0	0.3000	2.252559	0.096191	0.005916	0.000071	0.999534	0.000003	0.0	0.0
1.1	0.3300	2.462777	0.052519	0.006525	0.000043	0.999534	0.000001	0.0	0.0
1.2	0.3600	2.672144	0.019741	0.007179	0.000025	0.999534	0.000000	0.0	0.0
1.3	0.3900	2.880712	0.000324	0.007879	0.000013	0.999534	0.000000	0.0	0.0
1.4	0.4200	3.088567	0.000104	0.008629	0.000007	0.999534	0.000000	0.0	0.0
1.5	0.4500	3.295885	0.000033	0.009429	0.000004	0.999534	0.000000	0.0	0.0
1.6	0.4800	3.502730	0.000006	0.010279	0.000002	0.999534	0.000000	0.0	0.0
1.7	0.5100	3.709144	0.000001	0.011179	0.000001	0.999534	0.000000	0.0	0.0
1.8	0.5400	3.915144	0.000000	0.012129	0.000000	0.999534	0.000000	0.0	0.0
1.9	0.5700	4.120744	0.000000	0.013129	0.000000	0.999534	0.000000	0.0	0.0
2.0	0.6000	4.325944	0.000000	0.014179	0.000000	0.999534	0.000000	0.0	0.0
2.1	0.6300	4.530744	0.000000	0.015279	0.000000	0.999534	0.000000	0.0	0.0
2.2	0.6600	4.735144	0.000000	0.016429	0.000000	0.999534	0.000000	0.0	0.0
2.3	0.6900	4.939144	0.000000	0.017629	0.000000	0.999534	0.000000	0.0	0.0
2.4	0.7200	5.142744	0.000000	0.018879	0.000000	0.999534	0.000000	0.0	0.0
2.5	0.7500	5.345944	0.000000	0.020179	0.000000	0.999534	0.000000	0.0	0.0
2.6	0.7800	5.548744	0.000000	0.021529	0.000000	0.999534	0.000000	0.0	0.0
2.7	0.8100	5.751144	0.000000	0.022929	0.000000	0.999534	0.000000	0.0	0.0
2.8	0.8400	5.953144	0.000000	0.024379	0.000000	0.999534	0.000000	0.0	0.0
2.9	0.8700	6.154744	0.000000	0.025879	0.000000	0.999534	0.000000	0.0	0.0
3.0	0.9000	6.355944	0.000000	0.027429	0.000000	0.999534	0.000000	0.0	0.0
3.1	0.9300	6.556744	0.000000	0.029029	0.000000	0.999534	0.000000	0.0	0.0
3.2	0.9600	6.757144	0.000000	0.030679	0.000000	0.999534	0.000000	0.0	0.0
3.3	0.9900	6.957144	0.000000	0.032379	0.000000	0.999534	0.000000	0.0	0.0
3.4	1.0200	7.156744	0.000000	0.034129	0.000000	0.999534	0.000000	0.0	0.0
3.5	1.0500	7.355944	0.000000	0.035929	0.000000	0.999534	0.000000	0.0	0.0
3.6	1.0800	7.554744	0.000000	0.037779	0.000000	0.999534	0.000000	0.0	0.0
3.7	1.1100	7.753144	0.000000	0.039679	0.000000	0.999534	0.000000	0.0	0.0
3.8	1.1400	7.951144	0.000000	0.041629	0.000000	0.999534	0.000000	0.0	0.0
3.9	1.1700	8.148744	0.000000	0.043629	0.000000	0.999534	0.000000	0.0	0.0
4.0	1.2000	8.345944	0.000000	0.045679	0.000000	0.999534	0.000000	0.0	0.0
4.1	1.2300	8.542744	0.000000	0.047779	0.000000	0.999534	0.000000	0.0	0.0
4.2	1.2600	8.739144	0.000000	0.049929	0.000000	0.999534	0.000000	0.0	0.0
4.3	1.2900	8.935144	0.000000	0.052129	0.000000	0.999534	0.000000	0.0	0.0
4.4	1.3200	9.130744	0.000000	0.054379	0.000000	0.999534	0.000000	0.0	0.0
4.5	1.3500	9.325944	0.000000	0.056679	0.000000	0.999534	0.000000	0.0	0.0
4.6	1.3800	9.520744	0.000000	0.059029	0.000000	0.999534	0.000000	0.0	0.0
4.7	1.4100	9.715144	0.000000	0.061429	0.000000	0.999534	0.000000	0.0	0.0
4.8	1.4400	9.909144	0.000000	0.063879	0.000000	0.999534	0.000000	0.0	0.0
4.9	1.4700	10.102744	0.000000	0.066379	0.000000	0.999534	0.000000	0.0	0.0
5.0	1.5000	10.295944	0.000000	0.068929	0.000000	0.999534	0.000000	0.0	0.0
5.1	1.5300	10.488744	0.000000	0.071529	0.000000	0.999534	0.000000	0.0	0.0
5.2	1.5600	10.681144	0.000000	0.074179	0.000000	0.999534	0.000000	0.0	0.0
5.3	1.5900	10.873144	0.000000	0.076879	0.000000	0.999534	0.000000	0.0	0.0
5.4	1.6200	11.064744	0.000000	0.079629	0.000000	0.999534	0.000000	0.0	0.0
5.5	1.6500	11.255944	0.000000	0.082429	0.000000	0.999534	0.000000	0.0	0.0
5.6	1.6800	11.446744	0.000000	0.085279	0.000000	0.999534	0.000000	0.0	0.0
5.7	1.7100	11.637144	0.000000	0.088179	0.000000	0.999534	0.000000	0.0	0.0
5.8	1.7400	11.827144	0.000000	0.091129	0.000000	0.999534	0.000000	0.0	0.0
5.9	1.7700	12.016744	0.000000	0.094129	0.000000	0.999534	0.000000	0.0	0.0
6.0	1.8000	12.205944	0.000000	0.097179	0.000000	0.999534	0.000000	0.0	0.0
6.1	1.8300	12.394744	0.000000	0.100279	0.000000	0.999534	0.000000	0.0	0.0
6.2	1.8600	12.583144	0.000000	0.103429	0.000000	0.999534	0.000000	0.0	0.0
6.3	1.8900	12.771144	0.000000	0.106629	0.000000	0.999534	0.000000	0.0	0.0
6.4	1.9200	12.958744	0.000000	0.109879	0.000000	0.999534	0.000000	0.0	0.0
6.5	1.9500	13.145944	0.000000	0.113179	0.000000	0.999534	0.000000	0.0	0.0
6.6	1.9800	13.332744	0.000000	0.116529	0.000000	0.999534	0.000000	0.0	0.0
6.7	2.0100	13.519144	0.000000	0.119929	0.000000	0.999534	0.000000	0.0	0.0
6.8	2.0400	13.705144	0.000000	0.123379	0.000000	0.999534	0.000000	0.0	0.0
6.9	2.0700	13.890744	0.000000	0.126879	0.000000	0.999534	0.000000	0.0	0.0
7.0	2.1000	14.075944	0.000000	0.130429	0.000000	0.999534	0.000000	0.0	0.0
7.1	2.1300	14.260744	0.000000	0.134029	0.000000	0.999534	0.000000	0.0	0.0
7.2	2.1600	14.445144	0.000000	0.137679	0.000000	0.999534	0.000000	0.0	0.0
7.3	2.1900	14.629144	0.000000	0.141379	0.000000	0.999534	0.000000	0.0	0.0
7.4	2.2200	14.812744	0.000000	0.145129	0.000000	0.999534	0.000000	0.0	0.0
7.5	2.2500	15.0000	0.000000	0.148929	0.000000	0.999534	0.000000	0.0	0.0
7.6	2.2800	15.186744	0.000000	0.152779	0.000000	0.999534	0.000000	0.0	0.0
7.7	2.3100	15.372744	0.000000	0.156679	0.000000	0.999534	0.000000	0.0	0.0
7.8	2.3400	15.558144	0.000000	0.160629	0.000000	0.999534	0.000000	0.0	0.0
7.9	2.3700	15.742744	0.000000	0.164629	0.000000	0.999534	0.000000	0.0	0.0
8.0	2.4000	15.926744	0.000000	0.168679	0.000000	0.999534	0.000000	0.0	0.0
8.1	2.4300	16.110144	0.000000	0.172779	0.000000	0.999534	0.000000	0.0	0.0
8.2	2.4600	16.292744	0.000000	0.176929	0.000000	0.999534	0.000000	0.0	0.0
8.3	2.4900	16.474744	0.000000	0.181129	0.000000	0.999534	0.000000	0.0	0.0
8.4	2.5200	16.656144	0.000000	0.185379	0.000000	0.999534	0.000000	0.0	0.0
8.5	2.5500	16.836744	0.000000	0.189679	0.000000	0.999534	0.000000	0.0	0.0
8.6	2.5800	17.016744	0.000000	0.194029	0.000000	0.999534	0.000000	0.0	0.0
8.7	2.6100	17.195944	0.000000	0.198429	0.000000	0.999534	0.000000	0.0	0.0
8.8	2.6400	17.374744	0.000000	0.202879	0.000000	0.999534	0.000000	0.0	0.0
8.9	2.6700	17.552744	0.000000	0.207379	0.000000	0.999534	0.000000	0.0	0.0
9.0	2.7000	17.730144	0.000000	0.211929	0.000000	0.999534	0.000000	0.0	0.0
9.1	2.7300	17.906744	0.000000	0.216529	0.000000	0.999534	0.000000	0.0	0.0
9.2	2.7600	18.082744	0.000000	0.221179	0.000000	0.999534	0.000000	0.0	0.0
9.3	2.7900	18.258144	0.000000	0.225879	0.000000	0.999534	0.000000	0.0	0.0
9.4	2.8200	18.432744	0.000000	0.230629	0.000000	0.999534	0.000000	0.0	0.0
9.5	2.8500	18.606744	0.000000	0.235429	0.000000	0.999534	0.000000	0.0	0.0
9.6	2.8800	18.780144	0.000000	0.240279	0.000000	0.999534	0.000000	0.0	0.0
9.7	2.9100	18.952744	0.000000	0.245179	0.000000	0.999534	0.000000	0.0	0.0
9.8	2.9400	19.124744	0.000000	0.250129	0.000000	0.999534	0.000000	0.0	0.0
9.9	2.9700	19.295944	0.000000	0.255129	0.000000	0.999534	0.000000	0.0	0.0
10.0	3.0000	19.466744	0.000000	0.260179	0.000000	0.999534</			

DELTA*(X) = 0.273060-01
 DELTA*(Y) = 0.522810-01
 DELTA*(Z) = 0.101900-00
 THETA(X) = 0.118110-01
 THETA(Y) = 0.976520-02

LONGITUDINAL SKIN FRICTION = 0.179420 02 PSF
 TRANSVERSE SKIN FRICTION = -0.575460-02 PSF
 WALL HEAT TRANSFER RATE = -0.669370 00 BTU

ETA	Y	F	FN	G	GN	I	IN	V	EPLUS
0.0	0.0	0.0	0.36077	0.0	0.00300	0.99994	0.00300	0.0	0.0
0.0	0.1	0.25793	0.73473	0.0	0.00300	0.99994	0.00300	0.0	0.0
0.0	0.2	0.47029	0.83291	0.0	0.00300	0.99994	0.00300	0.0	0.0
0.0	0.3	0.63825	0.89444	0.0	0.00300	0.99994	0.00300	0.0	0.0
1.0	0.4	0.76504	0.93444	0.0	0.00300	0.99994	0.00300	0.0	0.0
1.0	0.5	0.85261	0.95607	0.0	0.00300	0.99994	0.00300	0.0	0.0
1.0	0.6	0.91626	0.96735	0.0	0.00300	0.99994	0.00300	0.0	0.0
2.0	0.7	0.95473	0.97119	0.0	0.00300	0.99994	0.00300	0.0	0.0
2.0	0.8	0.97713	0.97486	0.0	0.00300	0.99994	0.00300	0.0	0.0
3.0	0.9	0.98929	0.97825	0.0	0.00300	0.99994	0.00300	0.0	0.0
3.0	1.0	0.99265	0.98118	0.0	0.00300	0.99994	0.00300	0.0	0.0
3.0	1.1	0.99533	0.98394	0.0	0.00300	0.99994	0.00300	0.0	0.0
3.0	1.2	0.99741	0.98644	0.0	0.00300	0.99994	0.00300	0.0	0.0
3.0	1.3	0.99870	0.98873	0.0	0.00300	0.99994	0.00300	0.0	0.0
4.0	1.4	0.99929	0.99081	0.0	0.00300	0.99994	0.00300	0.0	0.0
4.0	1.5	0.99951	0.99194	0.0	0.00300	0.99994	0.00300	0.0	0.0
4.0	1.6	0.99962	0.99266	0.0	0.00300	0.99994	0.00300	0.0	0.0
5.0	1.7	0.99968	0.99309	0.0	0.00300	0.99994	0.00300	0.0	0.0
5.0	1.8	0.99973	0.99341	0.0	0.00300	0.99994	0.00300	0.0	0.0
5.0	1.9	0.99978	0.99365	0.0	0.00300	0.99994	0.00300	0.0	0.0
6.0	2.0	0.99982	0.99381	0.0	0.00300	0.99994	0.00300	0.0	0.0
6.0	2.1	0.99986	0.99390	0.0	0.00300	0.99994	0.00300	0.0	0.0
6.0	2.2	0.99989	0.99396	0.0	0.00300	0.99994	0.00300	0.0	0.0
7.0	2.3	0.99992	0.99400	0.0	0.00300	0.99994	0.00300	0.0	0.0
7.0	2.4	0.99994	0.99403	0.0	0.00300	0.99994	0.00300	0.0	0.0
7.0	2.5	0.99996	0.99405	0.0	0.00300	0.99994	0.00300	0.0	0.0
8.0	2.6	0.99998	0.99407	0.0	0.00300	0.99994	0.00300	0.0	0.0
8.0	2.7	0.99999	0.99408	0.0	0.00300	0.99994	0.00300	0.0	0.0
8.0	2.8	1.00000	0.99409	0.0	0.00300	0.99994	0.00300	0.0	0.0
9.0	2.9	1.00000	0.99410	0.0	0.00300	0.99994	0.00300	0.0	0.0
9.0	3.0	1.00000	0.99411	0.0	0.00300	0.99994	0.00300	0.0	0.0

NIT = 3

PHI = 90.00 DEG.
 CWall = 0.0

X0 = 0.219730-02
 X1 = 0.200000-01
 X2 = 0.1003260 01
 R = 0.1394120-01
 DXI = 0.200000-01
 HW = 0.1254050 00

NONDIMENSIONAL EDGE PROPERTIES

TE = 0.300300 03
 DUEOX = 0.4575600 01
 LOCAL EDGE REYNOLDS NUMBER = 0.1757800 00

DVFDW = -0.1204870-01

NONDIMENSIONAL BOUNDARY LAYER PARAMETERS

CFXINF = 0.4901100-01
 QW = -0.9017370-02
 CFXEDG = 0.1680980 01
 CHIMAX = 0.1656250-02

CFWINF = -0.5509060-03
 CFWEDG = -0.5509060-03

DIMENSIONAL BOUNDARY LAYER PARAMETERS

LONGITUDINAL SKIN FRICTION = 0.1874670 02 PSF
 TRANSVERSE SKIN FRICTION = -0.2107220 00 PSF
 WALL HEAT TRANSFER RATE = -0.6871200 00 BTU

DELTA*(X) = 0.2692230-01
 DELTA*(Y) = 0.1009880-01
 DELTA*(Z) = 0.1008310 00
 THETA(X) = 0.1186070-01
 THETA(Y) = 0.976520-02

APPENDIX IX

Program ICBL3D Data FORMAT Sheets

IBM

FORTRAN Coding Form

GX28-7327-6 U/M 050**
Printed in U.S.A.

PROGRAM PROGRAMMER	ICBL3D	DATE	GRAPHIC PUNCH	PAGE OF	CARD ELECTED NUMBER
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STATEMENT NUMBER		STATEMENT		STATEMENT SEQUENCE
CARD 4				
			KADETA	
			(49X, 13)	
			INDICATOR FOR THE ADJUSTMENT OF THE TRANSFORMED NORMAL COORDINATE	
CARD 5				
			KEND2	
			(49X, 13)	
			NUMBER OF CIRCUMFERENTIAL PLANES	
CARD 6				
			KONSET	
			(49X, 13)	
			SUBSCRIPT OF XSTA ARRAY, LOCATES ONSET OF TRANSITION	

**Number of forms per card may vary slightly

FORTRAN STATEMENT		IDENTIFICATION SEQUENCE
CARD 7	KPRT <input type="text"/> (49X, 13) PARAMETER WHICH CONTROLS PRINTING OF PROFILES IN ϕ DIRECTION	1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30
CARD 8	KTRANS <input type="text"/> (49X, 13) INDICATOR FOR TRANSITION MODEL	1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30
CARD 9	LAMTRB <input type="text"/> (49X, 13) INDICATES WHETHER FLOW IS LAMINAR OR TURBULENT	1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30

PROGRAM	PACKAGING INSTRUCTIONS	CLASSING	PAGE	OF
ICBL3D				
PROGRAMMER	DATE	REACH	CARD FILE NUMBER*	

STATEMENT NUMBER	FORTRAN STATEMENT	IDENTIFICATION REFERENCE
CARD 10	LPRT <input type="text"/> (49X, 13)	PRINT CONTROL PARAMETER IN THE STREAMWISE DIRECTION
CARD 11	NIT1 <input type="text"/> (49X, 13)	ITERATION COUNTER USED TO ADJUST STREAMWISE STEP SIZE
CARD 12	NIT2 <input type="text"/> (49X, 13)	ITERATION COUNTER USED TO ADJUST STREAMWISE STEP SIZE

IBM

FORTRAN Coding Form

GX28-7327-6 U/M 050-
Printed in U.S.A.

PROGRAM ICBL3D	DATE	PUNCHING INSTRUCTIONS	GRAPHIC PUNCH	PAGE OR CARD ELECTED NAME
-------------------	------	--------------------------	------------------	---------------------------------

FORTRAN STATEMENT		SEQUENCE
1	2	3
CARD 16	NSOLVE	
	NUMBER OF VARIABLES IN THE XSTA ARRAY	
	(49 X, 13)	
CARD 17	ADTEST	
	PROVIDES CONVERGENCE CRITERIA FOR STREAMWISE VELOCITY PROFILE	
	(49 X, E14. 6)	
CARD 18	AKSTAR	
	VAN DRIEST INNER EDDY VISCOSITY LAW CONSTANT (REC. VALUE .435)	
	(4 9X, E14. 6)	

IBM

FORTRAN Coding Form

GX28-7327-6 U/M 050**
Printed in U.S.A.

ICBL3D		DATE:		PUNCHING INSTRUCTIONS		GRAPHIC PUNCH		PAGE OF		EMPLOYEE NUMBER	
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FORTRAN STATEMENT		IDENTIFICATION
LINE	STATEMENT	NUMBER
CARD 25	EDYLA W	
	(49X, A3)	
CARD 26	EDYLA W INNER EDDY VISCOSITY LAW TO BE USED IN TURBULENT CASES	
	ETAFAC	
	(49X, E14.6)	
CARD 27	ETAFAC CONTROLS NORMAL GRID SPACING	
	ETAINF	
	MAXIMUM VALUE OF η .	
	(49X, E14.6)	

IBM

FORTRAN Coding Form

GX28-7327-6 U/M 050**
Printed in U.S.A.

PROGRAM		ICBL3D		PAGE		OF	
REMARKS		DATE		PAGE		OF	
PRELIMINARY INSTRUCTIONS		GRAPHIC FUNCTION		PAGE		OF	
REMARKS		DATE		PAGE		OF	

LINE	STATEMENT	REMARKS	DATE	PAGE	OF
1					
2					
3					
4					
5					
6					
7					
8					
9					
10					
11					
12					
13					
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CARD 28

PR

(49X, E14.6)

PRANDTL NUMBER OF FLUID

PR

CARD 29

RTW

(49X, E14.6)

RATIO WALL TEMPERATURE TO STAGNATION TEMPERATURE

RTW

CARD 30

CP

(49X, E14.6)

SPECIFIC HEAT OF FLUID

CP

PROPOSED

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01101001

5451

646

W	STATEMENT	IP
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Conclusion

Conclusion

CARD	31
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AMUINF

(49X.E14.6)

AMUINF COEFFICIENT OF VISCOSITY

CARD 32

PST AG

(49X.E 14.6)

[illegible]

CARD 33

LINE

(49 X. E14.6)

PINF	FREE STREAM	STATIC	STATIC PRESSURE
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IBM

FORTRAN Coding Form

GX28-7327-6 U/M 050**
Printed in U.S.A.

PROJECT		ICBL3D		DATE		PAGE		OF		CARD NO		NUMBER	
PROGRAMMING INSTRUCTIONS		DYNAMIC		PUNCH		PAGE		OF		CARD NO		NUMBER	

LINE	STATEMENT	IDENTIFICATION	SEQUENCE
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CARD 34

XBAR

(49X, E14.6)

RELATIVE LENGTH OF THE TRANSITION REGIME IN TURBULENT CASES

CARD 35

UFS

(49X, E14.6)

FREE-STREAM VELOCITY

CARD 36 TO CARD 36+NSOLVE

XSTA(1) 1=1, NSOLVE

(F12.6)

ARRAY OF SURFACE DISTANCES WHERE SOLUTIONS DESIRED

Number of cards per page 100

PROGRAM	ICBL3D	PROGRAMMING INSTRUCTIONS	GRAPHIC FUNCTION	PAGE	OF
PROGRAM		(DATE)			

STATEMENT NUMBER						FORTRAN STATEMENT																				IDENTIFICATION SEQUENCE									
7	6	5	4	3	2	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30
CARD 36+NSOLVE+1 TO LAST CARD																																			
XTW(I)						TWX(I)						XC1(I)						CIX(I)																	
(4E12.6)																																			
XTW(I)						ARRAY OF SURFACE DISTANCES CORRESPONDING TO WALL TEMPERATURES																													
TWX(I)						WALL TEMPERATURE AT XTW(I)																													
XC1(I)						ARRAY OF SURFACE DISTANCES CORRESPONDING TO INJECTION RATES																													
CIX(I)						INJECTION RATE AT XC1(I)																													

APPENDIX X

Program ICBL3D Listing

[illegible][illegible]

DATE 05/12/77

```

C SUBROUTINE ABCDE CALCULATES THE COEFFICIENTS OF THE GOVERNING
C EQUATIONS IN FINITE-DIFFERENCE FORM
C
C I=1
C K=2
C
C IF (KK.EQ.1.AND.LL.EQ.1) GO TO 60
C IF (KK.EQ.1) GO TO 50
C IF (LL.EQ.1) GO TO 50
C
C COEFFICIENTS FOR THE GENERAL CASE
C
DD 10 J=2,IM
DETA1=ETA(J+1)-ETA(J)+ETAFAC**2*(ETA(J)-ETA(J-1))
DETA2=ETA(J+1)-ETA(J)**2+ETAFAC*(ETA(J)-ETA(J-1))**2
A(J)=CRI*(2.000*ETAFAC*A0(J)/DETA2-ETAFAC**2*A1(J)/DETA1)
CC(J)=CRI*(2.000*A0(J)/DETA2+A1(J)/DETA1)
B5(J)=CRI*(-2.000*(1.000+ETAFAC)*A0(J)/DETA2-(1.000-ETAFAC**2)*A1(J)/DETA1+A2(J))
B9(J)=DETA1+A2(J)
B(J)=B8(J)+A5(J)/DX1+A5(J)/2.000/DW
DD(J)=-11.000-CRI/CCI*TA(J)*W(I,J-1,K)*BB(J)*W(I,J,K)+CC(J)*W(I,J,K-1)+A5(J)/DX1+A5(J)/2.000/DW
L+1,K))
D(J)=D(J)-A3(J)+A5(J)*W(I,J,K)/DX1+A5(J)*W(I+1,J,K-1)-W(I,J,K+1)
L*(I,J,K))/2.000/DW
CONTINUE
RETURN
CONTINUE
C
C COEFFICIENTS FOR THE STAGNATION LINE
C
I=2
K=1
DD 30 J=2,IM
DETA1=ETA(J+1)-ETA(J)+ETAFAC**2*(ETA(J)-ETA(J-1))
DETA2=ETA(J+1)-ETA(J)**2+ETAFAC*(ETA(J)-ETA(J-1))**2
A(J)=CRI*(2.000*ETAFAC*A0(J)/DETA2-ETAFAC**2*A1(J)/DETA1)
CC(J)=CRI*(2.000*A0(J)/DETA2+A1(J)/DETA1)
B5(J)=CRI*(-2.000*(1.000+ETAFAC)*A0(J)/DETA2-(1.000-ETAFAC**2)*A1(J)/DETA1+A2(J))
B9(J)=DETA1+A2(J)
B(J)=B8(J)+A5(J)/DX1+A5(J)/2.000/DW
DD(J)=-11.000-CRI/CCI*TA(J)*W(I,J-1,K)*BB(J)*W(I,J,K)+CC(J)*W(I,J,K-1)+A5(J)/DX1+A5(J)/2.000/DW
L+1,K))
D(J)=D(J)-A3(J)+A5(J)*W(I,J,K)/DX1+A5(J)*W(I+1,J,K-1)-W(I,J,K+1)
L*(I,J,K))/2.000/DW
CONTINUE
RETURN
CONTINUE
C
C COEFFICIENTS FOR THE WINDWARD STREAMLINE
C
I=1

```

[illegible]

3

DATE 05/12/77

```

C
10  IE2=(IE-1)/2
    IF (IST.EQ.0.00) GO TO 20
    IF (IST.EQ.2.00) GO TO 10
    ETAIN2=L.100*ETAINF
    WRITE (6,200) X,ETAINF,ETAIN2,IST
    GO TO 20
    ETAIN2=.900*ETAINF
    WRITE (6,230) X,ETAINF,ETAIN2,IST
    GO TO 20
    ETAIN2=ETAINF
    WRITE (6,300) X,ETAOLD,ETAIN2,IST
    GO TO 40
    CONTINUE
    IF (ETAFAC.EQ.1.000) DETAI=ETAIN2/DELOAT(IM)
    IF (ETAFAC.NE.1.000) DETAI=ETAIN2*(ETAFAC-1.000)/(ETAFAC**IM-1.000)
1)  DETAI(1)=0.000
    DETAI(2)=DETAI
    ETA(1)=J.000
    ETAI(1)=0.000
    ETAI(2)=DETAI(2)
    DO 30 N=2,IM
    DETAI(N+1)=DETAI(N)*ETAFAC
    ETAI(N+1)=ETAI(N)
    CONTINUE
    ETAI(IE)=ETAIN2
    ETAI(IE)=ETAINF
    GO TO 60
    CONTINUE
    IF (ETAFAC.EQ.1.000) DETAI=ETAOLD/DELOAT(IM)
    IF (ETAFAC.NE.1.000) DETAI=ETAOLD*(ETAFAC-1.000)/(ETAFAC**IM-1.000)
1)  DETAI(1)=0.000
    DETAI(2)=DETAI
    ETAI(1)=0.000
    ETAI(2)=DETAI(2)
    DO 50 N=2,IM
    DETAI(N+1)=DETAI(N)*ETAFAC
    ETAI(N+1)=ETAI(N)*DETAI(N+1)
    CONTINUE
    ETAI(IE)=ETAOLD
    ETAI(IE)=ETAINF
    GO TO 150
    IF (ETAI(1).GE.ETAOLD) GO TO 150
    J=J+1
    IF (ETAI(N).GT.ETA(J)) GO TO 70

```

[illegible]

[illegible]

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```

200 IF (J,1)=I2A(J) GO TO 200
    IF (G,2)=I2E2(J) GO TO 210
    IF (F,1)=EQ,3,000) GO TO 210
    IF (F,1)=F2C(J) GO TO 210
    IF (J,3)=F2C(J) GO TO 210
    IF (G,1)=I2E2(J),EQ,3,000) GO TO 210
    IF (G,1)=G2C(J) GO TO 210
    IF (F,1)=OT,2,000) GO TO 220
    IF (J,2)=F2D(J) GO TO 220
    IF (G,2)=I2E2(J),EQ,0,000) GO TO 220
    IF (G,2)=G2D(J) GO TO 220
    CONTINUE
220 IF (J,1)=I2A(J) GO TO 230
    IF (F,1)=I2E2(J),EQ,3,000) GO TO 230
    IF (G,2)=I2E2(J),EQ,1,1FN)
        CALL DERIV (FM,ETA2,IE,1,1FN)
    IF (G,1)=I2E2(J),NE,3,000) CALL DERIV (GM,ETA2,IE,1,1GN)
    IF (F,1)=EQ,3,000) GO TO 240
    IF (F,1)=F2C(J) GO TO 240
    CALL DERIV3 (F,1,2,ETA2,IE,1,1FN)
    CALL DERIV3 (F,1,2,ETA2,IE,1,1FN)
    IF (G,1)=I2E2(J),NE,0,000) CALL DERIV3 (G,1,2,ETA2,IE,1,1GN)
    IF (G,1)=I2E2(J),NE,0,000) GO TO 250
    CALL DERIV3 (F,2,1,ETA2,IE,1,1FN)
    CALL DERIV3 (F,2,1,ETA2,IE,1,1FN)
    IF (G,2)=I2E2(J),NE,0,000) CALL DERIV3 (G,2,1,ETA2,IE,1,1GN)
    IF (F,1)=EQ,3,000) GO TO 260
    IF (F,1)=I2E2(J),EQ,1,1FN)
        CALL DERIV3 (F,1,3,ETA2,IE,1,1FN)
    CALL DERIV3 (F,1,3,ETA2,IE,1,1FN)
    IF (G,1)=I2E2(J),NE,0,000) CALL DERIV3 (G,1,3,ETA2,IE,1,1GN)
    IF (F,1)=OT,2,000) GO TO 270
    CALL DERIV3 (F,2,2,ETA2,IE,1,1FN)
    CALL DERIV3 (F,2,2,ETA2,IE,1,1FN)
    IF (G,2)=I2E2(J),NE,0,000) CALL DERIV3 (G,2,2,ETA2,IE,1,1GN)
    CONTINUE
    RETURN
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SUBROUTINE AERO
  IMPLICIT REAL*8 (A-H,O-Z)
  COMMON /REAL*8 / RHOINF,PINF,TF,UF,AMUINF,CP,KEINF,PR
  COMMON /PREF / PREF,PREF,AMUREF
  COMMON /PSTAG / PSTAG,PSTAG,PNC,OMSTAG
  DATA ROIN/4HRHOI/,PIN/4HPINF/

  RHOINF IS IN SLUGS
  PINF IS IN PSIA
  PSTAG IS IN PSIA
  PREF IS IN PSIA
  CP IN FT**2/SEC**2/DEG.R
  R IN FT**2/SEC**2/DEG.R
  UFS IN FT/SEC
  TWALL,TREF,TSTAG IN DEG.R
  AMUREF IS IN (LB-SEC)/FT**2

  AMUREF=AMUINF
  RHOINF=2.00*(PSTAG-PINF)/UFS**2
  REINF=RHOINF*UF/AMUINF
  TSTAG=TF+.50J/CP*UFS**2
  RETURN
END

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SUBROUTINE BLUNT2 (ISNT)
  IMPLICIT REAL*8 (A-H,O-Z)
  REAL*8 NOISE
  COMMON /EDGE / UEDG,TEEG,VEDG,DUEGDX,DUEGDW,D2UEGDW,DVEGDW,DVEGDW,RE
  10G,DRDGDW,ZROD,XROG,XRO,XRO,DRDX,DRDX,D2RODX,D2RODX,D3RODX,D3RODX,D3RODX
  100,DRDX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX
  200,DRDX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX,D3RODX
  COMMON /FRSTRM / RHOINF,PINF,TF,UF,AMUINF,CP,KEINF,PR
  COMMON /GEOM / ALPHA,NOISE,KNOSE,WLST,X,XX,WX
  COMMON /INTEGR / IE,IM,KEND,KEND2,KLX,K,L,NBLNTI,IND,KPRT,LPRT,KPR,
  1LPR
  COMMON /VOLD / VOLD(61),CVOLD(61),RULD(61)
  COMMON /STGEOM / DVEGXW,DVEGXW,DVEGXW,DVEGXW,DVEGXW,DVEGXW,DVEGXW,DVEGXW
  1VWXXW,D2VWXXW,D3VWXXW,D4VWXXW,D5VWXXW,D6VWXXW,D7VWXXW,D8VWXXW,D9VWXXW
  COMMON /OLDEEG / R3,UE3,DVE3D
  COMMON /MSOLVE / DM,W
  COMMON /TVCURV / TVC
  DIMENSION A(40,2),B(40,2),C(40,2),D(40,2),E(40,2),F(40,2),G(40,2),
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5, HXSUM(2), HXSUM(2), HXNNSM(2), RNNNSM(2), BNNNSM(2), KOSUM(2), BLUN
7,XSUSUM(2), XONNSM(2), XOXSUM(2), XAXNSM(2), GX(40,2), RXNNS(2), BLUN
9XXS(2), UNNSUM(2), VANSUM(2), HWNSM(2), HXANSW(2) BLUN
10DIMENSION ARADS(40), AUS(40), AVS(40), AXSI(40), AHWS(40) BLUN
11IX=ARADS(40), AUX(40), AVX(40), ARXI(40), ARXS(40), AHWS(40) BLUN
12XS(40), XSU(40), AXX(40), AXXI(40), AXXS(40), BLUN
DATA YES/YES/ BLUN
C
PI=DARCOS(-1.D0)
IF (K-NE-1) GO TO 50
IF (L-EQ-1) IRD=0
M=2
*****
XS = BODY FIXED SURFACE DISTANCE *****
*****
AXIAL DISTANCE AND FOURIER COEFFICIENTS ARE READ FROM UNIT 10
CONTINUE
IRD=IRD+1 XS=ARADS,AUS,AVS,AHS,AHWS,AXSRFS,AUX,AVX,ARX,ARXS,AHXXS
READ (10) XO,AXOX
1AHWS,AXA,AXOX
IF (X-GT-XS-OM-IRD.LE-1) GO TO 20
BACKSPACE 10
BACKSPACE 10
IRD=IRD-2
GO TO 10
CONTINUE
IRD=IRD+1 XS=ARADS,AUS,AVS,AHS,AHWS,AXSRFS,AUX,AVX,ARX,ARXS,AHXXS
READ (10) XO,AXOX
1AHWS,AXA,AXOX
DO 30 J=1,KLSX
A(J,2)=ADSI(J)
B(J,2)=ASIS(J)
C(J,2)=AMSIS(J)
E(J,2)=ASRFS(J)
F(J,2)=AO(J)
G(X(J,2))=VXX(J)
GX(X(J,2))=VXX(J)
AX(X(J,2))=ARXS(J)
EX(X(J,2))=HWXS(J)
CX(X(J,2))=AXOX(J)
CONTINUE=XS
BACKSPACE 10

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DATE 05/12/77

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BACKSPACE 10
TRD=TRD-1
READ (10) XS,ARADS,AUS,AVS,AHXS,AHWS,AXSRFS,AUX,AVX,ARX,AHXS
1 AHXS,AXO,AXOX
DO 40 J=1,KLX
  AL(J)=AXRDS(J)
  BL(J)=AXR(J)
  CL(J)=AXR(J)
  DL(J)=AXR(J)
  EL(J)=AXR(J)
  FL(J)=AXR(J)
  GL(J)=AXR(J)
  HL(J)=AXR(J)
  IL(J)=AXR(J)
  JL(J)=AXR(J)
  KL(J)=AXR(J)
  LL(J)=AXR(J)
  ML(J)=AXR(J)
  NL(J)=AXR(J)
  OL(J)=AXR(J)
  PL(J)=AXR(J)
  QL(J)=AXR(J)
  RL(J)=AXR(J)
  SL(J)=AXR(J)
  TL(J)=AXR(J)
  UL(J)=AXR(J)
  VL(J)=AXR(J)
  WL(J)=AXR(J)
  XL(J)=AXR(J)
  YL(J)=AXR(J)
  ZL(J)=AXR(J)
  CONTINUE
  AXSL(J)=XS
  AXSL(J)=XS
  CONTINUE
  MM=M-1
  PHI=2.0*PI*W
  KKL=KLX-1
  FOURIER SERIES ARE USED TO COMPUTE THE EDGE PROPERTIES
DO 80 I=1,M
  USUM(I)=0.000
  RADSUM(I)=0.000
  VSUM(I)=0.000
  HXSUM(I)=0.000
  HWSUM(I)=0.000
  XFSUM(I)=0.000
  UXSUM(I)=0.000
  VXSUM(I)=0.000
  VNSUM(I)=0.000
  RXSUM(I)=0.000
  RXXSUM(I)=0.000
  UNSUM(I)=0.000
  RNSUM(I)=0.000
  RNSUM(I)=0.000
  RNSUM(I)=0.000
  HXNSUM(I)=0.000
  HXNSUM(I)=0.000
  HXNSUM(I)=0.000
  RXNSUM(I)=0.000
  RXNSUM(I)=0.000
  BLUN 750
  BLUN 760
  BLUN 770
  BLUN 780
  BLUN 790
  BLUN 800
  BLUN 810
  BLUN 820
  BLUN 830
  BLUN 840
  BLUN 850
  BLUN 860
  BLUN 870
  BLUN 880
  BLUN 890
  BLUN 900
  BLUN 910
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  BLUN 930
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  BLUN 970
  BLUN 980
  BLUN 990
  BLUN 1000
  BLUN 1010
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  BLUN 1080
  BLUN 1090
  BLUN 1100
  BLUN 1110
  BLUN 1120
  BLUN 1130
  BLUN 1140
  BLUN 1150
  BLUN 1160
  BLUN 1170
  BLUN 1180
  BLUN 1190
  BLUN 1200
  BLUN 1210
  BLUN 1220
  BLUN 1230
  BLUN 1240
  BLUN 1250
```


[illegible]

[illegible]

```

120 D3RDX=HXXSUM(MH)+FAC*(HXXSUM(MH))-HXXSUM(MH))
121 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))//2.00*PI
122 D3RDX=(XRXNSUM(MH)+FAC*(XRXNSUM(MH))-XRXNSUM(MH))//2.00*PI
123 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))//4.00*PI**2
124 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))
125 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))//2.00*PI
126 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))//2.00*PI**3
127 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))//8.00*PI**3
128 D3RDX=(RXXSUM(MH)+FAC*(RXXSUM(MH))-RXXSUM(MH))//4.00*PI**2
129 IF (I*NT.EQ.2) GO TO 130
130
131 CALCULATE THE X DERIVATIVES
132
133 D3RXXX=(RXXSUM(MH)-RXXSUM(MH))/XXSUM(MH))//XXSUM(MH))
134 D3RXXX=(XRXNSUM(MH)-XRXNSUM(MH))/XXSUM(MH))-XXSUM(MH))
135 D3RXXX=(XRXNSUM(MH)-XRXNSUM(MH))/XXSUM(MH))-XXSUM(MH))
136 IF (L*NE.1) GO TO 137
137 DVEGXX=(VXSUM(MH)-VXSUM(MH))/XXSUM(MH))//XXSUM(MH))
138 DVEGXX=(VXSUM(MH)-VXSUM(MH))/XXSUM(MH))//XXSUM(MH))
139 DVEGXX=(VXSUM(MH)-VXSUM(MH))/XXSUM(MH))//XXSUM(MH))
140 CONTINUE
141 CALL HOFFN
142 CONTINUE
143
144 IF (I*NT.EQ.0) GO TO 140
145 WRITE (6,150) UEDG,DUEGDX,DUEGDM,VEDG,DVEGDX,DVEGDM
146 WRITE (6,160) REDG,DRODX,D2RODX,DRQDM,D2RODM
147 WRITE (6,170) DRDAM,D3RXXX,D3RXXX,D3RXXX,D3RXXX,D3RXXX
148 WRITE (6,180) HXO,HWO,DHXODX,DHMOXX,DHXODM,DHMOXX
149 WRITE (6,190) XO,DXODX,DXODM
150 CONTINUE
151 RETURN
152
153 C
154
155 157H VEG = ,E13.6,10H DVEGDX = ,E13.6,10H DUEGDM = ,E13.
156 157H VEG = ,E13.6,10H DVEGDX = ,E13.6,10H DVEGDM = ,E13.6)
157 157H VEG = ,E13.6,10H DVEGDX = ,E13.6,10H DRODX = ,E13.6)
158 157H DRODM = ,E13.6,10H D2RODM = ,E13.6)
159 157H DRODM = ,E13.6,10H D2RODX = ,E13.6,10H D3RXXX = ,E13.6)
160 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
161 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
162 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
163 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
164 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
165 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
166 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
167 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
168 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
169 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
170 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
171 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
172 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
173 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
174 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
175 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
176 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
177 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
178 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
179 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
180 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
181 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
182 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
183 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
184 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
185 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
186 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
187 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
188 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
189 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
190 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
191 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
192 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
193 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
194 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
195 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
196 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
197 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
198 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
199 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
200 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
201 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
202 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
203 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
204 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
205 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
206 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
207 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
208 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
209 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
210 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
211 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
212 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
213 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
214 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
215 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
216 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
217 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
218 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
219 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
220 157H DRODM = ,E13.6,10H D3RXXX = ,E13.6,10H D3RXXX = ,E13.6)
221 157H DRODM =
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```

1 IOL(3),IWL(2,IOL,3),Y(IOL),YOL(IOL)
COMMON /GEOM/ ALPHA,NOSE,NOSEWLST,X,XX,XX
COMMON /INJECT/ INJCT,NOINJ,MASTRN
COMMON /INTEGR/ IE,IM,KEND,KENDZ,KLX,K,L,NBLNTL,IND,KPRT,LPRT,KPR,
1 LPR
COMMON /SOLPNT/ VW(IOL),GW(IOL),TW(IOL),GMN(IOL),FVN(IOL),FW(IOL),
1 IWN(IOL),XIM,XM,RW
COMMON /TRANSN/ KTRANS,KONSET,XIF,CHI2(IOL),CHIMAX,XBAR
COMMON /TRBLNT/ ASTAR,AKSTAR,ALAMDA,YSUBL,EVSCI(IOL),EDYLA,EPLUS
1 (IOL),LAMIRB
COMMON /XSOLVE/ XSTAI(100),DXMAX,DX,DXOLD,DX1,NSOLVE
COMMON /SPECPL/ LPRTP
COMMON /SHARP/ SHARP,SHSHARP/
DATA BLUNT,SHARP/5HBLUNT,SHSHARP/

LPRTP=0
IF (INIT.GE.0) GO TO 30
CUT BACK X AND DX AND SET DXOLD=DX
DX=DX/2.0D0
X=X-DX
DXOLD=DX

RESET COUNTERS FOR TRANSITION
IF (X-GE.XSTAIKONSET)) GO TO 90
IF (KONSET.EQ.NSOLVE) GO TO 10
LAMIRB=1
XIF=0.0D0
CONTINUE

RESET COUNTER FOR INJECTION
IF (INJCT.EQ.NSOLVE) GO TO 20
IF (X.GT.XSTAIINJCT)) GO TO 20
MASTRN=0
IF (X-GE.XSTAI(NOINJ)) MASTRN=0
RETURN
CONTINUE

ADJUST DX USING NIT
IF (INIT.GT.NIT1) GO TO 40
DX=2.0D0*DX
IF (DX.GT.DXMAX) DX=DXMAX
GO TO 50
IF (INIT.LT.NIT2) GO TO 50
DX=0.5D0*DX
CONTINUE
IF (X.EQ.XSTAIINJCT)) DX=DX/10.0D0

```

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C      IF (X.EQ.XSTA(KONSET)) DX=DX/10.000
C      OXOLD=DX
C      SET DX TO GIVE A SOLUTION AT XSTA(IXSOLV(III))
C      DO 60 I=1,NSOLVE
C      J=1
C      IF (XSTA(I).GT.X.AND.XSTA(I).LE.(X+1.25DX*DX)) GO TO 70
C      CONTINUE
C      X=X+DX
C      GO TO 80
C      X=XSTA(J)-X
C      X=X+DX
C      LPRIP=1
C      CONTINUE
C      BEGIN THE TRANSITION REGIME IF X=XSTA(KONSET)
C      CALCULATE THE TRANSITION INTERMITTENCY FACTOR FOR
C      X-GE-XSTA(KONSET)
C      IF (KONSET.EQ.NSOLVE) GO TO 100
C      IF (X.LT.XSTA(KONSET)) GO TO 100
C      IF (X.GT.XSTA(KONSET)) GO TO 90
C      LAMTRB=2
C      WRITE (6,130) X,LAMTRB
C      CONTINUE
C      IF (KTRANS.EQ.0) XIF=1.000
C      IF (KTRANS.EQ.1) XIF=1.000-DEXP(-0.41200*2.91700**2*((X-XSTA(KONSET)
C      11) / (XSTA(KONSET) * (XBAR-1.000)))**2)
C      IF (INT.LT.0) GO TO 10
C      BEGIN MASS TRANSFER IF X=XSTA(INJCT)
C      IF (INJCT.EQ.NSOLVE) GO TO 110
C      IF (X.LE.XSTA(INJCT)) GO TO 110
C      IF (MASTRN.EQ.1) GO TO 110
C      MASTRN=1
C      WRITE (6,140) X,MASTRN
C      END MASS TRANSFER IF X=XSTA(NGINJ)
C      IF (NGINJ.EQ.NSOLVE) GO TO 120
C      IF (X.NE.XSTA(NGINJ)) GO TO 120
C      MASTRN=0
C      WRITE (6,150) X,MASTRN
C      RETURN
C      120
C      130
C      FORMAT (1H0,10X,27HBEGIN TRANSITION REGIME, X=E12.6,9H LAMTRB=,1
C      12/)

```

CHAN 570
 CHAN 580
 CHAN 590
 CHAN 600
 CHAN 610
 CHAN 620
 CHAN 630
 CHAN 640
 CHAN 650
 CHAN 660
 CHAN 670
 CHAN 680
 CHAN 690
 CHAN 700
 CHAN 710
 CHAN 720
 CHAN 730
 CHAN 740
 CHAN 750
 CHAN 760
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 CHAN 940
 CHAN 950
 CHAN 960
 CHAN 970
 CHAN 980
 CHAN 990
 CHAN 1000
 CHAN 1010
 CHAN 1020
 CHAN 1030
 CHAN 1040
 CHAN 1050
 CHAN 1060
 CHAN 1070


```

140  FOMAT (I10,I0X,23BEGIN MASS TRANSFER, X=I12.6,9H
150  FOMAT (I10,I0X,21END MASS TRANSFER, X=I12.6,9H
      END
      CAN1000  MASTRN=,12,1
      CAN1090  MASTRN=,12,1
      CAN1100  MASTRN=,12,1

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```

10  C C C
    DIMENSION FOLD(101), GOLD(101), TOLD(101), TS(101,2)
    DIMENSION GNA(2,101),
    DATA BLUNT, SHARP/5HBLUNT, 5HSHARP/
    NITTOT=0
    ASYM=1.000
    ETA2SV=0.000
    BEGIN THE LOOP FOR STEPPING DOWNSTREAM
    CONTINUE
    L=L+1
    IFL=1
    FIND(8,IFL)
    W=0.000
    IF (KEND.EQ.1.OR.ALPHA.LT.1.D-08) GO TO 20
    UM=WLST/OFLAT(KEND-1)
    CONTINUE
    KPR=KPR+1
    IF (L.EQ.1) KLAST=KEND
    BEGIN THE DO-LOOP FOR STEPPING AROUND THE BODY
    DO 400 K=1,KEND
    CALL TIMEON
    LC=3
    IF (L.EQ.1) LC=1
    IF (K.EQ.1.AND.L.GT.1) LC=2
    IF (K.EQ.KEND.AND.L.GT.1) LC=2
    LADEFA=0
    LX=WLST*180.000
    IF (KLAST.GT.0) GO TO 30
    WRITE(6,440)
    RETURN
    IF (K.GT.KLAST) GO TO 370
    CONTINUE
    30  C C C
    40  C C C C C
    OBTAIN EDGE, AND WALL, INVISCID VALUES AND SET
    MIDPOINT VALUES TO VALUES OBTAINED FOR THIS PLANE AT
    THE LAST STREAMWISE STATION
    CALL EGPROP (LC)
    CALL WALL
    DO 50 J=1,IE
    FW(J)=F(1,J,2)
    GW(J)=G(1,J,2)
    GVEL=GV(J)
    IF (K.EQ.1) GVEL=0.000
    TW(J)=T(1,J,2)
    FWN(J)=FN(1,J,2)
    GWN(J)=GN(1,J,2)
    TWN(J)=TN(1,J,2)

```

```

CONT 440
CONT 450
CONT 460
CONT 470
CONT 480
CONT 490
CONT 500
CONT 510
CONT 520
CONT 530
CONT 540
CONT 550
CONT 560
CONT 570
CONT 580
CONT 590
CONT 600
CONT 610
CONT 620
CONT 630
CONT 640
CONT 650
CONT 660
CONT 670
CONT 680
CONT 690
CONT 700
CONT 710
CONT 720
CONT 730
CONT 740
CONT 750
CONT 760
CONT 770
CONT 780
CONT 790
CONT 800
CONT 810
CONT 820
CONT 830
CONT 840
CONT 850
CONT 860
CONT 870
CONT 880
CONT 890
CONT 900
CONT 910
CONT 920
CONT 930
CONT 940

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50 CONTINUE
C IF (L.EQ.1) CALL VCALC (LC)
C
C SAVE THE PROFILES FROM THE LAST ITERATION
C
80 DO 70 J=1,IE
C FOLD(J)=F(2,J,2)
C GOLD(J)=G(2,J,2)
C TOLD(J)=T(2,J,2)
C CONTINUE
C
70 SOLVE THE STREAMWISE MOMENTUM CONSERVATION EQUATION
C
C IF (LAMTRB.EQ.2) CALL EDYVIS
C CALL XMOD (LC)
C CALL ABCDE (F)
C CALL SOLVE (F,FN,0.000,0.000,1.000)
C IF (L.GT.1) GO TO 90
C DO 80 J=1,IE
C F(1,J,3)=F(2,J,2)*2.000-F(2,J,1)
C FN(1,J,3)=FN(2,J,2)*2.000-FN(2,J,1)
C F(1,J,2)=F(2,J,2)
C FN(1,J,2)=FN(2,J,2)
C CONTINUE
C CONTINUE
C IF (K.GT.1) GO TO 110
C DO 100 J=1,IE
C F(2,J,1)=F(2,J,2)
C F(1,J,3)=F(1,J,2)
C FN(2,J,1)=FN(2,J,2)
C FN(1,J,3)=FN(1,J,2)
C CONTINUE
C CONTINUE
C DO 120 J=1,IE
C IF (K.EQ.KLAST) F(1,J,3)=F(2,J,2)-F(2,J,1)+F(1,J,2)
C IF (K.EQ.KLAST) FN(1,J,3)=FN(2,J,2)-FN(2,J,1)+FN(1,J,2)
C F(1,J,2)=F(2,J,2)*CRI+FN(1,J,2)*1.000-CRI
C FN(1,J,2)=FN(2,J,2)*CRI+FN(1,J,2)*1.000-CRI
C IF (L.EQ.1) F(1,J,1)=F(2,J,1)*1.000-CRI+FN(2,J,2)*CRI
C IF (L.EQ.1) FN(1,J,1)=FN(2,J,1)*1.000-CRI+FN(2,J,2)*CRI
C CONTINUE
C CONTINUE
C IF (ALPHA.LT.1.D-03) GO TO 180
C
C SOLVE THE CROSSFLOW MOMENTUM CONSERVATION EQUATION
C
C CALL VCALC (LC)
C IF (LAMTRB.EQ.2) CALL EDYVIS
C CALL PHIMOM (LC)
C CALL ABCDE (G)
C CALL SOLVE (G,GN,0.000,0.000,AL3)
C IF (L.GT.1) GO TO 140

```

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CONT 950
CONT 960
CONT 970
CONT 980
CONT 990
CONT 1000
CONT 1010
CONT 1020
CONT 1030
CONT 1040
CONT 1050
CONT 1060
CONT 1070
CONT 1080
CONT 1090
CONT 1100
CONT 1110
CONT 1120
CONT 1130
CONT 1140
CONT 1150
CONT 1160
CONT 1170
CONT 1180
CONT 1190
CONT 1200
CONT 1210
CONT 1220
CONT 1230
CONT 1240
CONT 1250
CONT 1260
CONT 1270
CONT 1280
CONT 1290
CONT 1300
CONT 1310
CONT 1320
CONT 1330
CONT 1340
CONT 1350
CONT 1360
CONT 1370
CONT 1380
CONT 1390
CONT 1400
CONT 1410
CONT 1420
CONT 1430
CONT 1440
CONT 1450

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130 DO 130 J=1,IE
140   G(1,J,3)=G(2,J,2)*2.000-G(2,J,1)
      GN(1,2,3)=GN(2,J,2)*2.000-GN(2,J,1)
      G(1,J,2)=G(2,J,2)
      GN(1,2,2)=GN(2,J,2)
      CONTINUE
      IF (K.GT.1) GO TO 160
      DO 150 J=1,IE
      G(2,J,1)=G(2,J,2)
      G(1,J,3)=G(1,J,2)
      GN(2,J,1)=GN(2,J,2)
      GN(1,J,3)=GN(1,J,2)
      CONTINUE
      DO 170 J=1,IE
      IF (K.EQ.KLAST) G(1,J,3)=G(2,J,2)-G(2,J,1)+G(1,J,2)
      IF (K.EQ.KLAST) GN(1,J,3)=GN(2,J,2)-GN(2,J,1)+GN(1,J,2)
      GM(J)=G(2,J,2)*CRI+1.000-CRI*G(1,J,2)
      GMN(J)=GN(2,J,2)*CRI+1.000-CRI*GN(1,J,2)
      IF (L.EQ.1) GM(J)=G(2,J,1)+1.000-CRI*G(2,J,2)*CRI
      IF (L.EQ.1) GMN(J)=GN(2,J,1)+1.000-CRI*GN(2,J,2)*CRI
      CONTINUE
      CALL VCALC (LC)
      NIT=NIT+1
      THE SOLUTION IS CHECKED FOR CONVERGENCE
      IF (NIT.LE.NIT3) GO TO 200
      WRITE (6,420) K,L,NIT
      IF (K.GT.1) GO TO 190
      NITOT=NIT*NITOT
      IF (NITOT.GT.(3*NIT3)) WRITE (6,430) K,L,NITOT
      IF (NITOT.GT.(3*NIT3)) STOP
      NIT=-1
      CALL CHANGX
      NIT=0
      GO TO 40
      CONTINUE
      KLAST=K-1
      GO TO 370
      CONTINUE
      CONVERGENCE TEST ON ALL POINTS OF THE F AND G ARRAYS
      DIF=0.000
      DO 210 J=2,IE
      DIF=ABS(F(2,J,2)-FOLD(J))/DABS(FOLD(J))
      IF (DIF.GT.DIF) DIF=DIF
      IF (GOLD(J).EQ.0.000) GO TO 210

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210 DIFF=DABS(G(2,J,2)-GOLD(J))/DABS(GOLD(J))
    CONTINUE
    IF (DIFF.GT.DIF) DIF=DIFF
    IF (DIF.GT.CONV) GO TO 60
    IF (INIT.EQ.1) GO TO 60
C
C
C SOLVE THE ENERGY CONSERVATION EQUATION
    IF (LAMTRB.EQ.2) CALL EDYVIS
    IS=1
    CALL ENERGY (LC)
    CALL ABOVE (1)
    CALL SOLVE (1,TN,0.000,1B,1.000)
    IF (L.GT.1) GO TO 230
    DO 220 J=1,IE
    T(1,J,3)=T(2,J,2)*2.000-T(2,J,1)
    T(1,J,3)=TN(2,J,2)*2.000-TN(2,J,1)
    T(1,J,2)=T(2,J,2)
    T(1,J,2)=TN(2,J,2)
    CONTINUE
    IF (K.GT.1) GO TO 250
    DO 240 J=1,IE
    T(2,J,1)=T(2,J,2)
    T(1,J,3)=T(1,J,2)
    T(2,J,1)=T(2,J,2)
    T(1,J,3)=T(1,J,2)
    CONTINUE
    CONTINUE
    DO 260 J=1,IE
    IF (K.EQ.KLAST) T(1,J,3)=T(2,J,2)-T(2,J,1)+T(1,J,2)
    IF (K.EQ.KLAST) TN(1,J,3)=TN(2,J,1)+TN(1,J,2)
    T(1,J)=T(2,J,2)*CRI+T(1,J,2)*(1.000-CRI)
    TN(1,J)=TN(2,J,2)*CRI+TN(1,J,2)*(1.000-CRI)
    IF (L.EQ.1) T(1,J)=T(2,J,1)*(1.000-CRI)+T(2,J,2)*CRI
    IF (L.EQ.1) TN(1,J)=TN(2,J,1)*(1.000-CRI)+TN(2,J,2)*CRI
    CONTINUE
    DO 270 J=1,IE
    GVEL=GV(J)
    IF (K.EQ.1) GVEL=0.000
    CONTINUE
260
270
C
C
C TEST THE ASYMPTOTIC NATURE OF THE SOLUTION AND ADJUST ETAINF
    IF (NECESSARY) GO TO 300
    IF (L.EQ.1) GO TO 300
    IF (KADENA.EQ.0) GO TO 300
    ASYM=DABS(F(2,IE,2)-F(2,IE-4,2))
    IF (ASYM.LT.ADTST) GO TO 280
    IT=1.000
    ET#OLD=ETAINF
    GO TO 290

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280 IF (KEND.GT.1) GO TO 300
    IF (LASYM.GT.1) ADTEST=13.0001 GO TO 303
    TST=2.000
    ETALD=ETAINF
290 CONTINUE
    IF (LADETA.NE.0) GO TO 300
    LADETA=1
    CALL ADDETA (TST,ASYM,ETALD)
    GO TO 60
300 CONTINUE
    IF (K.EQ.1) OR (K.EQ.(KEND-1)/2) NITHAF=NIT
    WRITE (8) (F1(J,2),F2(J,2),G(2,J,2),TN(2,J,2),FN(2,J,2),GN(2,J,2),
    1,J,2),RXS(2,J,2),J=1,IE),ETAINF
    CALCULATE AND WRITE OUT THE RESULTS OF THE CURRENT SOLUTION
    CALL PROPT
    CALL OUT2
    M=J+DM
    THE PROGRAM IS SET TO FIND THE SOLUTION AT THE NEXT
    COORDINATE POINT
    DO 320 J=1,IE
    1 IF (K.EQ.1) AND (KEND.GT.1) G(2,J,2)=0.000
    IF (K.EQ.1) AND (KEND.GT.1) GN(2,J,2)=0.000
    F(2,J,1)=F(2,J,2)
    FN(2,J,1)=FN(2,J,2)
    F(1,J,2)=F(1,J,3)
    FN(1,J,2)=FN(1,J,3)
    TN(2,J,1)=TN(2,J,2)
    TN(2,J,1)=TN(2,J,3)
    TN(1,J,2)=TN(1,J,3)
    G(2,J,1)=G(2,J,2)
    GN(2,J,1)=GN(2,J,2)
    G(1,J,2)=G(1,J,3)
    GN(1,J,2)=GN(1,J,3)
    RXS(2,J,1)=RXS(2,J,2)
    RXS(1,J,2)=RXS(1,J,3)
    IF (L.NE.1) GO TO 310
    F(1,J,2)=F(2,J,2)
    FN(1,J,2)=FN(2,J,2)
    TN(1,J,2)=TN(2,J,2)
    GN(1,J,2)=GN(2,J,2)
    F(1,J,3)=F(2,J,2)
    FN(1,J,3)=FN(2,J,2)
    TN(1,J,3)=TN(2,J,2)
    GN(1,J,3)=GN(2,J,2)

```

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```

SUBROUTINE DERIV3 (FX,I, KK,X, IMAX, IMIN, FPX)
  DIMENSION X(101), FX(2,101,3), FP(101), F(101), FPX(2,101,3)
  COMMON /INTEGR/ IE,IM,KEND2,KLX,KOU,L,NBLNT1,IND,KPRT,LPT,K
  I=1; J=1; IE=1; KK=1; IMAX=1; IMIN=1;
  DO 10 J=1,IE
    F(J)=FX(I,J,KK)
    CONTINUE
    DO 20 J=1,IMIN,IMAX
      K=J
      IF (K-IT*(IMIN+1)) K=IMIN+1
      IF (K-GT*(IMAX-1)) K=IMAX-1
      CALL FD3 (X(I,J),X(K-1),X(K),X(K+1),F(K),F(K+1),FP(J))
      CONTINUE
    DO 30 J=1,IE
      FPX(I,J,KK)=FP(J)
      CONTINUE
    END
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  990
  1000

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```
COMMON /PHI/ DAL3DX,DAL3DW,ALB6
COMMON /MSOLVE/ DWTM
DIMENSION ALSOLD(61)
DATA SHARP,BLUNT/SHSHARP,SHBLUNT/

SUBROUTINE IECOEFCALCULATE GROUPS OF EDGE QUANTITIES USED IN
THE COEFFICIENTS OF THE GOVERNING PARTIAL DIFFERENTIAL EQUATIONS

PI=DARCOS(-1.000)
IF (LC.EQ.1) GO TO 10
B1=XI*DUENDX/HX(1)/UEW
B2=XI*DUENDW/HW(1)/UEW
XLAMB=UEW**2/UEW
ETAX=DUENDX/DSQRT(2.00*UEW*X1)-DSQRT(UEW/2.00/X1**3)
ETAW=DUENDW/DSQRT(2.00*UEW*X1)
IF (LC.EQ.2) GO TO 30
B3=XI*DUENDX/HX(1)/UEW
B4=XI*DUENDW/HW(1)/UEW
B5=B1
B6=B2
AL1=VEW/UEW
AL2=1.00
AL3S=AL3
IF (K.EQ.2) AL3S=0.00
AL3=AL1
DAL3DX=(AL3-AL3OLD(K))/DXI
DAL3DW=(AL3-AL3OLD(K+1))-AL3OLD(K))/2.00/DW
AL3OLD(K)=AL3
GO TO 40
CONTINUE
AL1=DABS(DVEWDX)/DVEWDX
B1=1.00
B2=2.00*AL1
B3=1.00
B4=0.00
IF (DABS(DVEWDX).LT.1.0-05) B3=2.00
IF (K.EQ.1) GO TO 20
IF (DABS(DVEWDX).GE.1.0-05) B4=DVEWDX/HW(1)/DVEWDX
IF (DABS(DVEWDX).LE.1.0-05) B4=DVEWDX/HW(1)/DVEWDX
CONTINUE
B5=B3
B6=B4
AL2=AL1
AL3=1.00
AL3OLD(K)=0.00
XLAMB=0.00
GO TO 40
CONTINUE
B3=XI*DUENDX/DVEWDX
B4=0.00
B5=B3
```

10

20

30


```

B6=B4+
ALB6=X*DVEND*H*W(1)/UEW
ALL1=0.00
AL2=0.00
AL3=0.00
AL4=1.00
CONTINUE
IMT=0
IF (IMT.EQ.0) GO TO 50
WRITE (6,60) B1,B2,B3,B4,B5,B6
WHITE (6,70) ALL,AL2,AL3,XLAMB,DAL3DX
CONTINUE
RETURN
C
FORMAT (1H0,7HBL-6 = ,6E13.6)
FORMAT (1H0,8HALL-3 = ,3E13.6,9H XLAMB = ,F13.6,10H DAL3DX = ,F13.6)
END

```

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```

60 EPSIN(N)=RHOINF*AKSTAR**2*Y(N)**2*QAMP(N)*SCALAR(N)/AMUINF
  CONTINUE
  GO TO 110
C
70 CALCULATE THE INNER EDDY VISCOSITY, REICHARDT EQ.
  CONTINUE
  UPLLSF=1.000
  UPLUS=0.000
  UPL=0.000
  EPSIN(1)=0.000
  YPLUS(1)=0.000
  DO 90 N=2,IE
    VPLUS=C*WALL*JFS/DSQRT(TAU(1)/RHOINF)
    YPLUSA=3.6500/(VPLUS+0.34400)
    YPLUS(N)=Y(N)*DSQRT(TAU(1)/RHOINF)/AMUINF
    EPSIN(N)=AMUINF*0.400*(YPLUS(N)-YPLUSA*OTANH(YPLUS(N)/YPLUSA))
    IF (VPLUS.EQ.0.000) GO TO 90
  CONTINUE
  FACTR=DSQRT(1.000*VPLUS*UPLUS)
  EPSITR=EPSIN(N)*FACTR
  UPL2=UPLUS
  UPLUS=UPL+(1.000*UPLUS*VPLUS)/(1.000+EPSITR)+UPLLSF*(YPLUS(N)-Y
1PLUS(N-1))/2.000
  IF (UPL2.EQ.0.000) GO TO 80
  IF (DABS((UPLUS-UPL2)/UPL2)-GT.0.0100) GO TO 80
  EPSIN(N)=EPSIN(N)*DSQRT(1.000*UPLUS*VPLUS)
  UPLLSF=(1.000*UPLUS*VPLUS)/(1.000+EPSIN(N))
  UPL=UPLUS
  CONTINUE
  DO 100 N=2,IE
    EPSIN(N)=EPSIN(N)/AMUINF
  CONTINUE
100
110 CALCULATE THE OUTER EDDY VISCOSITY
  KLEB IS THE KLEBANOFF INTERMITTANCY FACTOR
C
120 DO 120 N=1,IE
  XSUB=1.000/(1.000+5.500*(Y(N)/YSUBL)**6)
  EPSOUT(N)=RHOINF*ALAMDA**2*YSUBL**2*SCALAR(N)*KLEB/AMUINF
  CONTINUE
  OUT=0.000
  DO 130 N=1,IE
  DO 130 N=1,IE
  IF (OUT.EQ.1.000) GO TO 130
  IF (EPSIN(N).GE.EPSOUT(N)) GO TO 130
  EPSOUT(N)=EPSIN(N)
  GO TO 140
  OUT=1.000
  EPSOUT(N)=EPSOUT(N)
  CONTINUE
130
140

```


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```

DZUEWA=DZUEDW
DVEWDW=DVEGDW
DVEWAX=DVEGAX
IF (LC.NE.1) GO TO 40
DVEWAX=DVEGXX
DVMWAX=DVGXX
CONTINUE
Z0=ZB00
ZB00=Z2
X=X1

```

40

C
C
C

```

PROPERTIES AT THE BOUNDARY LAYER EDGE ARE CALCULATED
IF (NOSE.EQ.BLUNT.AND.L.EQ.1) PNC=DSJRT(2.000*DUE20X)
DUDX=DUE2DX
DVIDX=DVE2DX
DRODX=DRE2DX
IF (L.EQ.1) GO TO 50
X=X-0.0001
X=X-0.0001
IF (LCI.LT.1.000) RW=REW
X=X-0.0001
IF (LCI.EQ.1.000) GO TO 50
X=X-0.0001
CALL EDGCOF (LC)
RETURN
END

```

50

```

SUBROUTINE ENERGY (LC)
IMPLICIT REAL*8(A-H,O-Z)
REAL*8 NOSE
COMMON /RSTRM/ RHOINF,PINF,TFS,UFS,AMUINF,CP,REINF,PR
COMMON /GEOM/ ALPHA,NOSE,RNOSE,WLST,X,XX,MX
COMMON /GEOCF/ B1,B2,B3,B4,B5,B6,AL1,AL2,AL3,XLAMB
COMMON /GEOCF/ HXOHX0(101),HMOHMO(101),BHV(101),BHT(101),
1,XXX(101),KXW(101),BMX(101),XLM(101),XLM(101),BAGF(101),BAGF(101),
2AGF(101),BAGF(101),BAGG(101),KXOAL(101)
COMMON /INTEG/ IE,IM,KEND,KEND2,KLX,K,L,NBLNT1,IND,KPRI,LPRI,KPR,
1LPR
COMMON /PDECOF/ AO(101),A2(101),A3(101),A4(101),A5(101)
COMMON /SOLPN/ VM(101),GM(101),TW(101),GM(101),FM(101),FW(101),
1TW(101),XIM,XM,RM
COMMON /TRANS/ KTRANS,KONSET,XIF,CH12(101),CHTMAX,XBAR
COMMON /IRBLNT/ ASTAR,AKSTAR,ALAMDA,YSUBL,EVSCTY(101),EDYLAW,EPLUSE
1(101),LAMTRB

```

ENER 10
ENER 20
ENER 30
ENER 40
ENER 50
ENER 60
ENER 70
ENER 80
ENER 90
ENER 100
ENER 110
ENER 120
ENER 130
ENER 140
ENER 150
ENER 160
ENER 170
ENER 180

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SUBROUTINE FD3 (X,X1,X2,X3,F1,F2,F3,FX)	FD30 10
IMPLICIT REAL*8 (A-H,O-Z)	FD30 20
	FD30 30
SUBROUTINE FD3 CALCULATES THE FIRST DERIVATIVE-FX-CORRESPONDING	FD30 40
TO POINT X USING 3 POINT LAGRANGIAN DIFFERENTIATION FORMULA	FD30 50

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CC

```
      ASSUMES X1 .LE. X .LE. X3.  
      A1=2.0*X-X2-X3  
      A2=2.0*X-X1-X3  
      A3=2.0*X-X1-X2  
      D1=(X1-X2)*(X1-X3)  
      D2=(X2-X1)*(X2-X3)  
      D3=(X3-X1)*(X3-X2)  
      C1=A1/D1  
      C2=A2/D2  
      C3=A3/D3  
      FX=C1*F1+C2*F2+C3*F3  
      RETURN  
      END
```

FD30 60
FD30 70
FD30 80
FD30 90
FD30 100
FD30 110
FD30 120
FD30 130
FD30 140
FD30 150
FD30 160
FD30 170
FD30 180
FD30 190
FD30 200

```
      SUBROUTINE FDS (X,X1,X2,X3,X4,X5,F1,F2,F3,F4,F5,FX)  
      IMPLICIT REAL*8 (A-H,O-Z)
```

```
      SUBROUTINE FDS CALCULATES THE FIRST DERIVATIVE-FX-CORRESPONDING  
      TO POINT X USING 5 POINT LAGRANGIAN DIFFERENTIATION FORMULA
```

CCCCC

```
      ASSUMES X1 .LE. X .LE. X5.  
      A1=(X-X4)*(X-X5)*(X-X2-X3)+(X-X2)*(X-X3)*(X-X4-X5)  
      A2=(X-X4)*(X-X5)*(X-X1-X2)+(X-X1)*(X-X3)*(X-X4-X5)  
      A3=(X-X4)*(X-X5)*(X-X1-X2)+(X-X1)*(X-X2)*(X-X4-X5)  
      A4=(X-X3)*(X-X4)*(X-X1-X2)+(X-X1)*(X-X2)*(X-X4-X5)  
      A5=(X-X3)*(X-X4)*(X-X1-X2)+(X-X1)*(X-X2)*(X-X4-X5)  
      D1=(X1-X2)*(X1-X3)*(X1-X4)*(X1-X5)  
      D2=(X2-X1)*(X2-X3)*(X2-X4)*(X2-X5)  
      D3=(X3-X1)*(X3-X2)*(X3-X4)*(X3-X5)  
      D4=(X4-X1)*(X4-X2)*(X4-X3)*(X4-X5)  
      D5=(X5-X1)*(X5-X2)*(X5-X3)*(X5-X4)  
      C1=A1/D1  
      C2=A2/D2  
      C3=A3/D3  
      C4=A4/D4  
      C5=A5/D5  
      FX=C1*F1+C2*F2+C3*F3+C4*F4+C5*F5  
      RETURN  
      END
```

FD50 10
FD50 20
FD50 30
FD50 40
FD50 50
FD50 60
FD50 70
FD50 80
FD50 90
FD50 100
FD50 110
FD50 120
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FD50 250
FD50 260

AD-A051 971

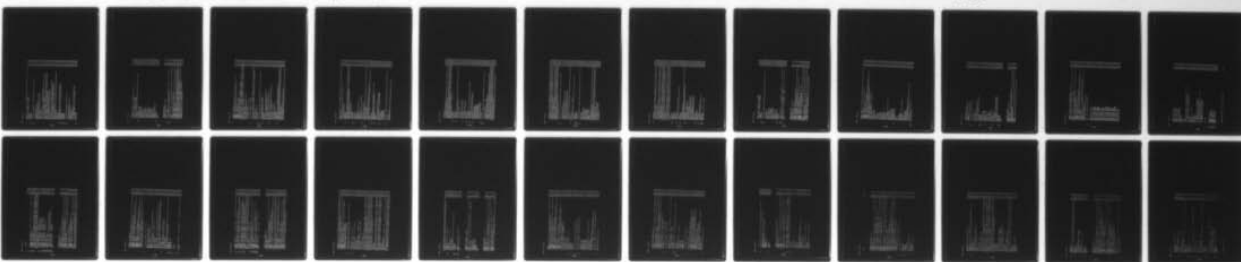
VIRGINIA POLYTECHNIC INST AND STATE UNIV BLACKSBURG --ETC F/G 20/4
THREE-DIMENSIONAL INCOMPRESSIBLE BOUNDARY LAYERS ON BLUNT BODIE--ETC(U).
MAY 77 D L DWOYER, C H LEWIS, P R GOGINENI

UNCLASSIFIED

VPI/SU-AERO-063-PT-2

NL

3 OF 3
AD A
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30  BAFF(I)=(DUEWDX/HXOHXO(I))*2*HXN(I)+DSORT(DUEWDX**3/2.D0)/HXOHXO
    1(I)**3/DSORT(2.D0/DUEWDX**3)
    BAGG(I)=DUEWDX/2.D0/DUEWDX**2
    RETURN
    CONTINUE
    .CALCULATE GEOMETRIC COEFFICIENTS FOR A GENERAL POINT
    DO 70 I=1,IE
    HXOHXO(I)=HX(I)/HX(I)
    HMOHMO(I)=HW(I)/HW(I)
    BHV(I)=HWN(I)/HW(I)+HXN(I)/HX(I)
    BHM(I)=HXN(I)/HX(I)-HWN(I)/HW(I)
    BHT(I)=BHV(I)
    KXX(I)=0.D0
    IF (K.NE.1) KXX(I)=XI*HXW(I)/HX(I)/HW(I)
    IF (K.EQ.1) KXOAL(I)=XI*HXW(I)/HW(I)/DVEWDX
    KKW(I)=XI*HWN(I)/HW(I)
    BMX(I)=(HWN(I)*HXN(I)/HW(I)-HWN(I)**2/HX(I)+HXN(I)**2/HX(I))/HW(I)
    BMM(I)=(HWN(I)*HXN(I)/HW(I)-HWN(I)**2/HX(I)+HXN(I)**2/HX(I))/HW(I)
    XLX(I)=XI*HWN(I)/HW(I)
    XLW(I)=XI*HWN(I)/HW(I)
    BAFFC=DUEWDX**2*HXN(I)/HXOHXO(I)**2/HX(I)
    BAFFC=0.D0
    IF (K.NE.1) BAFFC=DUEWDX**2*HWN(I)/HMOHMO(I)**2/HW(I)
    BAFF(I)=XI*BAFFC/2.D0/HX(I)/UEW**2*ETAX*HN(I)
    BAGF(I)=XI*BAFFC/2.D0/HX(I)/UEW**2*ETAX*HN(I)
    IF (K.NE.1) GO TO 50
    BAFF(I)=BAFFC*DSORT(XI/2.D0/UEW**3)/HW(I)*D2UEW/DVEWDX*HN(I)
    BAGG(I)=0.D0
    GO TO 60
    CONTINUE
    BAGG(I)=XI*BAFFC/2.D0/HW(I)/UEW**2*ETAW*HN(I)
    BAGG(I)=XI*BAFFC/2.D0/HW(I)/UEW**2*ETAW*HN(I)
    CONTINUE
    CONTINUE
    RETURN
    CONTINUE
    .CALCULATE COEFFICIENTS WITHOUT TVC
    IF (L.NE.1) GO TO 90
    KXX(I)=HXW(I)/HX(I)*HW(I)
    KKW(I)=1.D0/HX(I)
    XLX(I)=0.D0
    XLW(I)=0.D0
    IF (DABS(DVEWDX).GT.1.D-05) KXOAL(I)=0.D0
    IF (DABS(DVEWDX).LE.1.D-05) KXOAL(I)=3.D0
    GO TO 100
    CONTINUE

```

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 1000
 1010
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[illegible]

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```

1*AM=(D3RMMW/R0-2.00*D2RDMW*DR0DX/R0**2+2.70*DR0DMW**2*D2RDMW/R0**3)HOFN 710
2*AM/2.00/PIHOFN 720
TMMX=(D2RDMW/R0-DR0DMW*DR0DX/R0**2)*AM/2.00/PIHOFN 730
IF (L.EQ.1) GO TO 50HOFN 740
TMMX=(TMM-TMS(K))/DXIHOFN 750
TMS(K)=TMMHOFN 760
CONTINUEHOFN 770
TMMX=-TMM*(DR0DMW*DR0DX/2.00*PI**2/R0**2-DR0DMW**2*DR0DX/R0**3/2.00)HOFN 780
1/PI**2)*AM+(D3RMMW/R0-2.00*D2RDMW*DR0DX/R0**2-DR0DMW*DR0DX/R0**2+2.00)HOFN 790
2R0DMW*DR0DX**2*2.00/R0**3)*AM/2.00/PIHOFN 800
IF (L.EQ.1) GO TO 60HOFN 810
TMMX=(TMM-TMS(K))/DXIHOFN 820
TMS(K)=TMMHOFN 830
CONTINUEHOFN 840
GO TO 80HOFN 850
CONTINUEHOFN 860
TMM=0.00HOFN 870
TMMX=D3RMMW/DR0DX/2.00/PIHOFN 880
TMS(K)=TMMHOFN 890
TMS(K)=TMMHOFN 900
TMMX=D3RMMW*DR0DX/DR0DX**2/PIHOFN 910
CONTINUEHOFN 920
IF (K.GT.1) GO TO 100HOFN 930
TMMX=D3RMMW*AX/DR0DXHOFN 940
TMMX=-2.00*DR0DMW*DR0DX/DR0DX**2+(-D2RDMW*DR0DX/DR0DX**2+D2RDMW/DR0DX**2+D2RDMW/R0**3)HOFN 950
1RMMX/DR0DX)*AXHOFN 960
IF (L.EQ.1) GO TO 90HOFN 970
TMMX=(D3RMMW/R0-D2RDMW*DR0DX/R0**2)/2.00/PIHOFN 980
TMMX=(D2RDMW/R0)**3/2.00/PI**2+DR0DMW/R0-D3RMMW*DR0DX/R0**2)/2.00/PIHOFN 990
10/PIHOFN 1000
GO TO 100HOFN 1010
CONTINUEHOFN 1020
TMMX=(D3RMMW/DR0DX-D3RMMW*D2RDMW/DR0DX)/2.00/PIHOFN 1030
TMMX=-DR0DMW*DR0DX/DR0DX**2/PIHOFN 1040
CONTINUEHOFN 1050
TMM=0HOFN 1060
IF (TMM.EQ.0) GO TO 110HOFN 1070
WRITE (6,330) TX,TX,TMM,TMMHOFN 1080
WRITE (6,330) TMM,TMM,TMMHOFN 1090
CONTINUEHOFN 1100
SIX=DSIN(TX)HOFN 1110
SIX=DCOS(TX)HOFN 1120
SIX=DSIN(TX)HOFN 1130
SIX=DCOS(TX)HOFN 1140
CTM=DCOS(TM)HOFN 1150
HOFN 1160
HOFN 1170
HOFN 1180
HOFN 1190
HOFN 1200
HOFN 1210

```

C C CALCULATE RADIUS AND DERIVATIVES OF RADIUS FOR PROFILE POINTS

```

IINC=5
DO 210 I=2,15
R=DSQRT(R0**2+4*XN(I)*CTX(I)**2+2.00*R0*XN(I)*CTX*CTM)

```


[illegible]

[illegible]

190
200
210

[illegible]

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```
310 CALL DERIV (HKN,ETA,IE,1,HXNN)
C    CALL DERIV (HW,ETA,IE,1,HWN)
C    CALL DERIV (HMN,ETA,IE,1,HMNN)
C    RETURN
C    CONTINUE
C    CALCULATE METRIC COEFFICIENTS FOR PROFILE POINTS WHEN TVC = NO
DO 320 I=1,1E
  HX(I)=HXO
  HW(I)=HWO
  HMN(I)=DHXQDX
  HXN(I)=DHXQDX
  HMN(I)=DHXQDX
  HXN(I)=DHXQDX
  HXN(I)=0.00
  HXN(I)=0.00
  HMN(I)=0.00
  HMN(I)=0.00
  CONTINUE
  RETURN
320
C
330 FORMAT (IHO,SHIX =,E13.6,6H IM =,E13.6,7H TXX =,E13.6,6HIXW =,
340 E13.6,7H TXX =,E13.6,7H TXX =,E13.6)
  FORMAT (IHO,4H =,E13.6,6H RX =,E13.6,6H RW =,E13.6,7H RXX =,E13.6,7H
  113.6,7H RWX =,E13.6,7H RW =,E13.6)
  END
```

```
SUBROUTINE INIT
IMPLICIT REAL*8 (A-H,O-Z)
REAL*8 NOSE
COMMON /ASSVAR/ IFL,KBL
COMMON /CONVRG/ CONV,NIT1,NIT2,NIT3,NIT
COMMON /DEVAR/ F(2,101,3),FN(2,101,3),G(2,101,3),GN(2,101,3),T(2,
101,3),TNI(2,101,3),Y(101),VOL(101)
COMMON /FRSTRM/ RHOINF,PINF,TFS,UFS,AMUINF,CP,REINF,PR
COMMON /GASPRP/ PRANDT(101)
COMMON /GEOM/ ALPHA,NOSE,RNOSE,HLST,X,XX,WX
COMMON /INJECT/ INJCT,NOINJ,MASTRN
COMMON /INTEGR/ IE,IM,KEND,KEND2,KLX,K,L,NBLNTI,IND,KPRT,LPRT,KPR,
ILPR
COMMON /SOLPNT/ VM(101),GW(101),TW(101),GWN(101),FVN(101),FW(101),
LTMN(101),XTW,XM,RM
COMMON /SURFAS/ CMALL,CHIND,PEWIND,VWALL,TWALL,XTW(500),TWX(500),X
IC(1500),C(1500),TCONJ,KCI,XTW
COMMON /TRANSN/ KTRANS,KONSET,XIF,CHI2(101),CHIMAX,XBAR
INIT 10
INIT 20
INIT 30
INIT 40
INIT 50
INIT 60
INIT 70
INIT 80
INIT 90
INIT 100
INIT 110
INIT 120
INIT 130
INIT 140
INIT 150
INIT 160
INIT 170
INIT 180
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COMMON /IRBLNT/ ASTAR,AKSTAR,ALAMDA,YSUBL,EVSCTY(101),EDYLAW,EPLUS
1 I(101),LAMTR8
COMMON /TMPTR/ TEMP(101),IP(101),RTM,TB
COMMON /XSOLVE/ DW,M,XI,DXI,XIOLD
COMMON /XCORDE/ XSTA(100),DXMAX,DX,DXOLD,DXI,NSOLVE
COMMON /XSOLVE/ XSTA(100),DXMAX,DX,DXOLD,DXI,NSOLVE
COMMON /ZCQOQD/ ETAINF,ETAFAC,ETA(101),DETA(101),KADETA
COMMON /OLD/ VOL(61),CVOLD(61),ROLD(61)
DATA BLUNT,SHARP,SHBLUNT,SHSHARP/
READ (10) ALPHA,KLX
ALPHA=ALPHA*DARCOST-1.000)/180.000
WLST=0.500
ZWOLD=1.000
SINLST=0.000
X=0.000
DX=DXI
XX=0.000
XI=0.000
DXI=0.000
XXI=0.000
XII=0.000
RW=0.000
XW=0.000
XIOLD=0.000
DXOLD=DX
XIF=0.000
IF (LAMTR8.EQ.2) XIF=1.000
CWINO=CHAL
VWALL=0.000
DO 10 N=1,IE
PRANDINI=0.900
EPLUSINI=0.000
EVLZINI=0.000
EVSCTYINI=0.000
IND=1
KBL=1
L=0
NIT=0
LPR=LPRIT
KEND=KEND2
IF (ALPHA.LT.1.0-08) KEND=1
IM=IE-1
DW=1.000
DETA(1)=0.000
ETA(1)=0.000
ETA(1)=ETAINF
IF (ETAFAC.EQ.1.000) GO TO 20
DETA(2)=ETAINF*(ETAFAC-1.000)/(ETAFAC**IM-1.000)
GO TO 30
DETA(2)=ETAINF/DFLOAT(IM)
ETA(2)=DETA(2)

```

10

20
30


```

40      DO 40 I=2,IM
      IFAC=DETAL(I-1)
      DETAL(I)=DETAL(I-1)+DETA(I)
      CONTINUE
      DETA(IE)=DETALIM)*ETAFAC
      CALCULATE INITIAL PROFILES
      DO 70 I=1,N
      DO 50 J=1,N
      IF(I,JE) GO-DEXP(-ETALJ))
      G(I,J,N)=0.7500*F(I,J,N)
      IF(I,JE) GO-DEXP(-ETALJ))
      CONTINUE
      CONTINUE
      DO 80 I=1,61
      VOL(I)=0.00
      ROLD(I)=0.00
      CONTINUE
      DO 90 J=1,IE
      Y(I,J)=0.000
      VOL(J)=0.000
      G(I,J)=G2(J,I)
      F(I,J)=F1(J,I)
      CONTINUE
      DO 100 I=1,2
      DO 110 N=1,3
      F(I,N,ETA,IE,I,N)
      G(I,N,ETA,IE,I,N)
      F(I,N,ETA,IE,I,N)
      G(I,N,ETA,IE,I,N)
      CONTINUE
      CONTINUE
      RETURN
      END

```

```

SUBROUTINE INPUT
IMPLICIT REAL*8 (A-H,O-Z)
REAL*8 NOISE
COMMON /CONVRG/, CONV,NIT1,NIT2,NIT3,NIT
COMMON /EDGE/, UEDG,TEDG,VEDG,DUEGX,DUEGY,RE
DG,DROGDG,X,ZBOD,XEDG
COMMON /FINDIF/, A(10),BB(10),CC(10),DD(10),E(10)
J=1
I=CR1
COMMON /FRSTRM/, RHOINF,PINFTFS,UFS,AMUINF,CP,REINF,PR
INPUT
INPUT
INPUT
INPUT
INPUT
INPUT
INPUT
INPUT
INPUT
INPUT

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```

COMMON /GEOM/ ALPHA, NOSE, RNOS, WLST, X, XX, XX
COMMON /INJECT/ INJCT, NOINJ, MASTRN
COMMON /INTER/ IE, IM, KEND, KEND2, KLX, K, L, NBLNT1, IND, KPRT, LPRT, KPR,
INPU 100
INPU 110
INPU 120
INPU 130
INPU 140
INPU 150
INPU 160
INPU 170
INPU 180
INPU 190
INPU 200
INPU 210
INPU 220
INPU 230
INPU 240
INPU 250
INPU 260
INPU 270
INPU 280
INPU 290
INPU 300
INPU 310
INPU 320
INPU 330
INPU 340
INPU 350
INPU 360
INPU 370
INPU 380
INPU 390
INPU 400
INPU 410
INPU 420
INPU 430
INPU 440
INPU 450
INPU 460
INPU 470
INPU 480
INPU 490
INPU 500
INPU 510
INPU 520
INPU 530
INPU 540
INPU 550
INPU 560
INPU 570
INPU 580
INPU 590
INPU 600

COMMON /STAG/ PSTAG, ISTAG, PNC, QWSTAG
COMMON /SURFAS/ CWALL, CWIN, PCWIN, VWALL, TWALL, XTW(500), TWX(500), X
COMMON /CIX(500)/ CCONM, KCI, RTW
COMMON /TITLE/ LABEL(20)
COMMON /TEMP(101)/ TP(101), RTW, TB
COMMON /TRANS/ KTRANS, KONSET, XIF, CH12(101), CHIMAX, XHAP
COMMON /TRBLNT/ ASTAR, AKSTAR, ALAMDA, YSUBL, EVSCIY(101), EOYLAW, EPLUS
INPU 101, LAMTRB
COMMON /XSOLVE/ XSTA(100), DXMAX, DX, DXOLD, DXI, NSOLVE
COMMON /ZCOORD/ ZTAINF, ETAFAC, ETA(101), DETA(101), ADTEST, KADETA
COMMON /TVCURV/ SFC
DATA YES, NO, 3H YES, 3H NO
DATA BLUNT, SHARP, 5H BLUNT, 5H SHARP/

THE INPUT QUANTITIES ARE READ IN
J=5
READ (J,70) LABEL
READ (J,50) IE
READ (J,50) INJCT
READ (J,50) KADETA
READ (J,50) KEND2
READ (J,50) KONSET
READ (J,50) KPRT
READ (J,50) KTRANS
READ (J,50) LAMTRB
READ (J,50) LPRT
READ (J,50) MTL
READ (J,50) MTL2
READ (J,50) MTL3
READ (J,50) NOINJ
READ (J,100) SFC
READ (J,50) NSOLVE
READ (J,60) ADTEST
READ (J,60) AKSTAR
READ (J,60) ALAMDA
READ (J,60) ASTAR
READ (J,60) CWALL
READ (J,60) CONV
READ (J,60) DXMAX
READ (J,60) DXI
READ (J,80) EDYLAW
READ (J,60) ETAFAC
READ (J,60) ETAINF
READ (J,60) PR
READ (J,60) RTW
READ (J,60) TFS

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CCC

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10 READ (J,60) CP
   READ (J,60) AMUINF
   READ (J,60) PSTAG
   READ (J,60) PINF
   READ (J,60) XBAR
   READ (J,60) UFS
   DO 10 I=1,NSOLVE
     READ (J,40) XSTAIL(I)
   CONTINUE
20 KTCI=0
   I=0
   IF (J,90,END=30) XTW(I),XTW(I),XCI(I),CIX(I)
   READ (J,90,END=30) XTW(I),XTW(I)=XCI(I)
   IF (CIX(I),EQ,0.000) XCI(I)=XTW(I)
   IF (CIX(I),EQ,0.000) XCI(I)=XTW(I)
   GO TO 20
30 KTW(I),CIX(I),EQ,0) KCI=I-1
   IF KCI=0
     MASTRN=0
     NOSET=BLUNT
     NOISE=1.00
     CRI=1.00
     IF (KAMSTRB,EQ,0) KONSET=NSOLVE
     IF (INJCT,EQ,0) KONSET=NSOLVE
     IF (INJCT,EQ,0) INJCT=NSOLVE
     IF (INJCT,EQ,SHARD,AND,INJCT,EQ,1) INJCT=2
     IF (INJCT,EQ,1) MASTRN=1
     IF (MASTRN,EQ,1) INJCT=NSOLVE
     IF (NOINJ,EQ,0) NOINJ=NSOLVE
   RETURN
40
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80
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100

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INT3 10

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CCCCC
      IMPLICIT REAL*8(A-H,O-Z)
      SUBROUTINE INTER3 INTERPOLATES FOR THE VALUE F CORRESPONDING TO
      POINT X USING 3 POINT LAGRANGIAN INTERPOLATION.
      ASSUMES X1 .LE. X .LE. X3.
      A1=(X-X2)*(X-X3)
      A2=(X-X1)*(X-X3)
      A3=(X-X1)*(X-X2)
      D1=(X1-X2)*(X1-X3)
      D2=(X2-X1)*(X2-X3)
      D3=(X3-X1)*(X3-X2)
      C1=A1/D1
      C2=A2/D2
      C3=A3/D3
      F=C1*F1+C2*F2+C3*F3
      RETURN
      END
INT3 20
INT3 30
INT3 40
INT3 50
INT3 60
INT3 70
INT3 80
INT3 90
INT3 100
INT3 110
INT3 120
INT3 130
INT3 140
INT3 150
INT3 160
INT3 170
INT3 180
INT3 190
INT3 200

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CCCCCCC
      SUBROUTINE INTER5 (X,X1,X2,X3,X4,X5,F1,F2,F3,F4,F5,F)
      IMPLICIT REAL*8 (A-H,O-Z)
      SUBROUTINE INTER5 INTERPOLATES FOR THE VALUE F CORRESPONDING TO
      POINT X USING 5 POINT LAGRANGIAN INTERPOLATION FORMULA.
      ASSUMES X1 .LE. X .LE. X5.
      A1=(X-X2)*(X-X3)*(X-X4)*(X-X5)
      A2=(X-X1)*(X-X3)*(X-X4)*(X-X5)
      A3=(X-X1)*(X-X2)*(X-X4)*(X-X5)
      A4=(X-X1)*(X-X2)*(X-X3)*(X-X5)
      A5=(X-X1)*(X-X2)*(X-X3)*(X-X4)
      D1=(X1-X2)*(X1-X3)*(X1-X4)*(X1-X5)
      D2=(X2-X1)*(X2-X3)*(X2-X4)*(X2-X5)
      D3=(X3-X1)*(X3-X2)*(X3-X4)*(X3-X5)
      D4=(X4-X1)*(X4-X2)*(X4-X3)*(X4-X5)
      D5=(X5-X1)*(X5-X2)*(X5-X3)*(X5-X4)
      C1=A1/D1
      C2=A2/D2
      C3=A3/D3
      C4=A4/D4
      C5=A5/D5
      F=C1*F1+C2*F2+C3*F3+C4*F4+C5*F5
      RETURN
      END
INT5 10
INT5 20
INT5 30
INT5 40
INT5 50
INT5 60
INT5 70
INT5 80
INT5 90
INT5 100
INT5 110
INT5 120
INT5 130
INT5 140
INT5 150
INT5 160
INT5 170
INT5 180
INT5 190
INT5 200
INT5 210
INT5 220
INT5 230
INT5 240
INT5 250
INT5 260

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SUBROUTINE OUT1
IMPLICIT REAL*8 (A-H,O-Z)
REAL*8 NOSE
COMMON /CONVRG/ CONV,NIT1,NIT2,NIT3,NIT
COMMON /FINDIF/ A(101),BB(101),BI(101),CC(101),DD(101),D(101),E(101),
1)C(1)
COMMON /FRTRM/ RHOINF,PINF,TFS,UF5,AMUINF,CP,KEINF,PR
COMMON /GEOM/ ALPHA,NOSE,RNOSE,MLST,X,XX,XX
COMMON /INJECT/ INJCT,NOINJ,MASTRN
COMMON /INTEGR/ IE,IM,KEND,KEND2,KLX,K,L,NBLNT1,IND,KPRT,LPR1,KPR,
1)LPRT
COMMON /REF/ PREF,TREF,AMUREF
COMMON /STAG/ PSTAG,TSTAG,PNC,QNSTAG
COMMON /SURFAS/ CWALL,CWIND,PENIND,VWALL,TWALL,XTW(500),TWX(500),
1)CI(500),CJ(500),TCOM,KCI,KTM
COMMON /TEMPFR/ TEMPI(101),TPI(101),RTM,TB
COMMON /TRBLNT/ ASTAR,AKSTAR,ALANDA,YSUBL,EVSCTY(101),EDYLAM,EPLUS,
1)LOI,LAMTRB
COMMON /XCORDD/ XI,XI,DXI,XIOLD
COMMON /XSOLVE/ XSTAI(100),DXMAX,DX,DXGLD,DXI,NSOLVE
COMMON /ZCORDD/ ZSTAI(100),ETAINF,ETA(101),DETA(101),ADTEST,KADETA
COMMON /TITLE/ LABEL(20)
DATA BLUNT,SHARP,SHBLUNT,SHSHARP/
ATTN=ALPHA*180.000/DARCOS(-1.000)
WRITE(6,100) (LABEL(I),I=1,20)
WRITE(6,100)
WRITE(6,401) PSTAG
WRITE(6,901) TSTAG
WRITE(6,701) PNC
WRITE(6,801) PINF
WRITE(6,1001) UFS
WRITE(6,901) TFS
WRITE(6,1501) RHOINF
WRITE(6,1601) AMUINF
WRITE(6,1101) RTW
WRITE(6,1201) ATAK
WRITE(6,1701) CP
WRITE(6,1801) REINF
WRITE(6,1901) PR
WRITE(6,201)
DO 10 I=1,NSOLVE
WRITE(6,301) XSTAI(I)
CONTINUE
WRITE(6,1301)

```

10
C

[illegible]SUBROUTINE OUT2
IMPLICIT REAL*8
(A-H,O-Z)

[illegible]

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60  FORMAT (10X,6HXI  =,E13.6,4X,6HDXI  =,E13.6,4X,6HCWALL=,E13.6,5X,  OUT2 700
16HNIT  =,E13.6,5X,6HNDIMENSIONAL EDGE PROPERTIES//)  OUT2 710
70  FORMAT (10X,6HTE  =,E13.6,5X,6HUE  =,E13.6,5X,6HVE  =,E13.6,10X,OUT2 720
80  1,6HUEDX=,E13.6,5X,6HVEDX=,E13.6,5X,6HUEDW=,E13.6,5X,6HVEDW=,E13.6,10X,OUT2 730
23.6//)  OUT2 740
90  FORMAT (10X,6HNDIMENSIONAL BOUNDARY LAYER PARAMETERS//)  OUT2 750
100  FORMAT (10X,6HCFXINF=,E13.6,5X,7HCFXEDG=,E13.6,5X,7HCFXINF=,E13.6,  OUT2 760
15X,7HCFWEDG=,E13.6,10X,7HQM  =,E13.6,5X,7HCHIMAX=,E13.6//)  OUT2 770
110  FORMAT (10X,6HNDIMENSIONAL BOUNDARY LAYER PARAMETERS//)  OUT2 780
120  1A(1X) =,E13.6,5X,12HTHETA1X) =,E13.6,10X,27HTRANSVERSE SKIN FRICTION=,E13.6,5X,12HDELTAOUT2 800
27ION  =,E13.6,4H PSF, 5X,12HDELTA1A(PHI)=,E13.6,5X,11HTHETA1(PHI)=,E13.6,5X,12HDELTA 810
33.6//10X,27HSMALL HEAT TRANSFER RATE =,E13.6,4H BTU, 5X,12HDELTA 820
41) =,E13.6,  OUT2 830
130  FORMAT (10X,6H 48X, 3(5H****,10X)//)  OUT2 840
140  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 850
150  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 860
160  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 870
170  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 880
180  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 890
190  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 900
200  10X,2HNTN (7X,3HETA1,1X,1HY,10X,1HF,10X,2HFN,9X,1HG,10X,2HGN,9X,1HT,  OUT2 910
END  OUT2 920
      OUT2 930
      OUT2 940

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```

SUBROUTINE PHIMOM (LC)
IMPLICIT REAL*8(A-H,O-Z)
COMMON /DEVAR/ F(2,101,3),FN(2,101,3),G(2,101,3),GN(2,101,3),T(2,
101,3),TNC(2,101,3),Y(101),YOL(101)
COMMON /GEOM/ ALPHA,NOSE,RNOSE,WLST,X,XX,XX,XX
COMMON /IECOEF/ E1,E2,E3,B4,B5,B6,AL1,AL2,AL3,XLANB
COMMON /INTEGR/ IE,IM,KENO,KENO2,KLX,K,L,NBLNFI,END,KPRT,LPRT,KPR,
LLPR
COMMON /PDECOF/ AO(101),AI(101),A2(101),A3(101),A4(101),A5(101),
COMMON /SOLPNT/ VM(101),GM(101),TW(101),GMN(101),FVN(101),FM(101),
ITWNN(101),X1W,XW,XW
COMMON /TRANSN/ KTRANS,KONSET,XIF,CHI2(101),CHIMAX,XBAR
COMMON /TRBLNT/ ASTAR,AKSTAR,ALAMDA,YSUBL,EVSCTY(101),EDYLAM,EPLU
(101),LAMTRB
COMMON /XICORD/ XI,XI,DXI,XIOLD
COMMON /ZCOORD/ ZCOORD,ETAC,ETA(101),DETA(101),ADTEST,KADETA
COMMON /NCORD/ NCORD,XN(101),ETAX,ETAX
COMMON /EDGM/ UE,VEM,TEM,QUEMOM,D2UEMW,DVEMOM,REM
COMMON /HETA/ HETA,HX(101),HM(101),HXX(101),HXM(101),HXMN(101),
1,HXNN(101),HXX(101),HM(101),HMN(101),HMNN(101)

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COMMON /GECQIF/ HXOHXO(101),HWOHWO(101),BHV(101),BHW(101),BHT(101),PHI
1,KXX(101),XKX(101),BMX(101),BMW(101),XLX(101),XLW(101),BAFF(101),BPHI
2,AGF(101),BAF(101),BAGG(101),XKXOAL(101)
COMMON /STGEQ/ DVEGAW,DVEWXX,DVEZXX,DVEGXX,DVEWXX,DVEZXX,DVGXXW,DPHI
1VMXXW,DV2XXW,DHWOHWO(101),XIFNXX(101),XIFNXX(101),XIFNXX(101)
COMMON /INTGL/ XIFNXX(101),XIFNXX(101),XIFNXX(101),XIFNXX(101)
COMMON /PHI/ DAL3DX,DAL3DW,ALB6
DIMENSION ROMUL(101),ROMULIN(101)
DATA SHARP,BLUNT/SHSHARP,5HBLUNT/

SUBROUTINE PHIMOM SETS UP THE COEFFICIENTS OF THE PARTIAL
DIFFERENTIAL PHI MOMENTUM EQUATION
DO 10 J=1,IE
ROMUL(J)=1.000+XIF*EPLUS(J)
CONTINUE
CALL DREIV (ROMUL,ETA,IE,I,ROMULN)
CALL PRESSI (2)
DO 40 J=1,IE
A0(J)=ROMUL(J)
A1(J)=ROMUL(J)-BHM(J)*ROMUL(J)-VM(J)
A2(J)=BHM(J)*ROMUL(J)
A3(J)=BHM(J)*ROMUL(J)/2.00/HX(J)**2
A4(J)=O.D0
IF (LC*EQ.1) GO TO 30
IF (LC*EQ.2) GO TO 20
A2S=-AL2**2*B6/HWOHWO(J)**2/2.00+AL2*(XLW(J)-XLW(1))/2.00+BAGG(J)
A2(J)=A2(J)+A2S*GW(J)*ROMULN(J)*HWN(J)/HW(J)-.500*B5*FW(J)/HXPHI(J)
10HXO(J)=A2(J)*2.00*A2S*GW(J)*FW(J)/HXOHOXO(J)**2
A3(J)=B5/2.00*AL3/HWOHWO(J)**2+AL3*HWOHWO(J)**2/2.00/HWOHWO(J)**2-XKX(J)*PHI
1HWOHWO(J)**2/2.00/HWOHWO(J)**2+AL3*HWOHWO(J)**2/2.00/HWOHWO(J)**2-XKX(J)*PHI
2HXOHOXO(J)**2/2.00/HWOHWO(J)**2+AL3*HWOHWO(J)**2/2.00/HWOHWO(J)**2-XKX(J)*PHI
3BAGG(J)*AL3**2-AL2*AL3**2*XLW(J)-XLW(1))/2.00-BAGG(J)*(1.00-FW(J)**2)*PHI
4XO(J)**2/HX(J)+AL3*AL3*FW(J)/2.00/HWOHWO(J)**2-A2S*GW(J)**2
A5(J)=HWN(J)*AL2*GW(J)/2.00/HW(J)**2
GO TO 40
CONTINUE
A2(J)=A2(J)-KXW(J)*FW(J)/2.00/HXOHOXO(J)**2-B5*FW(J)/2.00/HXOHOXO(J)*PHI
1*2+BAGG(J)*KXW(J)*ROMULN(J)*HWN(J)/HW(J)-ALB6/HWOHWO(J)**2*GW(J)/2*PHI
2.00
A3(J)=B5/2.00/HXOHOXO(J)**2+KXW(J)/HXOHOXO(J)**2/2.00-KXOAL(J)*(1.00-FW(J)
10-FW(J)**2)*HWOHWO(J)/2.00/HXOHOXO(J)**2-BAGG(J)*(1.00-FW(J)**2)-BAGG
2GG(J)*AL3**2-XI*XIFNXX(J)/HW(J)/UEW/DVEWXX+.500*ALB6/HWOHWO(J)**2
GO TO 40
CONTINUE
A2S=-.500*AL1*B6
A2(J)=A2(J)+.500*FW(J)*B5/HXOHOXO(J)**2-.500*FW(J)/HXOHOXO(J)**2+.00*A2S*PHI
1*GM(J)
A3(J)=AL1*B6/2.00-KXOAL(J)*(1.00-FW(J)**2)/2.00/HXOHOXO(J)**2+1BAGG(J)*PHI
1*GM(J)+BAGG(J)/DVEWXX**1.500-1.00*XIFNXX(J)/D5GAT(2.00*DUEWXX)-DSORT*PHI
212.00*DUEWXX*XIFNXX(J)/HWX(1)*DUEWXX*2.00/HWXX(1)/DUEWXX-APHI
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31**2)*XIFNM(J)/DSQRT(2.00*DUEDW(J)-ALL*XIFNMW(J)/DSQRT(2.00*DUEDW(J)**2-
4)*HMX(I)*.500*HMX(I)**2+.500/HMX(I)**2-A25*GM(J)**2
A4(I)=0.00
A5(I)=-ALL*GM(J)/2.00/HMX(I)
CONTINUE
IF (IRITE=0)
  IF (IRITE.EQ.0) GO TO 60
  WRITE (6,70)
  DO 50 I=1,IE-5
  WRITE (6,80) A0(I),A1(I),A2(I),A3(I),A4(I),A5(I)
CONTINUE
CONTINUE
RETURN
C
70 FORMAT (1H0,15HIN PHIMOM A0-A1)
80 FORMAT (1H0,6E13.6)
END

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SUBROUTINE PRESSI (IDIREC)
  IMPLICIT REAL*8 (A-H,O-Z)
  COMMON /INTEGR/ IEIM,KEND,KEND2,KLX,K,L,NBLNT1,IND,KPRT,LPRT,KPR,
  LPR
  COMMON /INTEGR/ XIFNX(101),XIFNM(101),XIFNMW(101),XIFNMW(101)
  DO 10 I=1,IE
  XIFNX(I)=0.00
  XIFNM(I)=0.00
  XIFNMW(I)=0.00
  XIFNMW(I)=0.00
CONTINUE
RETURN
END

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SUBROUTINE PROPTY
  IMPLICIT REAL*8 (A-H,O-Z)
  COMMON /DEPVAR/ F(2,101,3),FNU(2,101,3),G(2,101,3),GHI(2,101,3),T(2,
  101,3),TNI(2,101,3),Y(101,101),Y(101,101)
  COMMON /EDGZ/ TE2,UE2,VE2,DUEDX,DUEDW,DUE2W,DUE2W,RE2,XPR,
  LPR
  COMMON /FRTSM/ RHOINF,PINF,TFS,UFS,AMUINF,CP,REINF,PR
  COMMON /GASPR/ PRANDT(101)
  COMMON /GEOM/ ALPH,NOSE,RNOSE,WLST,X,XX,MX
  COMMON /INTEGR/ IEIM,KEND,KEND2,KLX,K,L,NBLNT1,IND,KPRT,LPRT,KPR,
  LPR

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1000      COMMON /OUTPUT/ CFWEDG,CFWINE,CFXEDG,CFXINF,DEL,QW,QMINF,QWQWO,S,PROF
1001      COMMON /TAX/ DELSTX,DELPHI,THETAX,THEPHI,REX2
1002      COMMON /STAG/ PSTAG,STAG,PNC,STAG
1003      COMMON /SURF/ CWall,Crind,PENID,VWall,TWall,XTW(500),TW(1500),XPROF
1004      COMMON /X1/ X1(500),TCONW,KCL,KTW
1005      COMMON /IMPR/ TR/TERP(100),TPI(100),RTW,TB
1006      COMMON /KTRANS/ KTRANS,KONSET,XIF,CHI2(100),CHIMAX,XBAR
1007      COMMON /TRBLNT/ ASTAR,AKSTAR,ALAMDA,YSUBL,EVSCTV(100),EDYLAW,EPLUS
1008      COMMON /XICORD/ XI,XI,DXI,XIOLD
1009      COMMON /XSOLVE/ XSTAL(100),DXMAX,DX,OXOLD,DXI,NSOLVE
1010      COMMON /ZCOORD/ ZTAFAC,ETA(100),DETA(100),ADTEST,KADETA
1011      DIMENSION U(100),Y2(100),Z2(100),F2(100),G2(100)
1012      DATA BLUNT,SHARP,SHBLUNT,SHSHARP/
1013      IF (NOSE.EQ.SHARP) S=X/XSTAL(NSOLVE)
1014      IF (NOSE.EQ.BLUNT) S=X/RNOSE
1015      DO 10 J=1,IE
1016      F2(J)=F1(J,2)
1017      G2(J)=G1(J,2)
1018      IF (K.EQ.1) Z2(J)=0.000
1019      T2(J)=T1(J,2)
1020      CONTINUE
1021      LPR1=LPR
1022      LPR=LPR1
1023      THE OUTPUT QUANTITIES ARE CALCULATED
1024      IF (L.EQ.1.AND.NOSE.EQ.SHARP) GO TO 50
1025      CALCULATE PHYSICAL NORMAL DISTANCE PROFILE
1026      DELSTX=0.000
1027      DELPHI=0.000
1028      THETAX=0.000
1029      THEPHI=0.000
1030      CHIMAX=0.000
1031      CHI2(1)=0.000
1032      G2IE=G2(IE,2)
1033      IF (DABS(G2IE).LT.1.0-10) G2IE=1.000
1034      Y(1)=0.000
1035      YOL(1)=0.000
1036      EPSVD=1.00/DSQRT(REINF)
1037      IF (XI.EQ.0.00) YTRANS=1.00/PNC*EPSVD
1038      IF (XI.GT.0.00) YTRANS=DSQRT(XI/(2.00*UE2))*EPSVD
1039      CALCULATE DISPLACEMENT AND MOMENTUM THICKNESSES
1040      DO 20 J=2,IE
1041      Y(J)=Y(J-1)+YTRANS*DETA(J)

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20      VOL(J)=Y(J)/KSTAIN(SOLVE)
21      DELSTX=DELSTX+YTRANS*(1.000-F(2,J,2))+(1.000-F(2,J-1,2))*DELTA(J)
22      L/2=DO
23      THETA=THETA+YTRANS*(F(2,J,2)*(1.000-F(2,J,2))+F(2,J-1,2)*(1.000-
24      F(2,J-1,2))*DELTA(J)/2.000
25      THEPHI=THEPHI+YTRANS*(G(2,J,2)/G2IE*(1.000-G(2,J,2)/G2IE)+G(2,J-1,
26      L2)/G2IE*(1.000-G(2,J-1,2)/G2IE))*DELTA(J)/2.000
27      DELPHI=DELPHI+YTRANS*(1.000-G(2,J,2)/G2IE)*(1.000-G(2,J,2)/G2IE))
28      1*DELTA(J)/2.00
29      1*DELTA(J)=Y(J)*2*FEN(2,J,2)*UE2*(1.000/YTRANS)
30      IF CHI2(J).GT.CHIMAX) CHIMAX=CHI2(J)
31      CONTINUE
32      IF IG(2,IE,2).EQ.0.00) DELPHI=0.00
33      IF IG(2,IE,2).EQ.0.00) DELPHI=0.00
34      CALCULATE BOUNDARY LAYER THICKNESS
35      DO 30 N=1,IE
36      U(N)=DSORT1(F(2,N,2)*2+G(2,N,2)**2)/(F(2,IE,2)*2+G(2,IE,2)**2)
37      IF (X.EQ.1) U(N)=F(2,N,2)
38      IF (U(N).GE.0.99500) N=N-1
39      IF (U(N).GE.0.99500) GO TO 40
40      CONTINUE
41      DEL=(Y(NN)*Y(NN+1)-Y(NN))*10.99500-U(NN))/(U(NN+1)-U(NN))
42      CONTINUE
43      KEX2=KEINF*UE2*X
44      IF EPSVD*.LT.1.D-25) EPSVD=0.00
45      CFMINF=0.00
46      CFMAXF=0.00
47      CFMEDG=0.00
48      CFMEDG=0.00
49      QMINF=0.00
50      IF (X.EQ.0.000) GO TO 60
51      CALCULATE SURFACE PARAMETERS
52      DMDETA=GM(2,1,2)
53      IF (K.EQ.1) DMDETA=0.000
54      IF XINF=2.000*EPSVD*UE2**1.5*FEN(2,1,2)*DSORT(2.00/XI)
55      CFMINF=2.000*EPSVD*UE2**1.5*DMDETA*DSORT(2.00/XI)
56      CFMEDG=2.000*EPSVD*UE2**1.5*DSORT(2.00/XI)
57      CFMEDG=2.000*EPSVD*UE2**1.5*DMDETA*DSORT(2.00/XI)
58      QMINF=EPSVD*UE2**1.5*DSORT(2.00/XI)*TN(2,1,2)/PR
59      QMINF=QMINF*(1.000)
60      TAUX=CFMINF*HOFN*UE2**2/2.000
61      TAUEIN=CFMINF*HOFN*UE2**2/2.000
62      QW=QMINF*HOFN*UE2**1.2860-03/PR
63      IF (L.EQ.2) AND(NOSE.EQ.SHARP) QWSTAG=QW
64      QWSTAG=QW/QWSTAG
65      GO TO 70
66      IF (NOSE.EQ.SHARP) GO TO 70

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CALCULATE HEAT TRANSFER FOR A BLUNT CONE STAGNATION POINT
QHINF=EPSVD*IE2*DSORT(2.00*DUE2DX)*IN(2,1,2)/PR
QHINF=QHINF*(1.000)
Q=QHINF*RHINF*UPS**3*1.286D-03/PR
QSTAG=Q
QMOQ=1.000
70
CONTINUE
RETURN
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SUBROUTINE SOLVE (M,WN,EEI,FFI,EDGBC)
IMPLICIT REAL*8(A-H,O-Z)
REAL*8 NOSE
COMMON /FINOIF/ A(101),B(101),B(101),B(101),B(101),D(101),D(101),E(101)
1,CRI
COMMON /INTEGR/ IE,IM,KEND,KEND2,KLX,KOM,L,NBLNT1,IND,KPRT,LPRT,K
1PR,LPR
COMMON /ZCQOHD/ ETAINF,ETAFAC,ETA(101),DETA(101),ADTEST,KADETA
DIMENSION EE(101),FF(101),W(2,101,3),WN(2,101,3)
SUBROUTINE SOLVE CALCULATES THE SOLUTION OF A GENERAL PARABOLIC
PARTIAL DIFFERENTIAL EQUATION WHEN THE P-0-0-0 IS REPRESENTED
BY A SYSTEM OF IMPLICIT, THREE-POINT FINITE-DIFFERENCE EQUATIONS.
THE THOMAS ALGORITHM AS SOLVED BY RICHTMEYER IS USED TO SOLVE THE
SYSTEM.
EE(1)=EEI
EE(2)=EEI
DO 10 J=2,IM
DO 10 I=1,2,IM
A=0
B=0
C=0
D=0
E=0
F=0
G=0
H=0
I=0
J=0
K=0
L=0
M=0
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[illegible]

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70      A52=8HV(J-1)
      CONTINUE
      VM(J)=VM(J-1)+(A31*DFDX(I)+A32*DFDX(I2+A41*DDGW1+A42*DDGW2+A51*VM(J)
      I+A52*VM(J-1)+A11*FW(J)+A12*FW(J-1)+A21*GW(J)+A22*GW(J-1))*DETA(J)/
      2*DO
      CONTINUE
      DIFV=0.00
      DO 90 J=2,IE
      DIFVP=DABSI(VM(J)-V(J))/VM(J)
      IF (DIFVP.GT.DIFV) DIFV=DIFVP
      V(J)=VM(J)
      CONTINUE
      IF (DIFV.LE.CONV) GO TO 100
      IIT=1
      GO TO 20
      CONTINUE
      RETURN
      END
```

100

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100      SUBROUTINE WALL
101      IMPLICIT REAL*8(A-H,O-Z)
102      REAL*8 NDE,VAR, F(2,101,3),FW(2,101,3),G(2,101,3),GN(2,101,3),T(2,
103      101,3),TW(2,101,3),FV(101,101)
104      COMMON /UEGG,UEGG,VEGG,DUEGDX,DUEGOW,D2UEWM,DVEGDX,DVEGOW,RE
105      ICG,URDGOX,ZBQD,XEDG,UEWM,DUEWDX,DUEWOW,D2UEWM,DVEWDX,DVEWOW,RE
106      COMMON /FERSTM,RHONF,PINF,UF,UF5,AMUINF,CP,REINF,PR
107      COMMON /GEOM/ALPHA,NOSE,RNDSE,WLST,X,XX,XX
108      COMMON /ECCOE/BI,B2,B3,B4,B5,B6,AL1,AL2,AL3,XLAMB
109      COMMON /INJECT/ INJCT,NOINJ,MASRN
110      COMMON /INTEGR/ IE,IM,KEND,KEND2,KLX,K,L,NBLN1,IND,KPRT,LPRT,KPR,
111      LPR
112      COMMON /SOLPNT/ VM(101),GM(101),TW(101),GN(101),FW(101),
113      ITW(101),XIM,XM,RW
114      COMMON /STAG/ PSTAG,STAG,PNC,WMSTAG
115      COMMON /SURFAS/ CHALL,CWINO,PEW,IND,VHALL,TWALL,XTH(500),TWX(500),XW
116      ICI(500),XIX(500),TCOM,KCI,KI
117      COMMON /IMPRT/ /TEMP(101),PI(101),RTW,TB
118      COMMON /XICORD/ XI,XXI,DAI,XIOLD
119      DATA BLUNT,SHARP,SHBLUNT,SHSHARP/
120      PI=0ARCO(1-1.000)
121      INTERPOLATE FOR VALUES OF CHALL AT THE WINDWARD STREAMLINE
122      EPSVD=1.00/DSQRT(REINF)
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10 IF (MASTRN.EQ.0) GO TO 30
   IF (K.GT.1) GO TO 20
   PEWIND=PEW
   IF (KCI.EQ.0) GO TO 20
   IF (XW.GT.XCI(KCI)) GO TO 30
   J=0
   J=J+1
   IF (XW.GT.XCI(J)) GO TO 10
   IF (J.LT.2) J=2
   IF (J.GT.KCI-1) J=KCI-1
   CALL INTER3 (XW,XCI(J-1),XCI(J),XCI(J+1),CIX(J-1),CIX(J),CIX(J+1))
   CMIN=CMIN
   GO TO 40
   CMIN=CWIND
   GO TO 40
   CONTINUE
   IF (MX.LT.90.000) GO TO 40
   CMIN=0.000
   CONTINUE

20 INTERPOLATE FOR VALUES OF THALL AT THE WINDWARD STREAMLINE

   IF (K.GT.1) GO TO 70
   IF (KTH.EQ.0) GO TO 60
   J=0
   J=J+1
   IF (XW.GT.XTH(J)) GO TO 50
   IF (J.LT.2) J=2
   IF (J.GT.XTH-1) J=XTH-1
   CALL INTER3 (XW,XTH(J-1),XTH(J),XTH(J+1),TWX(J-1),TWX(J),TWX(J+1))
   CMIN=CWIND
   GO TO 70
   CMIN=CWIND
   GO TO 70
   CONTINUE

60 CALCULATE THE VALUE OF BIG V AT THE WALL
   VMALL=0.00
   IF (INOSE.EQ.SHARP.AND.XIW.EQ.0.000) GO TO 90
   IF (XTH.EQ.0.000) GO TO 80
   VMALL=DSQRT(XI/2.00/UEW)*CMIN
   GO TO 90
   VMALL=DSQRT(.500/DUEWDX)*CMIN
   CONTINUE
   RETURN
END

```

SUBROUTINE XNOM (LC)

XNOM 10

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10

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```
20      A5(J)=-X1*GW(J)*AL2/2.00/HW(J)/HWOHW(J)
        GO TO 30
        CONTINUE
        A2(J)=A2(J)-.500*FW(J)/HXOHX(J)**2-AL1**2*GW(J)
        A3(J)=-.500/HXOHX(J)**2+AL2**2*HXOHX(J)*[1.00-CW(J)**2]/2.00/HX(J)
        1)+AL1**2/2.00*DSQR(2.00)*(XIFXX(J)+DVE*DX*XIFXX(J)/DSQR(DUE*DX**2
        2))/DUE*DX**2
        A4(J)=0.00
        A5(J)=-AL1*GW(J)/HWX(1)
        CONTINUE
        IF ITRITE.EQ.0) GO TO 50
        WRITE (6,80)
        DO 40 I=1,IE,5
        WRITE (6,70) A0(I),A1(I),A2(I),A3(I),A4(I),A5(I)
        CONTINUE
        RETURN
        C
        90      FORMAT (1H0,12HIN XHOM A0-5)
        70      FORMAT (1H0,6E13.6)
        END
```

XHOM 530
XHOM 540
XHOM 550
XHOM 560
XHOM 570
XHOM 580
XHOM 590
XHOM 600
XHOM 610
XHOM 620
XHOM 630
XHOM 640
XHOM 650
XHOM 660
XHOM 670
XHOM 680
XHOM 690
XHOM 700
XHOM 710
XHOM 720
XHOM 730
XHOM 740